

**(b) Unsafe Condition**

This AD defines the unsafe condition as a tooth rupture in the MGB. This condition could result in failure of the MGB, loss of power to the main rotor, and subsequent loss of control of the helicopter.

**(c) Comments Due Date**

We must receive comments by September 3, 2013.

**(d) Compliance**

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

**(e) Required Actions**

(1) Within 100 hours time-in-service (TIS), and thereafter at intervals not to exceed 100 hours or 12 months, whichever occurs first, take an oil sample in accordance with the Accomplishment Instructions, Part 1, of Eurocopter Alert Service Bulletin EC 135–63A–012, Revision 5, dated September 6, 2011 (ASB EC135–63A–012).

(2) Within 25 hours TIS after taking the oil sample in paragraph (e)(1), analyze the oil sample in accordance with the Accomplishment Instructions, Part 2.A. through Part 2.C. of ASB EC135–63A–012, except that you are not required to contact Eurocopter.

(i) If the analysis indicates Stage II as specified by the Accomplishment Instructions, Part 2.B., of ASB EC135–63A–012, within 25 hours TIS, remove and inspect the oil filter element for a chip, defined as any solid piece of metal but not metallic fuzz or fine particles.

(A) If there are no chips, clean the oil filter element and chip detector, inspect the drive stage toothing, perform a ground run, and inspect for leaking oil in accordance with the Accomplishment Instructions, Part 4.A through 4.G, of ASB EC135–63A–012. Change the oil.

(B) If there is a chip, replace the MGB with an airworthy MGB before further flight.

(ii) If the analysis indicates Stage III as specified by the Accomplishment Instructions, Part 2.B., of ASB EC135–63A–012 and if the water content is between 0.1 and 0.5 percent, within 10 hours TIS, remove and inspect the oil filter element for a chip.

(A) If there are no chips, clean the oil filter element and chip detector, inspect the drive stage toothing, perform a ground run, and inspect for leaking oil in accordance with the Accomplishment Instructions, Part 4.A through 4.G, of ASB EC135–65A–012. Change the oil.

(B) If there is a chip, replace the MGB with an airworthy MGB before further flight.

(3) Before the MGB has accumulated 300 hours TIS, determine whether two or more chip indications have occurred. If two or more chip indications have occurred, inspect the drive stage toothing, perform a ground run, and inspect for leaking oil in accordance with the Accomplishment Instructions, Part 4.A through 4.G, of ASB EC135–65A–012.

(4) Any time there is a chip indication, remove and inspect the chip detector for deposits (fine particles or metallic fuzz) or chips, and remove and inspect the oil filter element for a chip.

(i) If there are no chips and a minimal amount of particles or metallic fuzz, corresponding to Figure 5, Stage A of ASB EC135–65A–012, clean the chip detector and the oil filter element and enter the chip indication on the MGB log card before further flight.

(ii) If there are no chips and some particles or metallic fuzz, corresponding to Figure 5, Stage B of ASB EC135–65A–012, clean the chip detector and the oil filter element and enter the chip indication on the MGB log card before further flight, and within 10 hours TIS inspect the drive stage toothing, perform a ground run, and inspect for leaking oil in accordance with the Accomplishment Instructions, Part 4.A through 4.G, of ASB EC135–65A–012. Perform a ground run for 15 minutes at the flight-idle power setting, and then re-inspect the chip detector for a chip, particles, and metallic fuzz.

(iii) If there is a chip, replace the MGB with an airworthy MGB.

**(f) Alternative Methods of Compliance (AMOC)**

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Chinh Vuong, Aerospace Engineer, Safety Management Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, TX 76137; telephone (817) 222–5110; email [Chinh.Vuong@faa.gov](mailto:Chinh.Vuong@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

**(g) Additional Information**

The subject of this AD is addressed in the European Aviation Safety Agency (EASA) AD No. 2009–0106R1, dated November 3, 2011. You may view the EASA AD at <http://www.regulations.gov> in Docket No. FAA–2013–0554.

**(h) Subject**

Joint Aircraft Service Component (JASC)  
Code: 6320, Main Rotor Gearbox.

Issued in Fort Worth, Texas, on June 18, 2013.

**Kim Smith,**

*Directorate Manager, Rotorcraft Directorate,  
Aircraft Certification Service.*

[FR Doc. 2013–15958 Filed 7–2–13; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2012–0886; Directorate Identifier 2008–SW–067–AD]

**RIN 2120–AA64**

**Airworthiness Directives; Agusta S.p.A. Helicopters**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Supplemental notice of proposed rulemaking (SNPRM); reopening of comment period.

**SUMMARY:** We are revising an earlier notice of proposed rulemaking (NPRM) that proposed to issue an airworthiness directive (AD) for Agusta S.p.A. (Agusta) Model AB139 and AW139 helicopters with a certain wire strike protection system (WSPS) top cable cutter assembly installed. The existing NPRM proposes to require reworking or replacing the top cable cutter assembly to increase clearance between the WSPS and the main rotor (M/R) blades. The NPRM was prompted by a report of in-flight contact between the top cable-cutter assembly and two M/R blades. This action revises the proposals in the NPRM by requiring that the reworked or replaced part be marked with “BT 139–126 Rev./” or “FAA” at the end of the part number to reflect the field modification. Since these actions impose an additional burden over that proposed in the NPRM, we are reopening the comment period to allow the public the chance to comment on these proposed changes.

**DATES:** We must receive comments on this SNPRM by September 3, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Docket:* Go to <http://www.regulations.gov>. Follow the online instructions for sending your comments electronically.
- *Fax:* 202–493–2251.
- *Mail:* Send comments to the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590–0001.
- *Hand Delivery:* Deliver to the “Mail” address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the

Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the economic evaluation, any comments received, and other information. The street address for the Docket Operations Office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt. For service information identified in this proposed AD, contact Agusta, Via Giovanni Agusta, 520 21017 Cascina Costa di Samarate (VA), Italy, telephone 39 0331-229111, fax 39 0331-229605/222595, or at [http://customersupport.agusta.com/technical\\_advice.php](http://customersupport.agusta.com/technical_advice.php). You may review the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

**FOR FURTHER INFORMATION CONTACT:** Sharon Miles, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, TX 76137; telephone (817) 222-5110; email [sharon.y.miles@faa.gov](mailto:sharon.y.miles@faa.gov).

#### **SUPPLEMENTARY INFORMATION:**

##### **Comments Invited**

We invite you to participate in this rulemaking by submitting written comments, data, or views. We also invite comments relating to the economic, environmental, energy, or federalism impacts that might result from adopting the proposals in this document. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit only one time.

We will file in the docket all comments that we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning this proposed rulemaking. Before acting on this proposal, we will consider all comments we receive on or before the closing date for comments. We will consider comments filed after the comment period has closed if it is possible to do so without incurring expense or delay. We may change this proposal in light of the comments we receive.

##### **Discussion**

On August 16, 2012, we issued an NPRM to amend 14 CFR part 39 to

include an AD that would apply to Agusta Model AB139 and AW139 helicopters with a certain WSPS top cable cutter assembly installed. The NPRM was published in the **Federal Register** on August 29, 2012 (77 FR 52270). The NPRM was prompted by a report of in-flight contact between the top cable-cutter assembly and two M/R blades. Based on this report, the European Aviation Safety Agency (EASA), which is the technical Agent for the member States of the European Union, issued EASA AD No. 2008-0148, dated August 5, 2008. The NPRM proposed to require reworking or replacing the top cable cutter assembly to increase clearance between the WSPS and the M/R blades, which is the same corrective action required by EASA AD No. 2008-0148.

##### **Actions Since Previous NPRM Was Issued**

Since we issued the previous NPRM (77 FR 52270, August 29, 2012), we determined that the required actions should include re-identifying the reworked or replaced part by marking "BT 139-126 Rev.1" or "FAA" at the end of the part number to reflect the field modification. Because this proposed change expands the scope of the NPRM, we are reopening the comment period to provide additional opportunity for public comment.

##### **Comments**

We gave the public the opportunity to comment on the previous NPRM (77 FR 52270, August 29, 2012), but we did not receive any comments.

##### **FAA's Determination**

We are proposing this SNPRM because we evaluated all known relevant information and determined that an unsafe condition is likely to exist or develop on other products of these same type designs. Certain changes described above expand the scope of the original NPRM. As a result, we have determined that it is necessary to reopen the comment period to provide additional opportunity for the public to comment on this SNPRM.

##### **Related Service Information**

We reviewed Agusta Bollettino Tecnico No. 139-126, dated June 20, 2008 (BT), which applies to Model AB139 and AW139 helicopters with certain serial-numbered WSPSs. The BT specifies, within 200 flight hours, reworking the top cable cutter assembly and marking it with "BT 139-126 Rev.1" in a visible and permanent manner. EASA classified this BT as mandatory and issued AD No. 2008-

0148 to ensure the continued airworthiness of these helicopters.

##### **Proposed Requirements of the SNPRM**

This proposed SNPRM would require:

- Within 200 hours time-in-service, removing the WSPS upper installation, P/N 4G9540A00111, including cutter assembly, P/N 423-83001-1.
- Before installing a WSPS upper installation, P/N 4G9540A00111, either reworking the top cable cutter assembly, P/N 423-83001-1, and re-identifying the part by adding "BT 139-126 Rev.1" or "FAA" to the end of the part number, or replacing the top cable cutter assembly, P/N 423-83001-1, with an airworthy top cable cutter assembly that has been reworked and re-identified.

##### **Costs of Compliance**

We estimate that this AD would affect about 39 helicopters of U.S. registry. We also estimate that it would take about 3 work-hours per helicopter to rework the top cable cutter assembly and to add "BT 139-126 Rev.1" or "FAA" at the end of the part number, 1 work-hour to replace the top cable cutter assembly, and 1 work-hour to remove the WSPS upper installation. The average labor rate is \$85 per work-hour. Required parts would cost about \$9,000 per helicopter to replace the cutter. Based on these figures, we estimate the cost of this AD on U.S. operators would be \$255 per helicopter to rework the top cable cutter assembly, \$9,085 per helicopter to replace the top cable cutter assembly, and \$85 per helicopter to remove the WSPS.

##### **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared an economic evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by Reference, Safety.

## The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Agusta S.p.A. Helicopters:** Docket No. FAA–2012–0886; Directorate Identifier 2008–SW–067–AD.

### (a) Applicability

This AD applies to Agusta Model AB139 and AW139 helicopters, with a wire strike protection system (WSPS) top cable cutter assembly, part number (P/N) 423–83001–1, installed, which is part of the WSPS, P/N 4G9540F00211 or P/N 4G9540F00311, certificated in any category.

### (b) Unsafe Condition

This AD defines the unsafe condition as in-flight contact between the top cable cutter assembly and main rotor (M/R) blades. This

condition could result in damage to the M/R blades and subsequent loss of helicopter control.

### (c) Comments Due Date

We must receive comments by September 3, 2013.

### (d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

### (e) Required Actions

(1) Within 200 hours time-in-service, remove the WSPS upper installation, P/N 4G9540A00111, including top cable cutter assembly, P/N 423–83001–1.

(2) Before installing a WSPS upper installation, P/N 4G9540A00111, either:

(i) Rework the top cable cutter assembly, P/N 423–83001–1, in accordance with the Compliance Instructions, paragraph 3.1 through 3.5, and Figure 1 of Agusta Bolletino Tecnico No. 139–126, dated June 20, 2008. Re-identify the top cable cutter assembly in a visible and permanent way by adding "BT 139–126 Rev.1" or "FAA" at the end of the part number; or

(ii) Replace the top cable cutter assembly, P/N 423–83001–1, with an airworthy top cable cutter assembly that has been reworked and re-identified in accordance with paragraph (e)(2)(i) of this AD.

(3) Do not install a top cable cutter assembly, P/N 423–83001–1, on any helicopter unless it has been reworked and re-identified in accordance with paragraph (e)(2)(i) of this AD.

### (f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Sharon Miles, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, TX 76137; telephone (817) 222–5110; email [sharon.y.miles@faa.gov](mailto:sharon.y.miles@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

### (g) Additional Information

The subject of this AD is addressed in European Aviation Safety Agency (EASA) AD No. 2008–0148, dated August 5, 2008. You may view the EASA AD at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA–2012–0886.

### (h) Subject

Joint Aircraft Service Component (JASC) Code: 5320: Fuselage Miscellaneous Structure.

Issued in Fort Worth, Texas, on June 18, 2013.

**Kim Smith,**

*Directorate Manager, Rotorcraft Directorate, Aircraft Certification Service.*

[FR Doc. 2013–15951 Filed 7–2–13; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

## Federal Aviation Administration

## 14 CFR Part 39

**[Docket No. FAA–2013–0467; Directorate Identifier 2013–NM–023–AD]**

**RIN 2120–AA64**

## Airworthiness Directives; Airbus Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for all Airbus Model A318, A319, A320, and A321 series airplanes. This proposed AD was prompted by reports of certain sliding windows that were difficult to operate after landing. This proposed AD would require a detailed inspection to identify part numbers of sliding windows and sliding window seals, and modification if necessary. This proposed AD also includes an optional replacement. We are proposing this AD to detect and correct incorrect seals, which could lead to the functional loss of the sliding window as an exit, possibly preventing the flightcrew from safely evacuating the airplane during an emergency.

**DATES:** We must receive comments on this proposed AD by August 19, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** (202) 493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For Airbus service information identified in this proposed AD, contact Airbus, Airworthiness Office—EIAS, 1