

2022, or within 90 days after the effective date of this AD, whichever occurs later. Accomplishing the revision of the existing maintenance or inspection program required by this paragraph terminates the actions required by paragraph (g) of this AD for Sections 01, "Airworthiness limitations—Introduction;" 02, "Certification maintenance requirements—General;" 04, "ALI structural inspections—General;" 05, "Life limited parts (systems)—General;" 06, "Life limited parts (structures)—General;" 07, "Fuel system limitations—General;" 08, "Critical design configuration control limitations—General;" 09, "Power plant limitations—General;" 10, "Structural repair limitations—General;" and 11, "Limit of validity—General;" of Airbus Canada Limited Partnership A220 Airworthiness Limitations, BD500–3AB48–11400–02, Issue 011.00, dated June 18, 2020, only.

(j) New No Alternative Actions, Intervals, or CDCCLs

After the existing maintenance or inspection program has been revised as required by paragraph (i) of this AD, no alternative actions (e.g., inspections), intervals, or CDCCLs may be used unless the actions, intervals, and CDCCLs are approved as an AMOC in accordance with the procedures specified in paragraph (k)(1) of this AD.

(k) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, New York ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the certification office, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7300. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, New York ACO Branch, FAA; or Transport Canada Civil Aviation (TCCA); or Airbus Canada Limited Partnership's TCCA Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(l) Additional Information

(1) Refer to TCCA AD CF–2022–18, dated April 14, 2022, for related information. This TCCA AD may be found in the AD docket regulations.gov by searching for and locating Docket No. FAA–2022–1308.

(2) For more information about this AD, contact Gabriel Kim, Aerospace Engineer, Airframe and Propulsion Section, FAA, New

York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7300; email 9-avs-nyaco-cos@faa.gov.

(m) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following service information was approved for IBR on December 5, 2022.

(i) Airbus Canada Limited Partnership A220 Airworthiness Limitations, BD500–3AB48–11400–02, Issue 014.00, dated February 3, 2022.

(ii) [Reserved]

(4) The following service information was approved for IBR on March 30, 2021 (86 FR 10799, February 23, 2021).

(i) Airbus Canada Limited Partnership A220 Airworthiness Limitations, BD500–3AB48–11400–02, Issue 011.00, dated June 18, 2020.

(ii) [Reserved]

(5) For service information identified in this AD, contact Airbus Canada Limited Partnership, 13100 Henri-Fabre Boulevard, Mirabel, Québec J7N 3C6, Canada; telephone 450–476–7676; email a220_crc@abc.airbus; website a220world.airbus.com.

(6) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(7) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on October 17, 2022.

Ross Landes,

Deputy Director for Regulatory Operations, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022–23012 Filed 10–28–22; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2022–1313; Project Identifier MCAI–2021–01418–T]

RIN 2120–AA64

Airworthiness Directives; BAE Systems (Operations) Limited Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to supersede Airworthiness Directive (AD) 99–25–11, which applies to certain BAE Systems (Operations) Limited Model BAe 146 and Avro 146–RJ series airplanes. AD 99–25–11 requires repetitive inspections for cracks along the face of the retraction attachment boss in the nose landing gear (NLG) sidewall; and corrective action, if necessary. Since the FAA issued AD 99–25–11, additional cracking was found that indicated additional airplanes are subject to the unsafe condition. This proposed AD would continue to require the actions in AD 99–25–11, and expand the applicability. The FAA is proposing this AD to address the unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by December 15, 2022.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to regulations.gov. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact BAE Systems (Operations) Limited, Customer Information Department, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland, United Kingdom; telephone +44 1292 675207; fax +44 1292 675704; email RApublications@baesystems.com; website baesystems.com/Businesses/RegionalAircraft/index.htm. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

Examining the AD Docket

You may examine the AD docket at regulations.gov by searching for and locating Docket No. FAA–2022–1313; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the mandatory continuing airworthiness

information (MCAI), any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Todd Thompson, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone 206-231-3228; email Todd.Thompson@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2022-1313; Project Identifier MCAI-2021-01418-T" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend the proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to *regulations.gov*, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this proposed AD.

Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Todd Thompson, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des

Moines, WA 98198; telephone 206-231-3228; email Todd.Thompson@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

Background

The FAA issued AD 99-25-11, Amendment 39-11454 (64 FR 72522, December 28, 1999) (AD 99-25-11), for certain BAE Systems (Operations) Limited Model BAe 146 and Avro 146-RJ series airplanes. FAA AD 99-25-11 requires repetitive eddy current inspections for cracks along the face of the retraction attachment boss in the NLG sidewall; and corrective action, if necessary. FAA AD 99-25-11 was prompted by issuance of mandatory continuing airworthiness information by a foreign civil aviation authority. The FAA issued AD 99-25-11 to address cracking along the face of the retraction attachment boss in the NLG sidewall, which could result in premature extension of the NLG or result in depressurization of the airplane.

FAA AD 99-25-11 corresponds to British AD 015-10-98.

Actions Since AD 99-25-11 Was Issued

Since the FAA issued AD 99-25-11, the FAA has determined that additional airplanes are subject to the identified unsafe condition.

The Civil Aviation Authority (CAA), which is the aviation authority for the United Kingdom, has issued CAA AD G-2021-0016R1, dated February 18, 2022 (U.K. CAA AD G-2021-0016R1) (also referred to after this as the MCAI), to correct an unsafe condition for certain BAE Systems (Operations) Limited Model BAe 146 and Avro 146-RJ series airplanes. U.K. CAA AD G-2021-0016R1 superseded EASA AD 2007-0305, dated December 20, 2007, which superseded British AD 015-10-98. The FAA did not issue an AD corresponding to EASA AD 2007-0305.

Evidence of cracking was found on several in-service airplanes in the bore and along the face of the retraction jack attachment boss on the left-hand NLG sidewall. Undetected cracking of the NLG sidewall could lead to explosive decompression of the fuselage near to the flightcrew (since the NLG sidewall forms part of the nose fuselage pressure shell), leading to significant structural damage to the airframe and/or incapacitation of the flightcrew.

The effectivity of each revision of ISB.53-152 before Revision 8 was limited to airplanes that were not modified by torque tightening modification HCM01641A in production. BAE Systems (Operations)

Limited has received reports of two airplanes with cracks at the NLG retraction jack attachment boss; those airplanes were post-modification HCM01641A and as such were not subject to the requirements of FAA AD 99-25-11. As a result of new findings and further analysis, BAE Systems (Operations) Limited issued Revision 8 of ISB.53-152, dated February 19, 2018, which extends the effectivity to all BAe 146 and Avro 146-RJ airplanes, except for airplanes post-modification HCM20011A, HCM20012A, HCM20013A, HCM20313A, HCM20314A, or HMC20315A.

Revisions prior to Revision 8 of ISB.53-152 included provisions for continued operation with certain crack conditions, which was also allowed in FAA AD 99-25-11 if approved as specified in paragraph (b) of this AD. The U.K. CAA and the FAA have determined that continued operation with known cracks is not acceptable. Therefore, this proposed AD would not allow flight with cracks.

The FAA is proposing this AD to address cracking along the face of the retraction attachment boss in the NLG sidewall, which could result in premature extension of the NLG or result in depressurization of the airplane. See the MCAI for additional background information.

Related Service Information Under 1 CFR Part 51

BAE Systems (Operations) Limited has issued Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018. This service information describes procedures for repetitive eddy current inspections for cracking in the bore and along the face of the retraction attachment boss in the left-hand NLG sidewall and, and repair or replacement of a cracked sidewall.

This proposed AD would also require British Aerospace Service Bulletin SB.53-152, dated October 8, 1998, which the Director of the Federal Register approved for incorporation by reference as of February 1, 2000 (64 FR 72522, December 28, 1999).

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA's Determination

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition

described in the MCAI and service information referenced above. The FAA is proposing this AD because the FAA evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop on other products of the same type design.

Proposed Requirements of This NPRM

This proposed AD would retain all of the requirements of AD 99–25–11 and add airplanes to the applicability. This proposed AD would also require accomplishing the actions specified in

the service information described previously.

Costs of Compliance

The FAA estimates that this proposed AD affects 20 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Retained actions from AD 99–25–11	1 work-hour × \$85 per hour = \$85 per inspection cycle.	\$0	\$85 per inspection cycle.	Up to \$1,700 per inspection cycle.
New proposed actions	2 work-hours × \$85 per hour = \$170	0	\$170 per inspection cycle.	\$3,400 per inspection cycle.

The FAA has received no definitive data on which to base the cost estimates for the on-condition actions specified in this proposed AD.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

The FAA has determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by:
 - a. Removing Airworthiness Directive (AD) 99–25–11, Amendment 39–11454 (64 FR 72522, December 28, 1999); and
 - b. Adding the following new AD:

BAE Systems (Operations) Limited: Docket No. FAA–2022–1313; Project Identifier MCAI–2021–01418–T.

(a) Comments Due Date

The FAA must receive comments by December 15, 2022.

(b) Affected ADs

This AD replaces AD 99–25–11, Amendment 39–11454 (64 FR 72522, December 28, 1999) (AD 99–25–11).

(c) Applicability

This AD applies to BAE Systems (Operations) Limited Model BAe 146–100A, –200A, and –300A airplanes; and Model Avro 146–RJ70A, 146–RJ85A, and 146–RJ100A airplanes; certificated in any category, without modification HCM20011A, HCM20012A, HCM20013A, HCM20313A, HCM20314A, or HMC20315A.

(d) Subject

Air Transport Association (ATA) of America Code 53, Fuselage.

(e) Reason

This AD was prompted by a report of a crack found on the left-hand sidewall well on the nose landing gear (NLG), and by the determination that additional airplanes are subject to the identified unsafe condition. We are issuing this AD to address cracking along the face of the retraction attachment boss in the NLG sidewall, which could result in premature extension of the NLG or result in depressurization of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Retained Repetitive Inspections, With New Terminating Action

This paragraph restates the requirements of paragraph (a) of AD 99–25–11, with new terminating action. For airplanes listed in British Aerospace Service Bulletin SB.53–152, dated October 8, 1998: Prior to the accumulation of 8,000 total flight cycles, or within 200 flight cycles after February 1, 2000 (the effective date of AD 99–25–11), whichever occurs later, perform an eddy current inspection to detect cracking along the face of the retraction attachment boss in the nose landing gear sidewall, in accordance with British Aerospace Service Bulletin SB.53–152, dated October 8, 1998. Thereafter, repeat the eddy current inspection at intervals not to exceed 2,600 flight cycles, except as provided in paragraph (j) of this AD.

(h) Retained Repair, With Revised Repair Approval

This paragraph restates the requirements of paragraph (b) of AD 99–25–11, with revised repair approval.

(1) If any crack is detected before the effective date of this AD, during any inspection required by paragraph (g) of this AD, prior to further flight, repair or reinspect in accordance with a method approved by either the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate; or the Civil Aviation Authority (or its delegated agent). For a repair method to be approved by the Manager, International Branch, ANM–116, as required by this

paragraph, the Manager's approval letter must specifically reference this AD.

(2) If any crack is detected on or after the effective date of this AD during any inspection required by paragraph (g) of this AD: Before further flight, either repair using a method approved by the Manager, Large Aircraft Section, International Validation Branch, FAA; or the U.K. Civil Aviation Authority (U.K. CAA); or BAE Systems (Operations) Limited's U.K. CAA Design Organization Approval (DOA); or do the replacement specified in paragraph (i) of this AD. If approved by the DOA, the approval must include the DOA-authorized signature.

(i) New Requirements: Repetitive Inspections and Corrective Actions

(1) For all airplanes: Before the accumulation of 7,375 total flight cycles, or within 625 flight cycles after the effective date of this AD, or within 2,600 flight cycles since the most recent inspection required by paragraph (g) of this AD, whichever occurs latest, do an eddy current inspection for cracking in the bore and along the face of the retraction jack attachment boss in the left-hand nose landing gear sidewall, in accordance with the Accomplishment Instructions of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018. Before further flight, repair or replace any cracked sidewall, as applicable, in accordance with BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018. Repeat the inspection thereafter at intervals not to exceed 6,700 flight cycles, except as provided in paragraphs (i)(1)(i) and (ii) of this AD.

(i) For airplanes on which a repair identified for Option A, D, or E in Table 1 of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018, has been done: Inspect within 20,000 flight cycles after the repair, and repeat thereafter at intervals not to exceed 4,000 flight cycles.

(ii) For airplanes on which the replacement with part number HC537L0002-000, -002, or -004 identified in Option F in Table 1 of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018, has been done: Inspect within 20,000 flight cycles after the repair, and repeat thereafter at intervals not to exceed 4,000 flight cycles.

(2) For airplanes on which re-inspection of cracks was allowed as specified in paragraph (h)(1) of this AD: Within 2,600 flight cycles after the most recent inspection required by paragraph (g) of this AD, repair or replace any cracked sidewall, as applicable, in accordance with BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018.

(j) Terminating Action

(1) Accomplishment of the initial inspection and applicable corrective actions required by paragraph (i) of this AD terminates the repetitive inspection requirements of paragraph (g) of this AD.

(2) Accomplishment of the action identified for Option B or C in Table 1 of

BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018, terminates the repetitive inspection requirements of paragraphs (g) and (i)(1) of this AD.

(3) Accomplishment of the replacement with part number HC537L0002-006 identified for Option F in Table 1 of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018, terminates the repetitive inspection requirements of paragraphs (g) and (i)(1) of this AD.

(k) Credit for Previous Actions

(1) This paragraph provides credit for actions required by paragraphs (i) and (j)(2) of this AD, if those actions were performed before the effective date of this AD using the service information identified in paragraphs (k)(1)(i) and (ii) of this AD.

(i) BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 6, dated March 5, 2014.

(ii) BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 7, dated May 7, 2014.

(2) This paragraph provides credit for the actions required by paragraph (j)(3) of this AD, if those actions were performed before the effective date of this AD using the service information identified in paragraphs (k)(2)(i) and (ii) of this AD, provided the sidewall replacement for Option F was part number HC537L0002-006.

(i) BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 6, dated March 5, 2014.

(ii) BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 7, dated May 7, 2014.

(l) No Reporting Requirement

Although BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-152, Revision 8, dated February 19, 2018, specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(m) Other FAA AD Provisions

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, Large Aircraft Section, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the Large Aircraft Section, International Validation Branch, send it to the attention of the person identified in paragraph (n)(2) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: As of the effective date of this AD, for any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation

Branch, FAA; or the U.K. CAA; or BAE Systems (Operations) Limited's U.K. CAA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(n) Related Information

(1) For related information, refer to U.K. CAA AD G-2021-0016R1, dated February 18, 2022. This mandatory continuing airworthiness information (MCAI) may be found in the AD docket at regulations.gov by searching for and locating Docket No. FAA-2022-1313.

(2) For more information about this AD, contact Todd Thompson, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone 206-231-3228; email Todd.Thompson@faa.gov.

(3) For service information identified in this AD, contact BAE Systems (Operations) Limited, Customer Information Department, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland, United Kingdom; telephone +44 1292 675207; fax +44 1292 675704; email RApublications@baesystems.com; website baesystems.com/Businesses/RegionalAircraft/index.htm. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

Issued on October 20, 2022.

Christina Underwood,

Acting Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022-23201 Filed 10-28-22; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2022-1267; Airspace Docket No. 22-AAL-23]

RIN 2120-AA66

Proposed Revocation of Federal Colored Airway A-9; Bettles, AK

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This action proposes to remove Colored Federal airway A-9 due to the planned decommissioning of the Evansville, AK (EAV), Non-Directional Beacon (NDB), which provides navigation guidance for the affected route. The Evansville, AK (EAV), NDB is scheduled to be decommissioned effective on June 15, 2023.

DATES: Comments must be received on or before December 15, 2022.