Ontario K1A 0N5, Canada; telephone 888–663–3639; email *AD-CN@tc.gc.ca*; internet *https://tc.canada.ca/en/aviation*.

- (4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.
- (5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: https://www.archives.gov/federal-register/cfr/ibrlocations.html.

Issued on February 16, 2022.

## Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022-04145 Filed 2-25-22; 8:45 am]

BILLING CODE 4910-13-P

#### **DEPARTMENT OF TRANSPORTATION**

## **Federal Aviation Administration**

# 14 CFR Part 39

[Docket No. FAA-2021-1062; Project Identifier MCAI-2021-00886-T; Amendment 39-21957; AD 2022-05-06]

#### RIN 2120-AA64

# Airworthiness Directives; Airbus SAS Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all Airbus SAS Model A300 B2-1C, B2K-3C, B2-203, B4-2C, B4-103, and B4-203 airplanes. This AD was prompted by a determination that new or more restrictive airworthiness limitations related to pylon maintenance are necessary. This AD requires revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations for pylon maintenance, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective April 4,

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of April 4, 2022.

**ADDRESSES:** For EASA material incorporated by reference (IBR) in this AD, contact EASA, Konrad-Adenauer-

Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this IBR material on the EASA website at https://ad.easa.europa.eu. You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available in the AD docket at https://www.regulations.gov by searching for and locating Docket No. FAA-2021-1062.

# **Examining the AD Docket**

You may examine the AD docket at https://www.regulations.gov by searching for and locating Docket No. FAA-2021-1062; or in person at Docket Operations between 9 a.m. and 5 p.m.. Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, Large Aircraft Section, FAA, International Validation Branch, 2200 South 216th St., Des Moines, WA 98198; telephone 206–231–3225; email dan.rodina@faa.gov.

## SUPPLEMENTARY INFORMATION:

### **Background**

EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2021–0181, dated July 30, 2021 (EASA AD 2021–0181) (also referred to as the MCAI), to correct an unsafe condition for all Airbus SAS Model A300 B2–1C, B2K–3C, B2–203, B4–2C, B4–103, and B4–203 airplanes.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Airbus SAS Model A300 B2–1C, B2K–3C, B2–203, B4–2C, B4–103, and B4–203 airplanes. The NPRM published in the **Federal Register** on December 17, 2021 (86 FR 71592). The NPRM was prompted by a determination that new or more restrictive airworthiness limitations related to pylon maintenance are necessary. The NPRM proposed to require revising the existing maintenance or inspection program, as

applicable, to incorporate new or more restrictive airworthiness limitations for pylon maintenance, as specified in EASA AD 2021–0181.

The FAA is issuing this AD to address fatigue cracking, damage, and corrosion in principal structural elements; such fatigue cracking, damage, and corrosion could result in reduced structural integrity of the airplane. See the MCAI for additional background information.

# Discussion of Final Airworthiness Directive

### Comments

The FAA received no comments on the NPRM or on the determination of the cost to the public.

## Conclusion

The FAA reviewed the relevant data and determined that air safety requires adopting this AD as proposed. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products.

# **Related Service Information Under 1 CFR Part 51**

EASA AD 2021–0181 specifies new or more restrictive airworthiness limitations for pylon maintenance. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

# **Costs of Compliance**

The FAA estimates that this AD affects 1 airplane of U.S. registry. The FAA estimates the following costs to comply with this AD:

The FAA has determined that revising the existing maintenance or inspection program takes an average of 90 workhours per operator, although the agency recognizes that this number may vary from operator to operator. Since operators incorporate maintenance or inspection program changes for their affected fleet(s), the FAA has determined that a per-operator estimate is more accurate than a per-airplane estimate. Therefore, the agency estimates the average total cost per operator to be \$7,650 (90 work-hours × \$85 per work-hour).

# **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, part A, subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

# **Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## §39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2022–05–06 Airbus SAS:** Amendment 39–21957; Docket No. FAA–2021–1062; Project Identifier MCAI–2021–00886–T.

## (a) Effective Date

This airworthiness directive (AD) is effective April 4, 2022.

#### (b) Affected ADs

This AD affects AD 2018–19–17, Amendment 39–19417 (83 FR 48207, September 24, 2018) (AD 2018–19–17).

### (c) Applicability

This AD applies to all Airbus SAS Model A300 B2–1C, B2K–3C, B2–203, B4–2C, B4–103, and B4–203 airplanes, certificated in any category.

#### (d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

# (e) Unsafe Condition

This AD was prompted by a determination that new or more restrictive airworthiness limitations for pylon maintenance are necessary. The FAA is issuing this AD to address fatigue cracking, damage, and corrosion in principal structural elements; such fatigue cracking, damage, and corrosion could result in reduced structural integrity of the airplane.

# (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

### (g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2021–0181, dated July 30, 2021 (EASA AD 2021–0181).

# (h) Exceptions to EASA AD 2021-0181

- (1) Where EASA AD 2021–0181 refers to its effective date, this AD requires using the effective date of this AD.
- (2) The requirements specified in paragraphs (1) and (2) of EASA AD 2021–0181 do not apply to this AD.
- (3) Paragraph (3) of EASA AD 2021–0181 specifies revising "the approved AMP" within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after the effective date of this AD.
- (4) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2021–0181 is at the applicable "associated thresholds" as incorporated by the requirements of paragraph (3) of EASA AD 2021–0181, or within 90 days after the effective date of this AD, whichever occurs later.
- (5) The provisions specified in paragraph (4) of EASA AD 2021–0181 do not apply to this AD.
- (6) The "Remarks" section of EASA AD 2021–0181 does not apply to this AD.

# (i) Provisions for Alternative Actions and Intervals

After the existing maintenance or inspection program has been revised as required by paragraph (g) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the

"Ref. Publications" section of EASA AD 2021–0181.

### (j) Terminating Action for AD 2018-19-17

Accomplishing the actions required by this AD terminates the corresponding requirements of AD 2018–19–17, for the tasks identified in the service information referred to in EASA AD 2021–0181 only.

### (k) Additional AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Large Aircraft Section, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the Large Aircraft Section, International Validation Branch, send it to the attention of the person identified in paragraph (l) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, Large Aircraft Section, International Validation Branch, FAA; or EASA; or Airbus SAS's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) Required for Compliance (RC): Except as required by paragraph (k)(2) of this AD, if any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

## (l) Related Information

For more information about this AD, contact Dan Rodina, Aerospace Engineer, Large Aircraft Section, FAA, International Validation Branch, 2200 South 216th St., Des Moines, WA 98198; telephone 206–231–3225; email dan.rodina@faa.gov.

### (m) Material Incorporated by Reference

- (1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

- (i) European Union Aviation Safety Agency (EASA) AD 2021–0181, dated July 30, 2021.
  - (ii) [Reserved]
- (3) For EASA AD 2021–0181, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs*@easa.europa.eu; internet www.easa.europa.eu. You may find this EASA AD on the EASA website at https://ad.easa.europa.eu.
- (4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.
- (5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: https://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on February 17, 2022.

#### Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service. [FR Doc. 2022–04147 Filed 2–25–22; 8:45 am]

BILLING CODE 4910-13-P

#### DEPARTMENT OF TRANSPORTATION

#### **Federal Aviation Administration**

# 14 CFR Part 39

[Docket No. FAA-2020-0345; Product Identifier 2019-NM-154-AD; Amendment 39-21951; AD 2022-04-09]

RIN 2120-AA64

Airworthiness Directives; AVOX System Inc. (Formerly Scott Aviation) Oxygen Cylinder and Valve Assemblies and Oxygen Valve Assemblies

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain AVOX System Inc. (formerly Scott Aviation) oxygen cylinder and valve assemblies, and oxygen valve assemblies, installed on but not limited to various transport airplanes. This AD was prompted by reports of cylinder and valve assemblies having oxygen leakage from the valve assembly vent hole, caused by the absence of a guide that maintains appropriate spacing between certain parts. This AD requires an inspection of the oxygen valve assemblies, and oxygen cylinder and valve assemblies, to determine the serial number of the valve, cylinder, and entire assembly. For assemblies and parts with certain serial numbers, this

AD requires a detailed inspection for correct spacing of the gap between the bottom of the packing retainer and top of the valve body on the assemblies, and replacement of assemblies having unacceptable gaps. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective April 4, 2022.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of April 4, 2022.

**ADDRESSES:** For service information identified in this final rule, contact AVOX Systems Inc., 225 Erie Street, Lancaster, NY 14086; telephone 716-683-5100; internet https:// www.safranaerosystems.com. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at https:// www.regulations.gov by searching for and locating Docket No. FAA-2020-0345.

## **Examining the AD Docket**

You may examine the AD docket at https://www.regulations.gov by searching for and locating Docket No. FAA–2020–0345; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

## FOR FURTHER INFORMATION CONTACT:

Elizabeth Dowling, Aerospace Engineer, Mechanical Systems and Administrative Services Section, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7300; email *9-avs-nyaco-cos@faa.gov*.

## SUPPLEMENTARY INFORMATION:

## **Background**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain AVOX System Inc. (formerly Scott Aviation) oxygen cylinder and valve assemblies, and oxygen valve assemblies, installed on but not limited to various transport airplanes. The NPRM published in the **Federal Register** on May 1, 2020 (85 FR 25353). The NPRM was prompted by

reports of cylinder and valve assemblies having oxygen leakage from the valve assembly vent hole, caused by the absence of a guide that maintains appropriate spacing between certain parts. In the NPRM, the FAA proposed to require an inspection of the oxygen valve assemblies, and oxygen cylinder and valve assemblies, to determine the serial number of the valve, cylinder, and entire assembly. For assemblies and parts with certain serial numbers, the NPRM proposed to require a detailed inspection for correct spacing of the gap between the bottom of the packing retainer and top of the valve body on the assemblies, and replacement of assemblies having unacceptable gaps (removing affected assemblies and installing serviceable assemblies). The NPRM also proposed to require reporting and returning of affected parts to the manufacturer. The FAA is issuing this AD to address oxygen leakage from the cylinder, which could result in decreased or insufficient oxygen supply during a depressurization event; and heating or flow friction, which could cause an ignition event in the valve assembly.

# Discussion of Final Airworthiness Directive

### Comments

The FAA received comments from one commenter, Air Line Pilots Association, International (ALPA), who supported the NPRM without change.

The FAA received additional comments from five commenters, including American Airlines (AAL), Delta Air Lines (DAL), FedEx Express (FedEx), United Airlines (UAL), and an individual. The following presents the comments received on the NPRM and the FAA's response to each comment.

## Request To Revise Applicability

AAL and UAL suggested revising the applicability statement to include more aircraft manufacturers and models. AAL suggested adding all airplane models that affected assemblies could be installed on, in particular, Boeing Model 737-NG (Next Generation models are 737-600, -700, -700C, -800, -900, and -900ER series), 737-MAX, 777-200, and 777-300 series airplanes. UAL also suggested adding Model 737-NG airplanes. AAL stated that the applicability statement as proposed in the NPRM could mislead operators into believing that the AD would apply only to the airplanes identified in paragraphs (c)(1) through (12) of the AD. UAL believed the suggested change will be beneficial and assist operators in determining if their fleets are affected.