

action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Pratt & Whitney: Docket No. 2001-NE-25-AD.

Applicability: This airworthiness directive (AD) is applicable to Pratt & Whitney (PW) PW4090, PW4090-3, PW4074D, PW4077D, PW4090D, and PW4098 turbofan engines with 15th stage high pressure compressor (HPC) disks part numbers (P/N's) 56H015 or 57H715. These engines are installed on, but not limited to Boeing 777 airplanes.

Note 1: This AD applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD.

The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Compliance with this AD is required as indicated, unless already done.

To prevent 15th stage HPC disk failures from cracks, which could result in an uncontained engine failure, do the following:

Initial Inspection

(a) Perform an initial inspection for cracks in the front rail of the blade loading slots and front forward and middle knife edges of the 15th stage HPC disk, and replace disk in accordance with paragraphs 1.A. through 1.E.(4) of, "For Engines Installed on Aircraft"; or paragraphs 2.A. through 2.E.(4) of, "For Engines Removed From the Aircraft", of the Accomplishment Instructions of PW Service Bulletin PW 4G-112-A72-242, dated May 1, 2001, and the following Table 1:

TABLE 1.—15TH STAGE HPC DISK INITIAL INSPECTION

Action	If:	Then:
(1) Borescope-inspect disk, within 4,600 cycles-since-new (CSN) or before 90 days after the effective date of this AD, whichever occurs later..	(i) Borescope inspection shows a crack in any knife edge area..	Replace the disk with a serviceable disk before further flight.
.....	(ii) Borescope inspection shows a suspect crack in any loading slot..	Perform an eddy current inspection (ECI) to confirm crack within the next 25 cycles-in-service (CIS), and if cracked replace with a serviceable disk before further flight.

Repetitive Inspections

(b) Perform repetitive inspections in accordance with the inspection procedures in paragraph (a) of this AD at intervals of no more than 1,000 CIS since the last inspection.

New Cyclic Life Limit

(c) This AD establishes a new cyclic life limit for 15th stage HPC disks P/N's 56H015 and 57H715 of 8,000 cycles-since-new (CSN). Thereafter, except as provided in paragraph (d) of this AD, no alternative cyclic life limit may be approved for 15th stage HPC disks P/N's 56H015 and 57H715.

Alternative Methods of Compliance

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(e) Special flight permits may be issued in accordance §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197

and 21.199) to operate the aircraft to a location where the requirements of this AD can be done.

Issued in Burlington, Massachusetts, on November 14, 2001.

Donald E. Plouffe,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 01-29191 Filed 11-21-01; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NE-27-AD]

RIN 2120-AA64

Airworthiness Directives; Pratt & Whitney JT9D-59A, -70A, -7Q, and -7Q3 Turbofan Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The Federal Aviation Administration (FAA) proposes to adopt a new airworthiness directive (AD) that is applicable to Pratt & Whitney (PW) JT9D-59A, -70A, -7Q, and -7Q3 turbofan engines. This proposal would require fluorescent penetrant inspection of the high pressure turbine (HPT) second stage airseal knife edges for cracks, each time the airseal is accessible. This proposal is prompted by reports of cracks found in the knife edges of HPT second stage airseals during HPT disassembly. The actions specified by the proposed AD are intended to prevent failure of HPT second stage airseals due to cracks in the knife edges, which if not detected could result in uncontained engine failure and damage to the airplane.

DATES: Comments must be received by January 22, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2001-NE-27-AD, 12 New England Executive Park, Burlington, MA 01803-5299. Comments may be inspected at this location, by

appointment, between 8:00 a.m. and 4:30 p.m., Monday through Friday, except Federal Holidays. Comments may also be sent via the Internet using the following address: "9-ane-adcomment@faa.gov". Comments sent via the Internet must contain the docket number in the subject line. The service information referenced in the proposed rule may be obtained from Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565-8770; fax (860) 565-4503. This information may be examined, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

FOR FURTHER INFORMATION CONTACT: Tara Goodman, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7130, fax (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2001-NE-27-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRM's

Any person may obtain a copy of this NPRM by submitting a request to the FAA, New England Region, Office of the

Regional Counsel, Attention: Rules Docket No. 2001-NE-27-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

Discussion

The FAA has received 25 reports of cracks found in the knife edges of HPT second stage airseals, part numbers (P/N's) 5002537-01 and 807410, during HPT disassembly. To date, no failed airseal has caused an uncontained engine failure. Results from an evaluation conducted by PW reveal that engine operating temperatures and stresses in the stage 1-to-stage 2 airseal cavity are higher than anticipated. As a result, heavy rubbing and thermal mechanical fatigue in a hot compression environment are causing cracks to initiate in the rear knife edge. These cracks will propagate axially until the airseal fails. Eleven of the 25 cracked HPT second stage airseals found at overhaul were fractured through from snap to snap. This condition, if not corrected, could result in an uncontained engine failure and damage to the airplane.

Manufacturer's Service Information

The FAA has reviewed and approved the technical contents of PW service bulletin (SB) JT9D 6409, dated July 27, 2001, that describes procedures for fluorescent penetrant inspecting knife edges of HPT second stage airseals.

FAA's Determination of an Unsafe Condition and Proposed Actions

Since an unsafe condition has been identified that is likely to exist or develop on other PW JT9D-59A, -70A, -7Q, and -7Q3 turbofan engines of the same type design, the proposed AD would require fluorescent penetrant inspection of the knife edges of HPT second stage airseals for cracks each time the airseal is accessible. The actions would be required to be done in accordance with the SB described previously. The FAA has been informed by PW that a new design HPT second stage airseal is being developed. The FAA may revise this action to introduce the new design as terminating action.

Economic Analysis

There are approximately 564 engines of the affected design PW JT9D-59A, -70A, -7Q, and -7Q3 turbofan engines in the worldwide fleet. The FAA estimates that 176 engines installed on airplanes of U.S. registry would be affected by this proposed AD. The FAA also estimates that it would take approximately 1 work hour per engine to perform the fluorescent penetrant inspection, and that the average labor

rate is \$60 per work hour. Based on these figures, the total labor cost effect annually of the proposed AD on U.S. operators is estimated to be \$10,560.

Regulatory Analysis

This proposed rule does not have federalism implications, as defined in Executive Order 13132, because it would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Accordingly, the FAA has not consulted with state authorities prior to publication of this proposed rule.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption

ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended].

2. Section 39.13 is amended by adding the following new airworthiness directive:

Pratt & Whitney: Docket No. 2001-NE-27-AD.

Applicability: This airworthiness directive (AD) is applicable to Pratt & Whitney (PW) JT9D-59A, -70A, -7Q, and -7Q3 turbofan engines. These engines are installed on, but not limited to, Airbus Industrie A300 series, Boeing 747 series, and McDonnell Douglas DC-10 series airplanes.

Note 1: This AD applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Compliance with this AD is required as indicated, unless already done.

To prevent failure of high pressure turbine (HPT) second stage airseals due to cracks in the knife edges, which if not detected could result in uncontained engine failure and damage to the airplane, do the following:

Inspections

(a) Perform a fluorescent penetrant inspection of the HPT second stage airseal knife edges for cracks in accordance with Accomplishment Instructions, Paragraphs 1 through 3 of PW Service Bulletin (SB) JT9D 6409, dated July 27, 2001, each time the HPT stage 1 and stage 2 rotors are separated. Remove from service those airseals that are found cracked.

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(c) Special flight permits may be issued in accordance §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the aircraft to a location where the requirements of this AD can be done.

Issued in Burlington, Massachusetts, on November 14, 2001.

Donald E. Plouffe,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.
[FR Doc. 01-29190 Filed 11-21-01; 8:45 am]

BILLING CODE 4910-13-U

ENVIRONMENTAL PROTECTION AGENCY

40 CFR Part 136

[FRL-7106-8]

Guidelines Establishing Test Procedures for the Analysis of Pollutants; Whole Effluent Toxicity Test Methods; Extension of Comment Period

AGENCY: Environmental Protection Agency (EPA).

ACTION: Proposed rule; extension of comment period.

SUMMARY: The U.S. Environmental Protection Agency (EPA) is extending the comment period for the proposed rule to revise and ratify its approval of several analytical test procedures measuring "whole effluent toxicity." The proposed rule was published in the **Federal Register** on September 28, 2001 (66 FR 49794), and the comment period was scheduled to end on November 27, 2001. The comment period will be extended for 45 days and will now end on January 11, 2002.

DATES: Comments must be postmarked, delivered by hand, or electronically mailed on or before January 11, 2002. Comments provided electronically will be considered timely if they are submitted by 11:59 p.m. Eastern Standard Time (EST) on January 11, 2002.

ADDRESSES: Send written or electronic comments on the proposed rule (66 FR 49794) to "Whole Effluent Toxicity (WET) Test Method Changes" Comment Clerk (WET-IX); Water Docket (4101); U.S. Environmental Protection Agency; Ariel Rios Building; 1200 Pennsylvania Avenue, NW., Washington, DC 20460. EPA requests that commenters submit copies of any references cited in comments. Commenters also are requested to submit an original and three copies of their written comments and enclosures. Commenters that want receipt of their comments acknowledged should include a self-addressed, stamped envelope. All written comments must be postmarked or delivered by hand. No facsimiles (faxes) will be accepted. Hand deliveries should be delivered to EPA's Water Docket at 401 M Street, SW., Room EB 57, Washington, DC 20460.

Comments may be submitted electronically to: OW-Docket@epa.gov. Electronic comments must be submitted as a Word Perfect 5/6/7/8 file or an ASCII file, avoiding the use of special characters and any form of encryption. Comments and data also will be

accepted on disks in WordPerfect 5/6/7/8 or ASCII file format. Electronic comments may be filed online at any Federal Depository Library. All electronic comments must be identified by docket number (WET-IX). Electronic comments will be transferred into a paper version for the official record. EPA will attempt to clarify electronic comments if there is an apparent error in transmission.

A record for the proposed rulemaking (66 FR 49794) has been established under docket number WET-IX. A copy of the supporting documents cited in the proposed rule is available for review at EPA's Water Docket, East Tower Basement (Room EB 57), 401 M Street, SW., Washington, DC 20460. For access to docket materials, call (202) 260-3027 on Monday through Friday, excluding Federal holidays, between 9 a.m. and 3:30 p.m. EST to schedule an appointment.

The proposed rule (66 FR 49794) has been placed on the Internet for public review and downloading at the following location: <http://www.epa.gov/fedrgstr/>. Other documents referenced in the proposed rule also are available on the Internet. The final report of EPA's WET Interlaboratory Variability Study (Volumes 1 and 2) and the document titled, Proposed Changes to Whole Effluent Toxicity Method Manuals are available on the Internet at <http://www.epa.gov/waterscience/WET>.

FOR FURTHER INFORMATION CONTACT: For regulatory information regarding this notice or the proposed rule, contact Marion Kelly, Engineering and Analysis Division (4303), Office of Science and Technology, Office of Water, U.S. Environmental Protection Agency, 1200 Pennsylvania Avenue, NW., Washington, DC 20460 (e-mail: kelly.marion@epa.gov) or call (202) 260-7117. For technical information regarding the proposed rule, contact Teresa J. Norberg-King, National Health and Environmental Effects Research Laboratory, Mid-Continent Ecology Division, Office of Research and Development, U.S. Environmental Protection Agency, 6201 Congdon Boulevard, Duluth, MN 55804 (e-mail: norberg-king.teresa@epa.gov) or call (218) 529-5163.

SUPPLEMENTARY INFORMATION: On September 28, 2001, EPA published in the **Federal Register** (66 FR 49794) a proposed rule to ratify its approval of several whole effluent toxicity (WET) test methods, which the Agency standardized in an earlier rulemaking (60 FR 53529; October 16, 1995). The proposed rule published on September 28, 2001 also would modify the WET