

transferred certain regulations under the Farm Service Agency (FSA) and the Grain Inspection, Packers and Stockyards Administration (GIPSA) to the Agricultural Marketing Service (AMS) to reflect changes in the organizational structure and delegated authorities within the United States Department of Agriculture (USDA). The final rule also made corresponding revisions to the regulations to reflect the organizational changes.

DATES: Effective October 23, 2019.

FOR FURTHER INFORMATION CONTACT:

Dawana J. Clark, Legislative and Regulatory Review Staff, Office of the Administrator, AMS, USDA; Telephone: (202) 720-7540, Fax: (202) 690-3767, or Email: Dawana.Clark@usda.gov.

SUPPLEMENTARY INFORMATION: In FR Doc. 2019-18201 appearing on page 45644 in the **Federal Register** of Friday, August 30, 2019, portions of amendatory instructions to sections in 9 CFR parts 201, 202, and 203 were either incorrect or incomplete.

In an instruction amending § 201.108-1, the section number read in error (§ 201.180-1). When amending paragraph (b) in § 202.2, we neglected to replace “GIPSA” with “AMS.” In amending § 203.7, we cited incorrect language to be changed. And finally, in the amendment to § 203.14, we incorrectly cited § 203.114.

This correcting amendment rule amends 9 CFR 201.108-1, 202.2, 203.7, and 203.14 as intended in the final rule published August 30, 2019, at 84 FR 45644.

List of Subjects

9 CFR Part 201

Confidential business information, Reporting and recordkeeping requirements, Stockyards, Surety bonds, Trade practices.

9 CFR Part 202

Administrative practice and procedure, Stockyards.

9 CFR Part 203

Reporting and recordkeeping requirements, Stockyards.

Accordingly, 9 CFR parts 201, 202, and 203 are amended by making the following correcting amendments:

PART 201—REGULATIONS UNDER THE PACKERS AND STOCKYARDS ACT

■ 1. The authority citation for part 201 continues to read as follows:

Authority: 7 U.S.C. 181–229c.

§ 201.108-1 [Amended]

■ 2. Amend § 201.108-1, in the introductory text, by removing both instances of the words “Packers and Stockyards Programs” and adding in their place “Packers and Stockyards Division”.

PART 202—RULES OF PRACTICE GOVERNING PROCEEDINGS AND UNDER THE PACKERS AND STOCKYARDS ACT

■ 3. The authority citation for part 202 continues to read as follows:

Authority: 7 U.S.C. 181–229c.

§ 202.2 [Amended]

■ 4. In § 202.2, amend paragraph (b) by removing the term “GIPSA” and adding in its place “AMS”.

PART 203—STATEMENTS OF GENERAL POLICY UNDER THE PACKERS AND STOCKYARDS ACT

■ 5. The authority citation for part 203 continues to read as follows:

Authority: 7 U.S.C. 2.22 and 2.81.

§ 203.7 [Amended]

■ 6. Amend § 203.7:

■ a. In the third sentence of paragraph (b)(2)(iii), by removing the word “area” and adding in its place “regional”.

■ b. In the fourth sentence of paragraph (b)(2)(iii), by removing the words “the Administration” and adding in their place the term “PSD”.

§ 203.14 [Amended]

■ 7. Amend § 203.14 in the note following paragraph 1 by removing the words “Grain Inspection, Packers and Stockyards Administration (Packers and Stockyards Programs)” and adding in their place “PSD”.

Dated: October 18, 2019.

Bruce Summers,

Administrator, Agricultural Marketing Service.

[FR Doc. 2019-23103 Filed 10-22-19; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2019-0524; Product Identifier 2019-NM-081-AD; Amendment 39-19738; AD 2019-19-02]

RIN 2120-AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain The Boeing Company Model 747-400 and 747-400F series airplanes. This AD was prompted by an evaluation that determined fatigue cracks could develop in the underwing longerons. This AD requires repetitive inspections of the underwing longerons and certain fuselage skins for any crack, and applicable on-condition actions. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 27, 2019.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of November 27, 2019.

ADDRESSES: For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740-5600; phone: 562-797-1717; internet: <https://www.myboeingfleet.com>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2019-0524.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2019-0524; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the regulatory evaluation, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-

30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Eric Lin, Aerospace Engineer, Airframe Section, FAA, Seattle ACO Branch, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3523; email: eric.lin@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain The Boeing Company Model 747-400 and 747-400F series airplanes. The NPRM published in the **Federal Register** on July 12, 2019 (84 FR 33189). The NPRM was prompted by an evaluation that determined fatigue cracks could develop in the underwing longerons. The NPRM proposed to require repetitive inspections of the underwing longerons and certain fuselage skins for any crack, and applicable on-condition actions.

The FAA is issuing this AD to address cracks in the underwing longerons,

which could result in fuel leakage into the pressurized fuselage and increase the risk of a fire, and to address cracks in the adjacent fuselage skin, which could result in rapid decompression. Either condition could adversely affect the structural integrity of the airplane.

Comments

The FAA gave the public the opportunity to participate in developing this final rule. The FAA has considered the comments received. Boeing and United Airlines indicated concurrence with the NPRM.

Conclusion

The FAA reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information Under 14 CFR Part 51

The FAA reviewed Boeing Alert Requirements Bulletin 747-53A2900 RB, dated April 11, 2019. This service information describes procedures for repetitive detailed inspections and ultrasonic inspections of the underwing longerons and the adjacent fuselage skin, and ultrasonic and surface high frequency eddy current (HFEC) inspections of certain fuselage skins, on the left and right sides of the airplane, for any crack, and applicable on-condition actions. On-condition actions include repair.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 20 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Detailed inspections of the underwing longerons.	2 work-hours × \$85 per hour = \$170 per inspection cycle.	\$0	\$170 per inspection cycle	\$3,400 per inspection cycle.
Ultrasonic and HFEC inspections of the adjacent fuselage skin.	9 work-hours × \$85 per hour = \$765 per inspection cycle.	0	\$765 per inspection cycle	\$15,300 per inspection cycle.
Ultrasonic inspections of the underwing longerons.	2 work-hours × \$85 per hour = \$170 per inspection cycle.	0	\$170 per inspection cycle	\$3,400 per inspection cycle.

The FAA has received no definitive data that would enable us to provide cost estimates for the on-condition actions specified in this AD.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an

unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes and associated appliances to the Director of the System Oversight Division.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and

responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2019–19–02 The Boeing Company: Amendment 39–19738; Docket No. FAA–2019–0524; Product Identifier 2019–NM–081–AD.

(a) Effective Date

This AD is effective November 27, 2019.

(b) Affected ADs

None.

(c) Applicability

This AD applies to The Boeing Company Model 747–400 and 747–400F series airplanes, certificated in any category, as identified in Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019.

(d) Subject

Air Transport Association (ATA) of America Code 53, Fuselage.

(e) Unsafe Condition

This AD was prompted by an evaluation that determined fatigue cracks could develop in the underwing longerons. The FAA is issuing this AD to address cracks in the underwing longerons, which could result in fuel leakage into the pressurized fuselage and increase the risk of a fire, and to address cracks in the adjacent fuselage skin, which could result in rapid decompression. Either condition could adversely affect the structural integrity of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

Except as specified by paragraph (h) of this AD: At the applicable times specified in the “Compliance” paragraph of Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019, do all applicable actions identified in, and in accordance with, the Accomplishment Instructions of Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019.

Note 1 to paragraph (g): Guidance for accomplishing the actions required by this AD can be found in Boeing Alert Service Bulletin 747–53A2900, dated April 11, 2019, which is referred to in Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019.

(h) Exceptions to Service Information Specifications

(1) For purposes of determining compliance with the requirements of this AD: Where Boeing Alert Requirements Bulletin

747–53A2900 RB, dated April 11, 2019, uses the phrase “the original issue date of Requirements Bulletin 747–53A2900 RB,” this AD requires using “the effective date of this AD.”

(2) Where Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019, specifies contacting Boeing for repair instructions: This AD requires doing the repair using a method approved in accordance with the procedures specified in paragraph (i) of this AD.

(i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by The Boeing Company Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO Branch, FAA, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

(j) Related Information

For more information about this AD, contact Eric Lin, Aerospace Engineer, Airframe Section, FAA, Seattle ACO Branch, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206–231–3523; email: eric.lin@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Requirements Bulletin 747–53A2900 RB, dated April 11, 2019.

(ii) [Reserved]

(3) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110–SK57, Seal Beach, CA 90740–5600; phone: 562–797–1717; internet: <https://www.myboeingfleet.com>.

(4) You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg.legal@nara.gov, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Des Moines, Washington, on September 19, 2019.

Suzanne Masterson,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2019–23074 Filed 10–22–19; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2019–0493; Product Identifier 2019–NM–043–AD; Amendment 39–19762; AD 2019–20–09]

RIN 2120–AA64

Airworthiness Directives; De Havilland Aircraft of Canada Limited (Type Certificate Previously Held by Bombardier, Inc.) Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2011–18–15, which applied to certain De Havilland Aircraft of Canada Limited Model DHC–8–400 series airplanes. AD 2011–18–15 required initial and repetitive torque checks of the bolt preload; detailed inspection of the barrel nuts and cradle for cracking, pitting, and corrosion if the bolt preload is correct; and replacement of certain hardware if necessary. This AD continues to require those actions. This AD also requires new inspections and replacement of certain hardware, which would terminate the repetitive torque checks and inspections; and removes airplanes from the applicability. This AD was prompted by in-service reports of cracked barrel nuts found at the front spar locations of the wing-to-fuselage attachment joints, and a loose washer in the barrel nut assembly. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 27, 2019.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of November 27, 2019.