

8th business day following the final day of the Request for Referendum period.

(c) *Appeal.* A person declared to be ineligible by FSA can appeal such decision and provide additional documentation to the FSA county office within 5 business days after the postmark date of the letter of notification of ineligibility. FSA will then make a final decision on the producer's eligibility and notify the producer of the decision.

(d) *Number of valid requests for referendum.* A person has been declared eligible and has provided and completed all of the required information on form LS-51-1.

(e) *Number of invalid request for a referendum.* An invalid request for referendum includes, but is not limited to the following:

(1) Form LS-51-1 is not signed or all required information has not been provided;

(2) Form LS-51-1 returned in-person or by facsimile was not received by the last business day of the Request for Referendum period;

(3) Form LS-51-1 returned by mail was not postmarked by midnight of the final day of the Request for Referendum period;

(4) Form LS-51-1 returned by mail was not received in the county FSA office prior to canvassing of the ballots;

(5) Form LS-51-1 is mutilated or marked in such a way that any required information on the form is illegible; or

(6) Form LS-51-1 not returned to the appropriate county FSA office.

§ 1220.624 Confidentiality.

The names of persons requesting a referendum shall be confidential and may not be divulged except as the Secretary may direct.

§ 1220.625 Counting requests.

(a) The requests for a referendum shall be counted by county FSA offices on the same day as the requests are canvassed if there are no ineligibility determinations to resolve. For those county FSA offices that do have ineligibility determinations, the requests shall be counted no later than the 14th business day following the final day of the Request for Referendum period.

(b) Requests for a referendum shall be counted as follows:

(1) Total number of producers who returned a Request for Referendum form LS-51-1;

(2) Number of ineligible producers requesting a referendum;

(3) Number of eligible producers requesting a referendum;

(4) Number of valid requests for a referendum; and

(5) Number of invalid requests for a referendum.

§ 1220.626 FSA county office report.

The county FSA office report shall be certified as accurate and complete by the CED or designee, acting on behalf of the Administrator, AMS, as soon as may be reasonably possible, but in no event later than 18th business day following the final day of the specified period, have prepared and certified the county summary of requests on a form provided by the Administrator, FSA. Each county FSA office shall transmit the results in its county to the FSA State office. The results in each county may be made available to the public upon notification by the Administrator, FSA, that the final results have been released by the Secretary. A copy of the report shall be posted for 30 days following the date of notification by the Administrator, FSA, in the county FSA office in a conspicuous place accessible to the public. One copy shall be kept on file in the county FSA office for a period of at least 12 months after notification by FSA that the final results have been released by the Secretary.

§ 1220.627 FSA State office report.

Each FSA State office shall transmit to the Administrator, FSA, as soon as possible, but in no event later than the 20th business day following the final day of the Request for Referendum period, a report summarizing the data contained in each of the reports from the county FSA offices. One copy of the State summary shall be filed for a period of not less than 12 months after the results have been released and available for public inspection after the results have been released.

§ 1220.628 Results of the request for referendum.

(a) The Administrator, FSA, shall submit to the Administrator, AMS, the reports from all State FSA offices. The Administrator, AMS, shall tabulate the results of the Request for Referendum. USDA will issue an official press release announcing the results of the Request for Referendum and publish the same results in the **Federal Register**. In addition, USDA will post the official results at the following Web site: <http://www.ams.usda.gov/lsg/mpb/rp-soy.htm>. Subsequently, State reports and related papers shall be available for public inspection upon request during normal business hours in the Marketing Programs Branch office, Livestock and Seed Program, AMS, USDA, Room 2638-South, STOP 0251, 1400 Independence Avenue, SW., Washington, DC.

(b) If the Secretary deems necessary, a State report or county report shall be reexamined and checked by such persons who may be designated by the Secretary.

§ 1220.629 Disposition of records.

Each FSA CED will place in sealed containers marked with the identification of the "Request for Soybean Referendum," all of the form LS-51-1's along with the accompanying documentation and county summaries. Such records will be placed in a secure location under the custody of the FSA CED for a period of not less than 12 months after the date of notification by the Administrator, FSA, that the final results have been announced by the Secretary. If the county FSA office receives no notice to the contrary from the Administrator, FSA, by the end of the 12 month period as described above, the CED or designee shall destroy the records.

§ 1220.630 Instructions and forms.

The Administrator, AMS, is authorized to prescribe additional instructions and forms not inconsistent with the provisions of this subpart.

Dated: January 21, 2004.

A.J. Yates,

Administrator, Agricultural Marketing Service.

[FR Doc. 04-1602 Filed 1-26-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-SW-15-AD]

RIN 2120-AA64

Airworthiness Directives; Agusta S.p.A. Model A109C, A109E, and A109K2 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to revise an existing airworthiness directive (AD) for Agusta S.p.A. (Agusta) Model A109C, A109E, and A109K2 helicopters. That AD currently requires inspecting the main rotor blade (blade) tip cap for bonding separation and a crack, and also requires a tap inspection of the tip cap for bonding separation in the blade bond area and a dye penetrant inspection of the tip cap leading edge along the welded joint line

of the upper and lower tip cap skin shells for a crack. This action would require those same actions, but would correct a blade part number (P/N) that was stated incorrectly in the Applicability section of the existing AD. This proposal is prompted by the need to correct a blade P/N. The actions specified by the proposed AD are intended to prevent failure of a blade tip cap, excessive vibration, and subsequent loss of control of the helicopter.

DATES: Comments must be received by March 29, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2001-SW-15-AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. You may also send comments electronically to the Rules Docket at the following address: 9-asw-adcomments@faa.gov. Comments may be inspected at the Office of the Regional Counsel between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Richard Monschke, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, Fort Worth, Texas 76193-0110, telephone (817) 222-5116, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their mailed comments submitted in response to this proposal must submit a self-addressed,

stamped postcard on which the following statement is made: "Comments to Docket No. 2001-SW-15-AD." The postcard will be date stamped and returned to the commenter.

Discussion

On December 19, 2000, Agusta issued Alert Bollettino Tecnico Nos. 109-106, 109K-22, and 109EP-1, all Revision B, which specified inspecting for debond and cracks at the tip cap of blades, P/N 709-0103-01, all dash numbers, through serial numbers 1428 with a prefix of "A5" or "EM".

The Ente Nazionale per l'Aviazione Civile (ENAC), the airworthiness authority for Italy, classified these technical bulletins as mandatory and issued AD Nos. 2000-571, 2000-572, and 2000-573, all dated December 22, 2000, requiring an inspection of the tip cap of blades for disbonds or cracks on the specified Agusta Model A109C, A109E, and A109K2 helicopters.

These helicopter models are manufactured in Italy and are type certificated for operation in the United States under the provisions of 14 CFR 21.29 and the applicable bilateral agreement. Pursuant to the applicable bilateral agreement, ENAC has kept the FAA informed of the situation described above. The FAA has examined the findings of ENAC, reviewed all available information, and determined that AD action is necessary for products of these type designs that are certificated for operation in the United States.

On November 21, 2001, the FAA issued AD 2001-24-07, Amendment 39-12523 (66 FR 60144, December 3, 2001), which superseded AD 98-19-04, Amendment 39-11039, Docket No. 98-SW-40-AD. AD 98-19-04 required inspecting between the metal shells and honeycomb core for bonding separation, visually inspecting the blade tip for swelling or deformation, and visually inspecting the welded bead along the leading edge of the blade tip cap for a crack. AD 2001-24-07 retained those requirements, and added a requirement for a tap inspection of the tip cap for bonding separation in the blade bond area, and a dye penetrant inspection of the tip cap leading edge along the welded joint line of the upper and lower tip cap skin shells for a crack. Installing a tip cap, P/N 709-0103-29-109, on an affected blade is a terminating action for the requirements of the existing AD for that blade. That action was prompted by three occurrences in which the blade tip cap leading edge opened in flight due to cracks, resulting in excessive helicopter vibration. That condition, if not corrected, could result in failure of a

blade tip cap, excessive vibration, and subsequent loss of control of the helicopter.

Since issuing AD 2001-24-07, we discovered that a blade P/N was incorrectly stated in the Applicability section of the AD. P/N 709-0130-01—all dash numbers should have been stated as P/N 709-0103-01—all dash numbers.

The previously described unsafe condition is likely to exist or develop on other helicopters of these same type designs. Therefore, the proposed AD would revise AD 2001-24-07 to correct the P/N and to continue to require:

- A tap inspection of the upper and lower sides of the tip cap for bonding separation and in the tip cap to blade bond area;
- A visual inspection of the upper and lower side of the blade tip cap for swelling or deformation; and
- A dye penetrant inspection of the tip cap leading edge along the welded joint line of the upper and lower tip cap skin shells for a crack.

The FAA estimates that this proposed AD would affect 44 helicopters of U.S. registry, and the proposed actions would take approximately 6 work hours per helicopter to accomplish the initial and repetitive inspection at an average labor rate of \$65 per work hour. Based on these figures, we estimate the total cost impact of the proposed AD on U.S. operators to be \$17,160, assuming that no blade will need to be replaced as a result of these inspections.

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-12523 (66 FR 60144), and by adding a new airworthiness directive (AD), to read as follows:

Agusta S.p.A.: Docket No. 2001-SW-15-AD. Revises AD 2001-24-07, Amendment 39-12523.

Applicability: Model A109C, A109E, and A109K2 helicopters, with main rotor blade (blade), part number (P/N) 709-0103-01—all dash numbers, having a serial number (S/N) up to and including S/N 1428 with a prefix of either “EM-” or “A5-” installed, certificated in any category.

Compliance: Required within 10 hours time-in-service (TIS), unless accomplished previously, and thereafter at intervals not to exceed 25 hours TIS.

To prevent failure of a blade tip cap, excessive vibration, and subsequent loss of control of the helicopter, accomplish the following:

(a) Tap inspect the upper and lower sides of each tip cap for bonding separation between the metal shells and the honeycomb core using a steel hammer, P/N 109-3101-58-1, or a coin (quarter) in the area indicated as honeycomb core on Figure 1 of Alert Bollettino Tecnico Nos. 109-106, 109K-22, or 109EP-1, all Revision B, and dated December 19, 2000 (ABT), as applicable. Also, tap inspect for bonding separation in the tip cap to blade bond area (no bonding voids are permitted in this area).

(b) Visually inspect the upper and lower sides of each blade tip cap for swelling or deformation.

(c) Dye-penetrant inspect the tip cap leading edge along the welded joint line of the upper and lower tip cap skin shells for a crack in accordance with the Compliance Instructions, paragraph 3, of the applicable ABT.

(d) If any swelling, deformation, crack, or bonding separation that exceeds the prescribed limits in the applicable maintenance manual is found, replace the blade with an airworthy blade.

(e) Replacement blades affected by this AD must comply with the repetitive inspection requirements of this AD. Replacing an affected blade with a blade having an

airworthy blade tip cap, P/N 709-0103-29-109, is terminating action for the requirements of this AD for that blade.

(f) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Contact the Safety Management Office, Rotorcraft Directorate, FAA, for information about previously approved alternative methods of compliance.

Note: The subject of this AD is addressed in Ente Nazionale per l'Aviazione Civile (Italy) AD Nos. 2000-571, 2000-572, and 2000-573, all dated December 22, 2000.

Issued in Fort Worth, Texas, on January 16, 2004.

David A. Downey,

Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 04-1687 Filed 1-26-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2002-NM-300-AD]

RIN 2120-AA64

Airworthiness Directives; Dornier Model 328-100 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Dornier Model 328-100 series airplanes. This proposal would require repetitive inspections of certain support arms of the ground spoiler assemblies for cracking, and replacement of any ground spoiler assembly having cracking with a new ground spoiler assembly. This proposal would also require certain inspections for discrepancies of the ground spoiler assemblies and the flap of each wing; and corrective actions if necessary. This action is necessary to prevent failure of the support arms due to cracking, which could result in loss of function and/or separation of the affected ground spoiler assemblies from the airplane, and consequent reduced controllability of the airplane during landing or rejected take-off operations. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by February 26, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport

Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-30AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain “Docket No. 2002-NM-300-AD” in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from AvCraft Aerospace GmbH, P.O. Box 1103, D-82230 Wessling, Germany. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before