

current inspection requirements of paragraph (e)(1) of this AD for this location only.

However, the repetitive detailed inspection required by paragraph (c) of this AD is still required. If any "oil can" remains after the repair, repeat the eddy current inspection thereafter at intervals not to exceed 1,000 flight cycles.

Indication of Previous "Oil Can" Repairs

(f) If any previous "oil can" repair is found during the detailed inspection required by paragraph (b) or (c) of this AD, before further flight, do a detailed inspection of the web for cracks and "oil cans," as shown in Figure 4 or Figure 5 of the service bulletin, as applicable.

(1) If no crack and no "oil can" are found, repeat the detailed inspection in accordance with paragraph (c) of this AD.

(2) If any "oil can" is found, before further flight, do the eddy current inspection for cracks as shown in Figure 3 of the service bulletin.

(3) If no crack is found during the eddy current inspection required by paragraph (f)(2) of this AD, do the actions specified in paragraph (e)(1) or (e)(2) of this AD, as applicable.

Repair of Cracks

(g) If any crack is found during any inspection required by this AD, before further flight, repair in accordance with the service bulletin. If cracks or damage exceeds limits specified in the service bulletin and the service bulletin specifies to contact Boeing for appropriate action: Before further flight, repair per a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA; or per data meeting the type certification basis of the airplane approved by a Boeing Company Designated Engineering Representative who has been authorized by the Manager, Seattle ACO, to make such findings. For a repair method to be approved, the approval must specifically reference this AD.

Alternative Methods of Compliance

(h) In accordance with 14 CFR 39.19, the Manager, Seattle ACO, FAA, is authorized to approve alternative methods of compliance for this AD.

Issued in Renton, Washington, on January 29, 2004.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 04-2469 Filed 2-5-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-227-AD]

RIN 2120-AA64

Airworthiness Directives; Dassault Model Fan Jet Falcon Series Airplanes and Model Mystere-Falcon 20 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Dassault Model Fan Jet Falcon series airplanes and Model Mystere-Falcon 20 series airplanes. This proposal would require inspecting and testing for fatigue cracking due to stress corrosion in the vertical posts of the window frames in the flight compartment. This action is necessary to prevent fatigue cracking of the window frames, which could result in rapid depressurization of the fuselage and consequent reduced structural integrity of the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by March 8, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-227-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2002-NM-227-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from Dassault Falcon Jet, P.O. Box 2000, South Hackensack, New Jersey 07606. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1137; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002-NM-227-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-227-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the

airworthiness authority for France, notified the FAA that an unsafe condition may exist on certain Model Fan Jet Falcon series airplanes and Model Mystere-Falcon 20 series airplanes. The DGAC advises that cracking has been found in the vertical posts of window frames in the flight compartment that were not included in the maintenance program inspection area. The airplane manufacturer, Dassault Aviation, has revised certain work cards in the maintenance manual to include procedures for inspecting and testing these window frames. Cracking of the window frames could result in rapid depressurization of the fuselage and consequent reduced structural integrity of the airplane.

Explanation of Relevant Service Information

Dassault has issued revised Work Card 53-30-07, "Non-Destructive Ultrasonic Testing of Vertical Posts on Screw-Mounted Windows" of the Dassault Aviation Fan Jet Falcon Maintenance Manual, dated November 2001. The work card describes procedures for testing (including calibrating the equipment) the posts of the window frames in the flight compartment, including the right side window, left and right rear windows, front window, and pilot and co-pilot windows, for stress corrosion cracking.

Dassault has also issued revised Work Card 53-30-12, "Endoscopic Inspection of the Frames of Pilot, Co-Pilot, and Front Glass Panels (Aircraft Not Changed Per SB No. 701)" of the Dassault Aviation Fan Jet Falcon Maintenance Manual, dated November 2001. The work card describes procedures for using an endoscope and inspecting the window frames of the pilot, co-pilot, and front windows in the flight compartment for stress corrosion and subsequent cracking. Accomplishment of the actions specified in the service information is intended to adequately address the identified unsafe condition. The DGAC classified this service information as mandatory and issued French airworthiness directive 2001-600-028(B), dated December 12, 2001, to ensure the continued airworthiness of these airplanes in France.

FAA's Conclusions

These airplane models are manufactured in France and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral

airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of Requirements of Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified in the service information described previously. This proposed AD also includes a reporting requirement.

Difference Between the Proposed AD and the French Airworthiness Directive

The French airworthiness directive is applicable to "all Fan Jet Falcon and Mystere Falcon 20-(x)5 aircraft, all serial numbers." The applicability of this proposed AD is "Model Fan Jet Falcon series airplanes and Model Mystere-Falcon 20 series airplanes; certificated in any category; except those which have incorporated Dassault Service Bulletin AMD-BA FJF-701, Revision 1, dated October 22, 1987." This exception appears in Dassault Aviation Work Card 53-30-12, titled "Endoscopic Inspection of the Frames of Pilot, Co-Pilot, and Front Glass Panels (Aircraft Not Changed Per SB No. 701)," of the Dassault Aviation Fan Jet Falcon Maintenance Manual, dated November 2001. This difference has been coordinated with the DGAC.

Cost Impact

The FAA estimates that 220 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 4 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$57,200, or \$260 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time

required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Dassault Aviation: Docket 2002-NM-227-AD.

Applicability: Model Fan Jet Falcon series airplanes and Model Mystere-Falcon 20 series airplanes; certificated in any category; except those which have incorporated Dassault Service Bulletin AMD-BA FJF-701, Revision 1, dated October 22, 1987.

Compliance: Required as indicated, unless accomplished previously.

To prevent fatigue cracking of the window frames in the flight compartment, which could result in rapid depressurization of the

fuselage and consequent reduced structural integrity of the airplane, accomplish the following:

Inspection and Test of Flight Compartment Window Frames

(a) Do an inspection and test for stress corrosion and cracking as specified in paragraphs (a)(1) and (a)(2) of this AD, at the applicable time specified in paragraph (b) of this AD.

(1) Do a detailed inspection (using an endoscope) to detect stress corrosion and cracking of the window frames in the flight compartment, including the pilot, co-pilot, and front windows. Do the inspection in accordance with Dassault Aviation Work Card 53-30-12, titled "Endoscopic Inspection of the Frames of Pilot, Co-Pilot, and Front Glass Panels (Aircraft Not Changed Per SB No. 701)," of the Dassault Aviation Fan Jet Falcon Maintenance Manual, dated November 2001.

(2) Do an ultrasonic test for cracking in the posts of the window frames, including the right side window, left and right rear windows, front window, and pilot and co-pilot windows. Do the test in accordance with Dassault Aviation Work Card 53-30-07, titled "Non-Destructive Ultrasonic Testing of Vertical Posts on Screw-Mounted Windows," of the Dassault Aviation Fan Jet Falcon Maintenance Manual, dated November 2001.

Note 1: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

(b) Do the inspection and test required by paragraph (a) of this AD, at the times specified in paragraph (b)(1) or (b)(2) of this AD, as applicable.

(1) For airplanes having 35 or more years since the date of issuance of the original Airworthiness Certificate or the date of issuance of the Export Certificate of Airworthiness, whichever is first; or having accumulated 20,000 or more total flight cycles; as of the effective date of this AD: Within 7 months after the effective date of this AD.

(2) For airplanes not identified in paragraph (b)(1) of this AD: Within 25 months or 2,500 flight cycles after the effective date of this AD, whichever is first.

Repair

(c) If any stress corrosion or cracking is found during any inspection or test required by paragraph (a) of this AD: Before further flight, repair per a method approved by either the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate; or the Direction Générale de l'Aviation Civile (or its delegated agent).

Reporting Requirement

(d) At the applicable time specified in paragraph (d)(1) or (d)(2) of this AD: Submit

a report of the findings (positive and negative) of the inspection required by paragraph (a) of this AD to: Dassault Falcon Jet, Attn: Service Engineering/Falcon 20, fax: (201) 541-4706, at the applicable time specified in paragraph (d)(1) or (d)(2) of this AD. The report must include the airplane serial number, number of landings, number of flight hours, airplane age, and the number and length of any cracks found. Submission of the Charts of Records (part of French airworthiness directive 2001-600-028(B), dated December 12, 2001), is an acceptable method of complying with this requirement. Under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 *et seq.*), the Office of Management and Budget (OMB) has approved the information collection requirements contained in this AD and has assigned OMB Control Number 2120-0056.

(1) If the inspection was done after the effective date of this AD: Submit the report within 5 days after the inspection.

(2) If the inspection was done prior to the effective date of this AD: Submit the report within 5 days after the effective date of this AD.

Alternative Methods of Compliance

(e) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, is authorized to approve alternative methods of compliance for this AD.

Note 2: The subject of this AD is addressed in French airworthiness directive 2001-600-028(B), dated December 12, 2001.

Issued in Renton, Washington, on January 29, 2004.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-2470 Filed 2-5-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NM-127-AD]

RIN 2120-AA64

Airworthiness Directives; Short Brothers Model SD3-60 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all Short Brothers Model SD3-60 series airplanes. This proposal would require performing an inspection of the shear attachment fitting for the fin-to-fuselage front spar, and of the shear cleat for the

fin root rib at the aft spar location for corrosion; reporting inspection results; and performing corrective action, if necessary. This action is necessary to detect and correct corrosion in the area of the main spar web fittings of the vertical stabilizer, which could result in reduced structural integrity of the vertical stabilizer. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by March 8, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2003-NM-127-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2003-NM-127-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Short Brothers, Airworthiness & Engineering Quality, P.O. Box 241, Airport Road, Belfast BT3 9DZ, Northern Ireland. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1175; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.