

individual assistance (Assistance to Individuals and Households Program).

(2) If the President declares a Major Disaster limited to public assistance only, a private nonprofit facility which provides non-critical services under guidelines of the Federal Emergency Management Agency (FEMA) must first apply to SBA for disaster loan assistance for such non-critical services before it could seek grant assistance from FEMA.

* * * * *

3. Amend § 123.4 by revising the fourth sentence to read as follows:

§ 123.4 What is a disaster area and why is it important?

* * * In major disasters, economic injury disaster loans may be made for victims in contiguous counties or other political subdivisions, provided, however, that with respect to major disasters which authorize public assistance only, SBA shall not make economic injury disaster loans in counties contiguous to the disaster area.

* * *

4. Amend § 123.101 by revising paragraph (c) to read as follows:

§ 123.101 When am I not eligible for a home disaster loan?

* * * * *

(c) Your damaged property can be repaired or replaced with the proceeds of insurance, gifts or other compensation, including condemnation awards (with one exception), these amounts must either be deducted from the amount of the claimed losses or, if received after SBA has approved and disbursed a loan, must be paid to SBA as principal payments on your loan. You must notify SBA of any such recoveries collected after receiving an SBA disaster loan. The one exception applies to amounts received under the Individuals and Household Program of the Federal Emergency Management Agency solely to meet an emergency need pending processing of an SBA loan. In such an event, you must repay the financial assistance with SBA loan proceeds if it was used for purposes also eligible for an SBA loan;

* * * * *

5. Amend § 123.501 by revising paragraph (a), removing “and” at the end of paragraph (c), removing the period at the end of paragraph (d) and adding “, and” in its place, and adding a new paragraph (e) to read as follows:

§ 123.501 When is your business eligible to apply for a Military Reservist Economic Injury Disaster Loan?

* * * * *

(a) It is a small business as defined in 13 CFR part 121 when the essential employee was called to active duty,

* * * * *

(e) You and your affiliates and principal owners (20% or more ownership interest) have used all reasonably available funds, and you are unable to obtain credit elsewhere (see § 123.104).

Hector V. Barreto,

Administrator.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-SW-59-AD; Amendment 39-12913; AD 2002-21-07]

RIN 2120-AA64

Airworthiness Directives; Sikorsky Aircraft Corporation Model S-76A, S-76B and S-76C Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for the specified Sikorsky Aircraft Corporation (Sikorsky) model helicopters that requires removing and inspecting each main rotor spindle attachment bolt (bolt) to ensure that the correct bolts are installed. This amendment is prompted by the discovery of improper bolts installed on a helicopter during its production. The actions specified by this AD are intended to detect installation of incorrect bolts, which could result in reduced hub or bolt fatigue life, separation of the main rotor blade at the spindle attachment, and subsequent loss of control of the helicopter.

DATES: Effective November 25, 2002.

FOR FURTHER INFORMATION CONTACT: Kirk Gustafson, Aviation Safety Engineer, Boston Aircraft Certification Office, Engine and Propeller Directorate, FAA, 12 New England Executive Park, Burlington, MA 01803, telephone (781) 238-7190, fax (781) 238-7170.

SUPPLEMENTARY INFORMATION: A proposal to amend 14 CFR part 39 to include an AD for Sikorsky Model S-76A, S-76B and S-76C helicopters was published in the *Federal Register* on June 20, 2002 (67 FR 41875). That action proposed to require removing and

inspecting each bolt to ensure that the correct bolts are installed.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the public interest require the adoption of the rule with one change. A “note” has been added following paragraph (b)(3) referencing the alert service bulletin that pertains to the subject of the AD. The FAA has determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

The FAA estimates that 165 helicopters of U.S. registry will be affected by this proposed AD, that it will take approximately 6 work hours per helicopter to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$240 per helicopter. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$99,000, assuming all 40 bolts (per helicopter) are replaced.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the

Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2002-21-07 Sikorsky Aircraft Corporation:
Amendment 39-12913. Docket No. 2001-SW-59-AD.

Applicability: Model S-76A, S-76B and S-76C helicopters, except those having a serial number of 760501, or 760506 through 760515, certificated in any category.

Note 1: This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within 1,250-hours time-in-service or 2 years, whichever comes first, unless accomplished previously.

To detect installation of an incorrect main rotor spindle attachment bolt (bolt), which could result in reduced hub or bolt fatigue life, separation of the main rotor blade at the spindle attachment, and subsequent loss of control of the helicopter, accomplish the following:

(a) Remove and measure each bolt to ensure that the length is 1.181 ±.015 inches. There are 10 bolts per rotor spindle and 40 bolts per helicopter that require inspection.

(1) If 1 or 2 bolts are found on any spindle that are longer than 1.196 inches (1.181 inches + .015-inch permissible tolerance), visually inspect the main rotor hub internal threads for distortion and the hole-bottoms for scoring.

(i) If thread distortion or hole-bottom scoring is found, remove the rotor hub from service.

(ii) If no thread distortion or hole-bottom scoring is found, replace all 10 bolts with new airworthy bolts.

(2) If 3 or more bolts that exceed 1.196 inches are found on any spindle, remove and replace the main rotor hub with an airworthy main rotor hub.

(3) If any bolt is found that is shorter than 1.166 inches (1.181 inches - .015 permissible tolerance), replace it with a new airworthy bolt.

(b) Report the results of the inspections of the main rotor hubs whenever the bolts

exceed 1.196 inches in length, within 5 calendar days of the inspection, to the Manager, Boston Aircraft Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone: (781) 238-7150; fax: (781) 238-7170. Include the following information in the report:

(1) Serial number of the helicopter.

(2) Quantity of incorrect bolts.

(3) Description of thread distortion or hole-bottom scoring caused by each bolt. Information collection requirements contained in this AD have been approved by the Office of Management and Budget (OMB) under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 *et seq.*) and have been assigned OMB Control Number 2120-0056.

Note 2: Sikorsky Aircraft Corporation Alert Service Bulletin No. 76-65-52 (321), dated July 24, 2001, pertains to the subject of this AD.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Boston Aircraft Certification Office, Engine and Propeller Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Boston Aircraft Certification Office.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Boston Aircraft Certification Office.

(d) Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the requirements of this AD can be accomplished.

(e) This amendment becomes effective on November 25, 2002.

Issued in Fort Worth, Texas, on October 4, 2002.

Eric D. Bries,

*Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.*

[FR Doc. 02-26590 Filed 10-18-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-CE-08-AD; Amendment 39-12914; AD 2002-21-08]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Model PC-6 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that

applies to certain Pilatus Aircraft Ltd. (Pilatus) Model PC-6 airplanes. This AD requires you to inspect the aileron assembly for correct configuration and modify as necessary. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. The actions specified by this AD are intended to correct improper aileron assembly configuration, which could result in failure of the aileron mass balance weight. Such failure could lead to loss of control of the airplane.

DATES: This AD becomes effective on December 6, 2002.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of December 6, 2002.

ADDRESSES: You may get the service information referenced in this AD from Pilatus Aircraft Ltd., Customer Liaison Manager, CH-6371 Stans, Switzerland; telephone: +41 41 619 63 19; facsimile: +41 41 619 6224; or from Pilatus Business Aircraft Ltd., Product Support Department, 11755 Airport Way, Broomfield, Colorado 80021; telephone: (303) 465-9099; facsimile: (303) 465-6040. You may view this information at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2002-CE-08-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; facsimile: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Discussion

What Events Have Caused This AD?

The Federal Office for Civil Aviation (FOCA), which is the airworthiness authority for Switzerland, recently notified FAA that an unsafe condition may exist on certain Pilatus Model PC-6 airplanes. The FOCA reported an instance where unapproved mass balance weights and an improper aileron configuration were found on a Model PC-6 airplane. The FOCA determined the cause as improper configuration control and tracking.

What Is the Potential Impact if FAA Took No Action?

This condition, if not corrected, could result in failure of the aileron mass