- (8) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.
- (9) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations, or email fr.inspection@nara.gov.

Issued on November 20, 2024.

Peter A. White.

Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.

[FR Doc. 2024–27591 Filed 11–25–24; 8:45 am]

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2024-1692; Project Identifier MCAI-2024-00050-T; Amendment 39-22878; AD 2024-22-07]

RIN 2120-AA64

Airworthiness Directives; De Havilland Aircraft of Canada Limited (Type Certificate Previously Held by Bombardier, Inc.) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain De Havilland Aircraft of Canada Limited Model DHC–8–401 and –402 airplanes. This AD was prompted by a report of an in-flight event where isolation valve caution messages were received. This AD requires inspecting the fuse/shuttle valve serial numbers, and replacing certain fuse/shuttle valves, as specified in a Transport Canada AD, which is incorporated by reference (IBR). The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective December 31, 2024.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of December 31, 2024.

ADDRESSES:

AD Docket: You may examine the AD docket at regulations gov under Docket No. FAA–2024–1692; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory

continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M—30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For Transport Canada material identified in this AD, contact Transport Canada, Transport Canada National Aircraft Certification, 159 Cleopatra Drive, Nepean, Ontario K1A 0N5, Canada; telephone 888–663–3639; email TC.AirworthinessDirectives-Consignesdenavigabilite.TC@tc.gc.ca. You may find this material on the Transport Canada website at tc.canada.ca/en/aviation.
- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available at regulations.gov under Docket No. FAA–2024–1692.

FOR FURTHER INFORMATION CONTACT: Gabriel Kim, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410,

FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone: 516–228–7300; email: *9-avs-nyaco-cos@faa.gov.*

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain De Havilland Aircraft of Canada Limited Model DHC-8-401 and –402 airplanes. The NPRM published in the Federal Register on June 21, 2024 (89 FR 51988). The NPRM was prompted by AD CF-2024-01, dated January 11, 2024, issued by Transport Canada, which is the aviation authority for Canada (Transport Canada AD CF-2024–01) (also referred to as the MCAI). The MCAI states that an in-service event was reported where the crew received a number two isolation valve (ISO #2) caution message followed by a number one isolation valve (ISO #1) caution message. The landing gear was extended via an alternate extension system as the crew prepared for landing. Upon landing, the crew used the emergency brake to stop the airplane. The airplane stopped safely within the runway limits.

Subsequent maintenance activity discovered an external leak from the main landing gear (MLG) brake assembly, and it was found that the fuse/shuttle valve assembly did not function properly. Further investigation revealed that the fuse/shuttle valve assembly failure resulted from a factory assembly error, which occurred on a limited number of fuse/shuttle valves.

The assembly error can cause valve deformation leading to premature wear, and eventually fuse/shuttle valve failure. This condition, if not corrected, could result in the loss of powered landing gear extension/retraction, outboard and inboard spoilers, nose wheel steering, and normal braking, and possibly a runway excursion.

In the NPRM, the FAA proposed to require inspecting the fuse/shuttle valve serial numbers, and replacing certain fuse/shuttle valves, as specified in Transport Canada AD CF-2024-01. The FAA is issuing this AD to address the unsafe condition on these products.

You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA–2024–1692.

Discussion of Final Airworthiness Directive

Comments

The FAA received a comment from Air Line Pilots Association, International (ALPA). The following presents the comment received on the NPRM and the FAA's response to the comment.

Request To Reduce Compliance Time

ALPA stated the compliance time of 8,000 flight hours or 48 months whichever occurs first after the effective date of the proposed AD is excessive for such unsafe condition, that could result in the loss of powered landing gear extension/retraction, outboard and inboard spoilers, nose wheel steering, normal braking, and possibly a runway excursion. The FAA infers that ALPA is requesting the FAA reduce the compliance time.

The FAA does not agree with the request. The FAA has determined that Transport Canada's compliance time calculation is adequate. The low probability of a critical event is due to the single occurrence and high flight hours. In addition, multiple isolation valves can effectively mitigate hydraulic fluid leaks. After considering all the available information, the FAA has determined that the compliance time, as proposed, represents an appropriate interval of time in which the required actions can be performed in a timely manner within the affected fleet, while still maintaining an adequate level of safety. Additionally, the FAA notes that there has been only one event of an inservice aircraft, and in that event, the aircraft landed safely. With only one event and the high amount of flight hours in the fleet, the probability of the

unsafe condition occurring is low. It is possible that all the systems listed in this AD could fail simultaneously; however, that is unlikely. Aircraft hydraulic systems typically have an isolation valve, which in this case was activated and annunciated. The isolation valve was effective, and although hydraulic fluid leaked, the hydraulic pressure remained at normal levels. This AD has not been changed regarding this request.

Clarification of Unsafe Condition

Paragraph (e) of the proposed AD states the FAA is issuing this AD to address certain fuse/shuttle valves. The FAA has revised paragraph (e) of this AD to state the FAA is issuing this AD to address failure of certain fuse/shuttle valves to clarify it is the failure of the fuse/shuttle valves that could lead to the unsafe condition.

Conclusion

This product has been approved by the aviation authority of another country and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered the comment received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on this product. Except for minor editorial changes, and the change described previously, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Material Incorporated by Reference Under 1 CFR Part 51

Transport Canada AD CF-2024-01 specifies procedures for inspecting the fuse/shuttle valve serial numbers, and if any fuse/shuttle valve assemblies with the listed serial numbers are found, replacing the affected fuse/shuttle valves.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in ADDRESSES section.

Costs of Compliance

The FAA estimates this AD affects 54 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
1 work-hour × \$85 per hour = \$85	\$0	\$85	\$4,590

The FAA estimates the following costs to do any necessary on-condition action required based on the results of

any required actions. The FAA has no way of determining the number of

aircraft that might need this oncondition action:

ESTIMATED COSTS OF ON-CONDITION ACTIONS

Labor cost	Parts cost	Cost per product
5 work-hours × \$85 per hour = \$425 per fuse/shuttle valve	\$64,453 per fuse/shuttle valve assembly	\$64,878

The FAA included all known costs in its cost estimate. According to the manufacturer, however, some or all of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected operators.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds

necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2024–22–07 De Havilland Aircraft of Canada Limited (Type Certificate Previously Held by Bombardier, Inc.): Amendment 39–22878; Docket No. FAA–2024–1692; Project Identifier MCAI–2024–00050–T.

(a) Effective Date

This airworthiness directive (AD) is effective December 31, 2024.

(b) Affected ADs

None.

(c) Applicability

This AD applies to De Havilland Aircraft of Canada Limited Model DHC–8–401 and –402 airplanes, certificated in any category, as identified in Transport Canada AD CF–2024–01, dated January 11, 2024 (Transport Canada AD CF–2024–01).

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight Controls; 29, Hydraulic Power; and 32, Landing Gear.

(e) Unsafe Condition

This AD was prompted by a report of an in-flight event where isolation valve caution messages were received. The FAA is issuing this AD to address failure of certain fuse/shuttle valves. The unsafe condition, if not addressed, could result in the loss of powered landing gear extension/retraction, outboard and inboard spoilers, nose wheel steering, normal braking, and possibly a runway excursion.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, Transport Canada AD CF—2024—01.

(h) Exceptions to Transport Canada AD CF-2024-01

- (1) Where Transport Canada AD CF-2024—01 refers to its effective date, this AD requires using the effective date of this AD.
- (2) Where Transport Canada AD CF-2024-01 refers to hours air time, this AD requires using flight hours.

(i) Additional AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (j) of this AD. Information may be emailed to: AMOC@faa.gov. Before using any approved AMOC, notify your appropriate

principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or Transport Canada; or De Havilland Aircraft of Canada Limited's Transport Canada Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(i) Additional Information

For more information about this AD, contact Gabriel Kim, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone: 516–228–7300; email: 9-avs-nyaco-cos@faa.gov.

(k) Material Incorporated by Reference

- (1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.
- (i) Transport Canada AD CF-2024-01, dated January 11, 2024.
 - (ii) [Reserved]
- (3) For Transport Canada AD CF–2024–01, contact Transport Canada, Transport Canada National Aircraft Certification, 159 Cleopatra Drive, Nepean, Ontario K1A 0N5, Canada; telephone 888–663–3639; email TC.AirworthinessDirectives-Consignes denavigabilite.TC@tc.gc.ca; website tc.canada.ca/en/aviation.
- (4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.
- (5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on October 28, 2024.

Victor Wicklund,

 $\label{lem:potential} \textit{Deputy Director, Compliance \& Airworthiness} \\ \textit{Division, Aircraft Certification Service}.$

 $[FR\ Doc.\ 2024-27593\ Filed\ 11-25-24;\ 8:45\ am]$

BILLING CODE 4910-13-P

DEPARTMENT OF COMMERCE

Bureau of Industry and Security

15 CFR Parts 738, 740, 742 and 774 [Docket No. 241113-0293]

RIN 0694-AJ63

Implementation of Additional Controls on Pakistan

AGENCY: Bureau of Industry and Security, Department of Commerce. **ACTION:** Final rule.

SUMMARY: The Bureau of Industry and Security (BIS) is amending the Export Administration Regulations (EAR) by imposing new licensing requirements on exports, reexports, and transfers (incountry) to and within Pakistan of certain items identified on the Commerce Control List (CCL) that are not currently subject to a license requirement when destined for Pakistan. This change is being made to ensure that such transactions receive U.S. government review to reduce the risk of diversion to an end use or end user of concern.

DATES: This rule is effective November 25, 2024, except for amendatory instruction 6, which is effective November 25, 2024.

FOR FURTHER INFORMATION CONTACT: For questions on this rule, contact Philip Johnson at *RPD2@bis.doc.gov* or (202) 482–2440.

SUPPLEMENTARY INFORMATION:

Background

BIS implements export controls on dual-use and certain munitions items (commodities, software, and technology) under the EAR (15 CFR parts 730-774) to advance U.S. national security and foreign policy interests. Among other controls, the EAR restricts the export, reexport, and transfer (in-country) of items based on their classification on the CCL (supp. no. 1 to 15 CFR part 774) and relevant reason(s) for control (see 15 CFR part 742) applicable to the country of destination, as generally determined by the Commerce Country Chart (supp. no. 1 to 15 CFR part 738). BIS also maintains end-use and end-user controls pursuant to part 744 of the EAR, including controls on transactions involving less-sensitive items designated as EAR99, and controls that apply to specific activities of U.S. persons. Additionally, BIS maintains embargoes and special controls pursuant to part 746 of the EAR. With respect to Pakistan, BIS maintains a combination of CCL-based, end-use, and end-user controls. The end-use controls