

(1) Bell Model 206B, serial number (S/N) 004 through 4690 inclusive, including helicopters converted from Model 206A; and

**Note 1 to paragraph (a)(1):** Helicopters with a 206B3 designation are Model 206B helicopters.

(2) Bell Model 206L, S/N 45001 through 45153 inclusive, and 46601 through 46617 inclusive.

#### (b) Unsafe Condition

This AD defines the unsafe condition as a third stage turbine vibration. This condition could result in turbine failure, engine power loss, and subsequent loss of control of the helicopter.

#### (c) Affected ADs

This AD supersedes AD 2013–20–13, Amendment 39–17619 (78 FR 66252, November 5, 2013).

#### (d) Comments Due Date

The FAA must receive comments by April 26, 2021.

#### (e) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

#### (f) Required Actions

Within 25 hours time-in-service:

(1) For Bell Model 206B helicopters:

(i) Revise the existing Rotorcraft Flight Manual (RFM) for your helicopter by inserting Section 1, Operating Limitations, page 1–2A, of Bell Model 206B RFM BHT–206B–FM–1, Revision B–54, dated May 30, 2018 (BHT–206B–FM–1) or Section 1, Limitations, page 1–5, of Bell Model 206B3 RFM BHT–206B3–FM–1, Revision 17, dated May 30, 2018 (BHT–206B3–FM–1), as applicable to your helicopter. Inserting a different document with “Steady-state operation” information identical to page 1–2A of BHT–206B–FM–1 or page 1–5 of BHT–206B3–FM–1, as applicable to your helicopter, is acceptable for compliance with the requirements of this paragraph.

(ii) Revise the existing RFM for your helicopter by inserting Section 2, Normal Procedures, page 2–8 of BHT–206B–FM–1 or Section 2, Normal Procedures, page 2–10 of BHT–206B3–FM–1, as applicable to your helicopter. Inserting a different document with “Continuous Operation” information identical to page 2–8 of BHT–206B–FM–1 or page 2–10 of BHT–206B3–FM–1, as applicable to your helicopter, is acceptable for compliance with the requirements of this paragraph.

(iii) Remove placard part number (P/N) 230–075–213–121, if installed.

(iv) Install placard P/N 230–075–213–129 or placard P/N 230–075–213–131 on the instrument panel directly below the dual tachometer.

(2) For Bell Model 206L helicopters:

(i) Revise the existing RFM for your helicopter by inserting Section 1, Operating Limitations, page 1–4B, of Bell Model 206L RFM BHT–206L–FM–1, Revision 31, dated May 30, 2018 (BHT–206L–FM–1). Inserting a different document with “Steady-state operation” information identical to page 1–

4B of BHT–206L–FM–1 is acceptable for compliance with the requirements of this paragraph.

(ii) Revise the existing RFM for your helicopter by inserting Section 2, Normal Procedures, page 2–10 of BHT–206L–FM–1. Inserting a different document with “Continuous Operation” information identical to page 2–10 of BHT–206L–FM–1 is acceptable for compliance with the requirements of this paragraph.

(iii) Remove placard P/N 230–075–213–123, if installed.

(iv) Install placard P/N 230–075–213–129 or placard P/N 230–075–213–131 on the instrument panel below the dual tachometer.

#### (g) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, may approve AMOCs for this AD. Send your proposal to: Michael Hughlett, Aviation Safety Engineer, General Aviation & Rotorcraft Section, International Validation Branch, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5889; email 9-AVS-AIR-730-AMOC@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, the FAA suggests that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

#### (h) Additional Information

(1) Bell Alert Service Bulletin (ASB) 206–07–115, Revision D, and ASB 206L–07–146, Revision C, each dated July 9, 2018, which are not incorporated by reference, contain additional information about the subject of this AD. For a copy of this service information, contact Bell Textron Canada Limited, 12,800 Rue de l’Avenir, Mirabel, Quebec J7J 1R4; telephone 450–437–2862 or 800–363–8023; fax 450–433–0272; or at <https://www.bellcustomer.com>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177.

(2) The subject of this AD is addressed in Transport Canada AD No. CF–2018–23, dated August 22, 2018. You may view the Transport Canada AD on the internet in the AD Docket at <https://www.regulations.gov>.

#### (i) Subject

Joint Aircraft Service Component (JASC) Code: 7250, Turbine Section.

Issued on January 27, 2021.

#### Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–05093 Filed 3–10–21; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2020–1171; Product Identifier 2017–SW–124–AD]

RIN 2120–AA64

#### Airworthiness Directives; Airbus Helicopters Deutschland GmbH

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for all Airbus Helicopters Deutschland GmbH Model MBB–BK 117 C–2 and Model MBB–BK 117 D–2 helicopters. This proposed AD was prompted by a determination that a life limit for the adapter forward (FWD) of the outboard load system, repetitive inspections of other components of that system, and for certain helicopters, a modification of the outboard load system, are necessary to address the unsafe condition. This proposed AD would require a modification of the outboard load system for certain helicopters, repetitive inspections of the outboard load system and its components for any defect (including cracking, damage, corrosion, and incorrect installation) and applicable corrective actions, and implementation of a new life limit for the FWD adapter, as specified in a European Aviation Safety Agency (EASA) AD, which is proposed for incorporation by reference (IBR). The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by April 26, 2021.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For material that is proposed for IBR in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000;

email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet [www.easa.europa.eu](http://www.easa.europa.eu). You may find this material on the EASA website at <https://ad.easa.europa.eu>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. It is also available in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1171.

#### Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1171; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

#### FOR FURTHER INFORMATION CONTACT:

Kathleen Arrigotti, Aviation Safety Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3218; email: [kathleen.arrigotti@faa.gov](mailto:kathleen.arrigotti@faa.gov).

#### SUPPLEMENTARY INFORMATION:

#### Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2020-1171; Product Identifier 2017-SW-124-AD" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to <https://www.regulations.gov>, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this proposal.

#### Confidential Business Information

CBI is commercial or financial information that is both customarily and

actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Kathleen Arrigotti, Aviation Safety Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3218; email: [kathleen.arrigotti@faa.gov](mailto:kathleen.arrigotti@faa.gov). Any commentary that the FAA receives that is not specifically designated as CBI will be placed in the public docket for this rulemaking.

#### Discussion

The EASA (now European Union Aviation Safety Agency), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2017-0177, dated September 14, 2017 (EASA AD 2017-0177) (also referred to as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for all Airbus Helicopters Deutschland GmbH Model MBB-BK 117 C-2 helicopters, except the Model C-2e variant, and all Model MBB-BK 117 D-2 helicopters.

Airbus Helicopters Deutschland GmbH Model MBB-BK 117 C-2e variant helicopters are not a unique model on the U.S. type certificate but are considered a configuration of the Model MBB-BK117 C-2. The U.S. type certificate data sheet explains that the FAA determined that the type design changes involved did not rise to the level that required an FAA amended type certificate. However, the FAA does recognize that helicopters with these type design changes exist, therefore the designation Model MBB-BK117 C-2(e) is used, starting from Serial Number 9601. The Model MBB-BK117 C-2(e) is a visual flight rules only configuration of the Model MBB-BK117 C-2 utilizing a Garmin 500H flight display system.

This proposed AD was prompted by a determination that a life limit for the adapter FWD of the outboard load system, repetitive inspections of other components of that system, and for certain helicopters, a modification of the

outboard load system, are necessary to address the unsafe condition. The FAA is proposing this AD to address detachment of an external load or person from the helicopter hoist, resulting in personal injury, or injury to persons on the ground. See the MCAI for additional background information.

#### Related Service Information Under 1 CFR Part 51

EASA AD 2017-0177 describes procedures for modification of the outboard load system for certain Model MBB-BK 117 C-2 helicopters, repetitive inspections of the outboard load system and its components for any defect (including cracking, damage, corrosion, and incorrect installation) and corrective actions, and implementation of a new life limit for the FWD adapter (*i.e.*, repetitive replacements). The corrective actions include replacement of any defective component with a serviceable part.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is proposing this AD after evaluating all the relevant information and determining the unsafe condition described previously is likely to exist or develop in other products of the same type design.

#### Proposed AD Requirements

This proposed AD would require accomplishing the actions specified in EASA AD 2017-0177, described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this proposed AD.

#### Explanation of Required Compliance Information

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA initially worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has since coordinated with other manufacturers and civil aviation authorities (CAAs) to

use this process. As a result, EASA AD 2017–0177 will be incorporated by reference in the FAA final rule. This proposed AD would, therefore, require compliance with EASA AD 2017–0177 in its entirety, through that incorporation, except for any differences identified as exceptions in the regulatory text of this proposed AD. Using common terms that are the same as the heading of a particular section in the EASA AD does not mean that

operators need comply only with that section. For example, where the AD requirement refers to “all required actions and compliance times,” compliance with this AD requirement is not limited to the section titled “Required Action(s) and Compliance Time(s)” in the EASA AD. Service information specified in EASA AD 2017–0177 that is required for compliance with EASA AD 2017–0177 will be available on the internet at

<https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–1171 after the FAA final rule is published.

#### Costs of Compliance

The FAA estimates that this proposed AD affects 175 helicopters of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

#### ESTIMATED COSTS FOR REQUIRED ACTIONS \*

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
2 work-hours × \$85 per hour = \$170 .....	\$1,306	\$1,476	\$258,300

\* The FAA has received no definitive data that would enable providing cost estimates for the modification specified in this proposed AD.

The FAA estimates the following costs to do any necessary on-condition actions that would be required based on

the results of any required actions. The FAA has no way of determining the

number of helicopters that might need these on-condition actions:

#### ESTIMATED COSTS OF ON-CONDITION ACTION \*

Labor cost	Parts cost	Cost per product
2 work-hours × \$85 per hour = \$170 .....	\$(*)	\$170(*)

\* The FAA has not received any definitive data regarding the parts cost, therefore this table does not include estimated costs for parts.

According to the manufacturer, some or all of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected operators. The FAA does not control warranty coverage for affected operators. As a result, the FAA has included all known costs in the cost estimate.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

The FAA determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**Airbus Helicopters Deutschland GmbH:**  
Docket No. FAA–2020–1171; Product Identifier 2017–SW–124–AD.

#### (a) Comments Due Date

The FAA must receive comments by April 26, 2021.

#### (b) Affected Airworthiness Directives (ADs)

None.

#### (c) Applicability

This AD applies to Airbus Helicopters Deutschland GmbH Model MBB–BK 117 C–2 and Model MBB–BK 117 D–2 helicopters, certificated in any category, all manufacturer serial numbers, except the Model MBB–BK117 C–2(e) configuration.

**Note 1 to paragraph (c):** Model MBB–BK117 C–2 helicopters utilizing a Garmin 500H flight display system are designated by EASA as Model MBB–BK117 C–2e variants of the Model BK 117 C–2 helicopters, and by the FAA as a Model MBB–BK117 C–2(e) configuration.

**(d) Subject**

Joint Aircraft System Component (JASC)  
Code 2500, Cabin Equipment/Furnishings.

**(e) Reason**

This AD was prompted by a determination that a life limit for the adapter forward (FWD) of the outboard load system, repetitive inspections of other components of that system, and for certain helicopters, a modification of the outboard load system are necessary to address the unsafe condition. The FAA is issuing this AD to address detachment of an external load or person from the helicopter hoist, which could result in personal injury, or injury to persons on the ground.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Requirements**

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Aviation Safety Agency (now European Union Aviation Safety Agency) (EASA) AD 2017–0177, dated September 14, 2017 (EASA AD 2017–0177).

**(h) Exceptions to EASA AD 2017–0177**

(1) Where EASA AD 2017–0177 refers to its effective date, this AD requires using the effective date of this AD.

(2) The “Remarks” section of EASA AD 2017–0177 does not apply to this AD.

(3) Where the service information referenced in EASA AD 2017–0177 specifies contacting the applicable manufacturer of the dedicated equipment for a definition of a cycle and recalculation to hoist cycles, this AD does not require contacting the manufacturer for a definition of a cycle and recalculation to hoist cycles.

(4) Where paragraph (3) of EASA AD 2017–0177 specifies to do “applicable corrective actions,” for this AD, if there are any defective components, replace all defective components with serviceable components in accordance with FAA-approved procedures. For the purposes of this AD, a defect may be indicated by cracking, damage, corrosion, or incorrect installation.

(5) Although the service information referenced in EASA AD 2017–0177 specifies to discard certain parts, this AD requires removing those parts from service instead.

(6) Where the service information referenced in EASA AD 2017–0177 refers to flight hours (FH), this AD requires using hours time-in-service.

(7) Paragraph (9) of EASA AD 2017–0177 does not apply to this AD.

**(i) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Strategic Policy Rotorcraft Section, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly

to the Manager of the Strategic Policy Rotorcraft Section, send it to: Manager, Strategic Policy Rotorcraft Section, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5110; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local Flight Standards District Office/Certificate Holding District Office.

**(j) Related Information**

(1) For EASA AD 2017–0177, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet [www.easa.europa.eu](http://www.easa.europa.eu). You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110. This material may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–1171.

(2) For more information about this AD, contact Kathleen Arrigotti, Aviation Safety Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206–231–3218; email: [kathleen.arrigotti@faa.gov](mailto:kathleen.arrigotti@faa.gov).

Issued on January 5, 2021.

**Gaetano A. Sciortino,**

*Deputy Director for Strategic Initiatives,  
Compliance & Airworthiness Division,  
Aircraft Certification Service.*

[FR Doc. 2021–05086 Filed 3–10–21; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA–2021–0142; Project Identifier MCAI–2020–01400–T]**

**RIN 2120–AA64**

**Airworthiness Directives; Airbus SAS Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for certain Airbus SAS Model A350–941 and –1041 airplanes. This proposed AD was prompted by a report of in-production findings of missing or incorrect application of the lightning strike edge glow sealant protection at specific locations in the wing tanks. This proposed AD would require an

inspection for missing or incorrect application of the lightning strike edge glow sealant protection at certain locations in the wing tanks, and corrective action, as specified in a European Union Aviation Safety Agency (EASA) AD, which is proposed for incorporation by reference. The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by April 26, 2021.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** 202–493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For EASA material that will be incorporated by reference (IBR) in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet [www.easa.europa.eu](http://www.easa.europa.eu). You may find this IBR material on the EASA website at <https://ad.easa.europa.eu>. You may view this IBR material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0142.

**Examining the AD Docket**

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0142; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, any comments received, and other information. The street address for Docket Operations is listed above.

**FOR FURTHER INFORMATION CONTACT:** Kathleen Arrigotti, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3218; email [kathleen.arrigotti@faa.gov](mailto:kathleen.arrigotti@faa.gov).