

defined by the Bank in its AHP implementation plan.

* * * * *

(D) *Housing for homeless households.*

The creation of rental housing, excluding overnight shelters, reserving at least 20 percent of the units for homeless households, the creation of transitional housing for homeless households permitting a minimum of six months occupancy, or the creation of permanent owner-occupied housing reserving at least 20 percent of the units for homeless households. For purposes of this paragraph, the term "homeless households" shall have the meaning as defined by the Bank in its AHP implementation plan.

* * * * *

§ 951.7 [Amended]

5. Amend § 951.7 by:

a. In the section heading, adding the words "or after" between the words "to" and "project"; and

b. In the introductory text of paragraph (a), adding the words "or after" between the words "to" and "final."

§ 951.9 [Removed]

6. Remove § 951.9.

7. Revise § 951.10(c)(2) introductory text and paragraph (c)(2)(i) to read as follows:

§ 951.10 Initial monitoring requirements.

* * * * *

(c) * * *

(2) *Rental projects.* Each Bank must take the steps necessary to determine that, within 120 days after receiving the documentation described in paragraph (a)(2)(ii) of this section:

(i) The services and activities committed to in the AHP application have been provided in connection with the project; and

* * * * *

8. Amend § 951.12 by revising paragraph (e) to read as follows:

§ 951.12 Remedial actions for noncompliance.

* * * * *

(e) *Use of repaid subsidies—(1) Use of repaid AHP subsidies in other AHP-eligible projects.* Except as provided in paragraph (e)(2) of this section, amounts of AHP subsidy repaid to a Bank pursuant to this section, including any interest, shall be made available by the Bank for other AHP-eligible projects.

(2) *Re-use of repaid AHP subsidies in same project.* Where AHP direct subsidy has been provided by the project sponsor (or the member in the case of direct subsidy provided through the homeownership set-aside program) as

downpayment, closing cost, rehabilitation or interest rate buydown assistance to a household to purchase an owner-occupied unit pursuant to an approved AHP application, amounts of AHP subsidy repaid to the Bank, including any interest, may, if authorized, in the Bank's discretion, in its AHP implementation plan and within the period of time specified by the Bank in such plan, be made available by the project sponsor or member to another AHP-eligible household to purchase an owner-occupied unit in accordance with the terms of the approved AHP application.

* * * * *

9. Amend § 951.13 by adding paragraph headings to paragraph (c)(3)(i) and (c)(3)(ii), and adding paragraph (c)(3)(iii) to read as follows:

§ 951.13 Agreements.

* * * * *

(c) * * *

(3) * * *

(i) *Repayment of advance.* * * *

(ii) *Maintain advance outstanding with reset interest rate.* * * *

(iii) *Loan pool substitution.* If authorized, in the Bank's discretion, in its AHP implementation plan, continue to maintain the advance outstanding without the Bank resetting the interest rate, provided that:

(A) The loan, before its prepayment, was used by a household to purchase an owner-occupied unit pursuant to the project sponsor's approved AHP application;

(B) The loan was purchased by the member from the project sponsor and held by the member as part of a pool of loans financed by subsidized advances or direct subsidies and purchased from the project sponsor;

(C) Within the period of time specified by the Bank in its AHP implementation plan, the member makes the unused AHP subsidy resulting from the prepaid loan available to the project sponsor to reduce the interest rate on a new loan from the project sponsor to another AHP-eligible household to purchase an owner-occupied unit in accordance with the terms of the approved AHP application;

(D) Within the period of time specified by the Bank in its AHP implementation plan, the member purchases the new loan for inclusion in the loan pool; and

(E) After substitution of the new loan for the prepaid loan in the loan pool, the aggregate principal balance of the loan pool is the same as or higher than the

original principal balance of the loan pool.

* * * * *

Dated: December 11, 2001.

By the Board of Directors of the Federal Housing Finance Board.

J. Timothy O'Neill,
Chairman.

[FR Doc. 01-31569 Filed 12-26-01; 8:45 am]

BILLING CODE 6725-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 98-ANE-61-AD]

RIN 2120-AA64

Airworthiness Directives; Pratt & Whitney (PW) PW2000 Series Turbofan Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Supplemental notice of proposed rulemaking (NPRM); reopening of comment period.

SUMMARY: This action revises an earlier proposed airworthiness directive (AD), applicable to certain Pratt & Whitney (PW) PW2000 series turbofan engines, that would supersede an existing airworthiness directive (AD) by modifying the airworthiness limitations section of the manufacturer's manual and an air carrier's approved continuous airworthiness maintenance program to incorporate additional inspection requirements. This action revises the proposed rule by adding the low pressure compressor (LPC) hub assembly, high pressure turbine (HPT) 1st stage disk, and HPT 2nd stage hub to the additional inspection requirements. The regulatory section revises the manufacturer's Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness (ICA), and for air carrier operations revises the approved continuous airworthiness maintenance program. The actions specified by this proposed AD are intended to prevent critical life-limited rotating engine part failure, which could result in an uncontained engine failure and damage to the airplane.

DATES: Comments must be received by January 28, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 98-ANE-

61-AD, 12 New England Executive Park, Burlington, MA 01803-5299. Comments may be inspected at this location, by appointment, between 8 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays. Comments may also be sent via the Internet using the following address: "*9-ane-adcomment@faa.gov*." Comments sent via the Internet must contain the docket number in the subject line.

FOR FURTHER INFORMATION CONTACT:

Jason Yang, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7747, fax (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this proposal must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 98-ANE-61-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRM's

Any person may obtain a copy of this NPRM by submitting a request to the FAA, New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 98-ANE-61-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

Discussion

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to add an airworthiness directive (AD), applicable to certain Pratt & Whitney (PW) PW2000 series turbofan engines, was published as a notice of proposed rulemaking (NPRM) in the **Federal Register** on November 1, 2001 (66 FR 55138). That NPRM would have modified the airworthiness limitations section of the manufacturer's manual and an air carrier's approved continuous airworthiness maintenance program to incorporate additional inspection requirements. That NPRM was prompted by an FAA study of in-service events involving uncontained failures of critical rotating engine parts. That condition, if not corrected, could result in an uncontained engine failure and damage to the airplane.

Since the issuance of that NPRM, the FAA has become aware that the LPC hub assembly, HPT 1st stage disk, and HPT 2nd stage hub were inadvertently omitted from the list of parts for enhanced inspection, and must be added. This proposal would add to that NPRM, modification of the airworthiness limitations section of the manufacturer's manual and air carrier's approved continuous airworthiness maintenance program to incorporate additional inspection requirements.

Since this change expands the scope of the originally proposed rule, the FAA has determined that it is necessary to reopen the comment period to provide additional opportunity for public comment.

Economic Analysis

The FAA estimates that 724 engines installed on airplanes of US registry would be affected by this proposed AD, that it would take approximately 20 work hours per engine to do the proposed actions. The average labor rate is \$60 per work hour. The cost effect of the added inspections per engine is approximately \$1,200 per year, with the approximate total cost for the U.S. fleet of \$868,800 per year.

Regulatory Analysis

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this action does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1)

is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic effect, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Pratt & Whitney: Docket No. 98-ANE-61-AD. Supersedes AD 2000-21-09, Amendment 39-11914.

Applicability: This airworthiness directive (AD) is applicable to Pratt & Whitney (PW) PW2037, PW2040, PW2037M, PW2240, PW2337, PW2043, PW2643, and PW2143, series turbofan engines, installed on but not limited to Boeing 757 series and Ilyushin IL-96T series airplanes.

Note 1: This AD applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless already done.

To prevent critical life-limited rotating engine part failure, which could result in an

uncontained engine failure and damage to the airplane, do the following:

(a) Within 30 days after the effective date of this AD, revise the manufacturer's Time Limits section (TLS) of the manufacturer's engine manual, as appropriate for PW

PW2037, PW2040, PW2037M, PW2240, PW2337, PW2043, PW2643, and PW2143 series turbofan engines, and for air carriers revise the approved continuous airworthiness maintenance program, by adding the following:

“MANDATORY INSPECTIONS

(1) Perform inspections of the following parts at each piece-part opportunity in accordance with the instructions provided in PW2000 Engine Manuals 1A6231 and 1B2412:

Nomenclature	Part number	EM manual section	Inspection check	Subtask
Hub, LPC Assembly	ALL	72-31-04	-06.	
Disk, HPT 1st Stage	ALL	72-52-02	FPI entire disk per 72-52-00, Inspection/Check-02.	72-52-02-230-007
Hub, HPT 2nd Stage	ALL	72-52-16	Fpi entire hub per 72-52-00, Inspection/Check-02.	72-52-16-230-007
Hub, HPC Front	ALL	72-35-02	-05.	
Disk, HPC Drum Rotor Assembly (7-15)	ALL	72-35-03	-04	
Disk, HPC Drum Rotor Assembly (16-17)	ALL	72-35-10	-05	
Disk, HPC 16th Stage	ALL	72-35-06	-04	
Disk, HPC 17th Stage	ALL	72-35-07	-04	
LPC Drive Turbine Shaft	ALL	72-32-01	-06	
Hub, Turbine Rear	ALL	72-53-81	-06	
Disk, LPT 3rd Stage	ALL	72-53-31	-01	
Disk, LPT 4th Stage	ALL	72-35-41	-01	
Disk, LPT 5th Stage	ALL	72-32-51	-01	
Disk, LPT 6th Stage	ALL	72-53-61	-01	
Disk, LPT 7th Stage	ALL	72-53-71	-01	

(2) For the purposes of these mandatory inspections, piece-part opportunity means:

(i) The part is considered completely disassembled when done in accordance with the disassembly instructions in the manufacturer's engine manual to either part number level listed in the table above, and

(ii) The part has accumulated more than 100 cycles in service since the last piece-part opportunity inspection, provided that the part was not damaged or related to the cause for its removal from the engine.”

(b) Except as provided in paragraph (e) of this AD, and notwithstanding contrary provisions in § 43.16 of Federal Aviation Regulations (14 CFR 43.16), these enhanced inspections must be performed only in accordance with the TLS of the appropriate PW2000 series engine manuals.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their requests through an appropriate FAA Principal Maintenance Inspector (PMI), who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the aircraft to a location where the requirements of this AD can be done.

(e) FAA-certificated air carriers that have an approved continuous airworthiness maintenance program in accordance with the record keeping requirement of § 121.369 (c) of the Federal Aviation Regulations (14 CFR

121.369 (c)) of this chapter must maintain records of the mandatory inspections that result from revising the Time Limits section of the Instructions for Continuous Airworthiness (ICA) and the air carrier's continuous airworthiness program. Alternatively, certificated air carriers may establish an approved system of record retention that provides a method for preservation and retrieval of the maintenance records that include the inspections resulting from this AD, and include the policy and procedures for implementing this alternate method in the air carrier's maintenance manual required by § 121.369 (c) of the Federal Aviation Regulations (14 CFR 121.369 (c)); however, the alternate system must be accepted by the appropriate PMI and require the maintenance records be maintained either indefinitely or until the work is repeated. Records of the piece-part inspections are not required under § 121.380 (a)(2)(vi) of the Federal Aviation Regulations (14 CFR 121.380 (a)(2)(vi)). All other operators must maintain the records of mandatory inspections required by the applicable regulations governing their operations.

Issued in Burlington, Massachusetts, on December 17, 2001.

Jay J. Pardee,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 01-31557 Filed 12-26-01; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-SW-40-AD]

RIN 2120-AA64

Airworthiness Directives; MD Helicopters, Inc. Model 369D, 369E, 369F, and 369FF Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) for MD Helicopters, Inc. Model 369D, 369E, 369F, and 369FF helicopters with a tailboom modified according to Aerometals supplemental type certificate (STC) SH5055NM or SH4801NM. The AD would require an inspection to identify the part number (P/N) of the bolts that attach the tail rotor gearbox to the tailboom and replacing any bolt of inadequate grip length with an airworthy bolt. This AD would also require determining the number of bolt threads protruding from each nutplate and adding an additional washer if more than four threads protrude. This proposal is prompted by the discovery that bolts of inadequate grip length were specified to attach the tail rotor gearbox to the tailboom. The actions specified by the proposed AD are intended to prevent loss of a tail rotor gearbox due to bolts of inadequate