

**Other FAA AD Provisions**

(f) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, Standards Staff, FAA, ATTN: Albert J. Mercado, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements*: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 *et seq.*), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

**Related Information**

(g) Refer to Direction générale de l'aviation civile (DGAC) Airworthiness Directive No.: F-2006-028, dated February 1, 2006, approved by the European Aviation Safety Agency (EASA) on January 24, 2006; and EADS SOCATA TB Aircraft Mandatory Service Bulletin SB 70-135, ATA No. 55, dated December 2005, for related information.

**Material Incorporated by Reference**

(h) You must use EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-135, ATA No. 55, dated December 2005, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact EADS SOCATA, Direction des Services, 65921 Tarbes Cedex 9, France; telephone: 33 (0)5 62.41.73.00; fax: 33 (0)5 62.41.76.54; or SOCATA Aircraft, INC., North Perry Airport, 7501 Airport Road, Pembroke Pines, Florida 33023; telephone: (954) 893-1400; fax: (954) 964-4141.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri, on January 31, 2007.

**Margaret Kline,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-1878 Filed 2-7-07; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA-2006-26234 Directorate Identifier 2006-CE-64-AD; Amendment 39-14928; AD 2007-03-17]**

**RIN 2120-AA64**

**Airworthiness Directives; EADS SOCATA Model TBM 700 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as loose rivets on frames C18 BIS and C19, which could result in a reduced structural integrity of the tail area. We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective March 15, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of March 15, 2007.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:**

Albert J. Mercado, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090.

**SUPPLEMENTARY INFORMATION:****Streamlined Issuance of AD**

The FAA is implementing a new process for streamlining the issuance of ADs related to MCAI. The streamlined process will allow us to adopt MCAI safety requirements in a more efficient manner and will reduce safety risks to the public. This process continues to follow all FAA AD issuance processes to meet legal, economic, Administrative Procedure Act, and **Federal Register** requirements. We also continue to meet our technical decision-making responsibilities to identify and correct

unsafe conditions on U.S.-certificated products.

This AD references the MCAI and related service information that we considered in forming the engineering basis to correct the unsafe condition. The AD contains text copied from the MCAI and for this reason might not follow our plain language principles.

**Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on November 20, 2006 (71 FR 67084). That NPRM proposed to require an inspection of the rivets on frames C18 BIS and C19, and, if necessary, application of corrective actions. You may obtain further information by examining the MCAI in the AD docket.

**Comments**

We gave the public the opportunity to participate in developing this AD. We have considered the comment received.

**Comment Issue: Cost of Compliance**

EADS SOCATA comments that the inspection would take 0.5 work-hours. If necessary, rivets replacement would never take more than 5 work-hours and if parts are necessary, only rivets and shims are required, and their cost is negligible.

We will revise the work-hours estimate from 18 work-hours to 6 work-hours and the parts cost from \$2,300 to \$5 per EADS SOCATA's comments.

**Conclusion**

We reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD with the changes described previously. We determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

**Differences Between This AD and the MCAI or Service Information**

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies.

Any such differences are described in a separate paragraph of the AD, and take precedence over the actions copied from the MCAI.

### Costs of Compliance

We estimate that this AD will affect about 272 products of U.S. Registry. We also estimate that it will take 6 work-hours per product to comply with this AD. The average labor rate is \$80 per work-hour. Required parts will cost about \$5 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$131,920 or \$485 per product.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD Docket.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5227) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2007-03-17 EADS SOCATA Model TBM 700 Airplanes:** Amendment 39-14928; Docket No. FAA-2006-26234; Directorate Identifier 2006-CE-64-AD.

#### Effective Date

(a) This airworthiness directive (AD) becomes effective March 15, 2007.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to EADS SOCATA TBM 700 airplanes, all serial numbers, certificated in any category.

#### Reason

(d) The mandatory continuing airworthiness information (MCAI) states that loose rivets on frames C18 BIS and C19 were found, which, if not corrected, could result in a reduced structural integrity of the tail area.

#### Actions and Compliance

(e) Unless already done, within the next 100 hours time-in-service (TIS) or 12 months, whichever occurs later, after the effective date of this AD, and thereafter at intervals not

to exceed 100 hours TIS, accomplish a detailed inspection of the area and apply corrective actions as necessary by doing all the applicable actions in accordance with the accomplishment instructions of the EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-129, ATA No. 53, dated June 2005.

### FAA AD Differences

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

### Other FAA AD Provisions

(f) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Staff, FAA, ATTN: Albert J. Mercado, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et. seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

### Related Information

(g) Refer to Direction générale de l'aviation civile Airworthiness Directive No F-2005-132, dated August 3, 2005; and EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-129, ATA No. 53, dated June 2005, for related information.

### Material Incorporated by Reference

(h) You must use EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-129, ATA No. 53, dated June 2005, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact EADS SOCATA, Direction des Services, 65921 Tarbes Cedex 9, France; telephone: 33 (0)5 62.41.73.00; fax: 33 (0)5 62.41.76.54; or SOCATA Aircraft, INC., North Perry Airport, 7501 Airport Road, Pembroke Pines, Florida 33023; telephone: (954) 893-1400; fax: (954) 964-4141.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this

material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri, on January 30, 2007.

**Kim Smith,**

*Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-1877 Filed 2-7-07; 8:45 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2006-25192; Directorate Identifier 2006-NM-004-AD; Amendment 39-14930; AD 2007-03-19]

RIN 2120-AA64

#### **Airworthiness Directives; Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is superseding an existing airworthiness directive (AD), which applies to certain Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. That AD currently requires repetitive detailed and eddy current inspections of the main fittings of the main landing gears (MLG) to detect discrepancies, and related investigative/corrective actions if necessary. The AD also currently requires servicing the shock strut of the MLGs; inspecting the shock strut of the MLGs for nitrogen pressure, visible chrome dimension, and oil leakage; and servicing any discrepant strut. This new AD requires installing a new, improved MLG main fitting, which terminates the repetitive inspection and servicing requirements of the existing AD. This AD results from stress analyses that showed certain main fittings of the MLGs are susceptible to premature cracking, starting in the radius of the upper lug. We are issuing this AD to detect and correct premature cracking of the main fittings of the MLGs, which could result in failure of the fittings and consequent collapse of the MLGs during landing.

**DATES:** This AD becomes effective March 15, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of March 15, 2007.

On August 13, 2004 (69 FR 41421, July 9, 2004), the Director of the Federal

Register approved the incorporation by reference of Bombardier Alert Service Bulletin A601R-32-088, including Appendices A, B, and C, dated February 20, 2003.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for service information identified in this AD.

#### **FOR FURTHER INFORMATION CONTACT:**

Richard Beckwith, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, suite 410, Westbury, New York 11590; telephone (516) 228-7302; fax (516) 794-5531.

#### **SUPPLEMENTARY INFORMATION:**

##### **Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

##### **Discussion**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that supersedes AD 2004-14-16, amendment 39-13725 (69 FR 41421, July 9, 2004). The existing AD applies to certain Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. That NPRM was published in the **Federal Register** on June 27, 2006 (71 FR 36495). That NPRM proposed to continue to require installing a new, improved main landing gear (MLG) main fitting, which would terminate the repetitive inspection and servicing requirements of the existing AD.

##### **Comments**

We provided the public the opportunity to participate in the development of this AD. We have considered the comments that have been received on the NPRM.

##### **Request To Change Compliance Time to Cite Dates**

Bombardier notes that the proposed compliance time for the corrective

action is quite different from that of the parallel Canadian airworthiness directive. The parallel Canadian airworthiness directive specifies a fixed compliance date of December 31, 2008, for MLG main fittings that have part numbers 601R85001-81 and -82. Bombardier calculates that operators of U.S.-registered airplanes would have 12 months beyond that date to accomplish the proposed actions. Bombardier requests that we harmonize the compliance time in the NPRM with the compliance date in Canadian airworthiness directive CF-2003-09R1, dated September 21, 2005, which is the parallel Canadian airworthiness directive referred to in the NPRM. Bombardier points out that it worked with Messier-Dowty and Transport Canada Civil Aviation (TCCA) to consider carefully that date as it relates to fleet safety, MLG supplier capability/logistics, and the capacity of operators and overhaul facilities. Bombardier considers that the different compliance time will create confusion among U.S. operators and cause an unnecessary burden for all parties involved.

We partially agree. We agree that we should harmonize the compliance times in the NPRM with the compliance dates in the Canadian airworthiness directive. To that end, we developed the compliance time of "within 39 months after the effective date of this AD." This 39-month compliance time will give U.S. operators until May 2009 to comply with the AD. This amount of elapsed time is equivalent to that allowed by the Canadian airworthiness directive's compliance date of December 31, 2008. However, we find that this longer compliance time will not adversely affect the level of safety of the affected U.S.-registered airplanes. This issue has been coordinated with TCCA. No change has been made to the AD in this regard.

##### **Request To Incorporate by Reference (IBR) the Service Information**

The Modification and Replacement Parts Association (MARPA) requests that we either publish the relevant service information with the AD in the Docket Management System (DMS), or IBR it with the NPRM. MARPA states that the purpose of the IBR system is brevity, to keep from expanding the **Federal Register** needlessly by publishing documents already in the hands of the affected individuals. Traditionally, "affected individuals" have been aircraft owners and operators who are generally provided service information by the manufacturer. MARPA states that the group of affected individuals has expanded because