

period, it is unlikely that this interim rule will have any substantial effects on trade, or on large or small businesses.

Under these circumstances, the Administrator of the Animal and Plant Health Inspection Service has determined that this action will not have a significant economic impact on a substantial number of small entities.

Executive Order 12988

This rule has been reviewed under Executive Order 12988, Civil Justice Reform. This rule: (1) Preempts all State and local laws and regulations that are inconsistent with this rule; (2) has retroactive effect to June 5, 2007; and (3) does not require administrative proceedings before parties may file suit in court challenging this rule.

Paperwork Reduction Act

This interim rule contains no information collection or recordkeeping requirements under the Paperwork Reduction Act of 1995 (44 U.S.C. 3501 *et seq.*).

List of Subjects in 9 CFR Part 94

Animal diseases, Imports, Livestock, Meat and meat products, Milk, Poultry and poultry products, Reporting and recordkeeping requirements.

■ Accordingly, we are amending 9 CFR part 94 as follows:

PART 94—RINDERPEST, FOOT-AND-MOUTH DISEASE, FOWL PEST (FOWL PLAGUE), EXOTIC NEWCASTLE DISEASE, AFRICAN SWINE FEVER, CLASSICAL SWINE FEVER, AND BOVINE SPONGIFORM ENCEPHALOPATHY: PROHIBITED AND RESTRICTED IMPORTATIONS

■ 1. The authority citation for part 94 continues to read as follows:

Authority: 7 U.S.C. 450, 7701–7772, 7781–7786, and 8301–8317; 21 U.S.C. 136 and 136a; 31 U.S.C. 9701; 7 CFR 2.22, 2.80, and 371.4.

§ 94.8 [Amended].

■ 2. In § 94.8, the introductory text is amended by adding the word “Georgia,” after the word “Cuba.”

Done in Washington, DC, this 12th day of September 2007.

Kevin Shea,

Acting Administrator, Animal and Plant Health Inspection Service.

[FR Doc. E7–18315 Filed 9–17–07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2007–27864; Directorate Identifier 2007–CE–038–AD; Amendment 39–15161; AD 2007–17–03]

RIN 2120–AA64

Airworthiness Directives; Pacific Aerospace Corporation, Ltd. Model 750XL Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2007–17–03, which was published in the **Federal Register** on August 21, 2007 (72 FR 46541), and applies to certain Pacific Aerospace Corporation Ltd Model 750XL airplanes. AD 2007–17–03 requires inspecting the inboard end of the rear spar for security of the blind rivets, inspecting the radii of the rear spar upper and lower flanges for cracking, inspecting the aft flange of the inboard rib for cracking, replacing the rear spar if cracks are found in any of the inspections, and replacing rear spar blind rivets with bolts or rivets. The paragraph following the section heading Materials Incorporated by Reference does not have a paragraph designator. This document corrects that paragraph by inserting the paragraph designator “(i)” preceding the paragraph text.

DATES: The effective date of this AD (2007–17–03) remains September 25, 2007.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4146; fax: (816) 329–4090.

SUPPLEMENTARY INFORMATION:

Discussion

On August 8, 2007, the FAA issued AD 2007–17–03, Amendment 39–15161 (72 FR 46541, August 21, 2007), which applies to certain Pacific Aerospace Corporation Ltd Model 750XL airplanes. AD 2007–17–03 requires you to inspect the inboard end of the rear spar for security of the blind rivets, inspecting the radii of the rear spar upper and lower flanges for cracking, inspecting the aft flange of the inboard rib for cracking, replacing the rear spar if cracks are found in any of the inspections, and replacing rear spar blind rivets with bolts or rivets. The paragraph following the section heading

Materials Incorporated by Reference does not have a paragraph designator. This document corrects that paragraph by inserting the paragraph designator “(i)” preceding the paragraph text.

Need for the Correction

This correction is needed to specify the paragraph designator for the paragraph that follows the Materials Incorporated by Reference section heading.

Correction of Publication

■ Accordingly, the publication of August 21, 2007 (72 FR 46541), of Amendment 39–15161; AD 2007–17–03 which was the subject of FR Doc. E7–15978, is corrected as follows:

§ 39.13 [Corrected]

■ On page 46542, in the second column, at the beginning of the paragraph following the Materials Incorporated by Reference section heading, the paragraph designator (i), indented two spaces, precedes the paragraph.

Action is taken herein to correct this reference in AD 2007–17–03 and to add this AD correction to § 39.13 of the Federal Aviation Regulations (14 CFR 39.13).

The effective date remains September 25, 2007.

Issued in Kansas City, Missouri, on September 10, 2007.

Kim Smith,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7–18136 Filed 9–17–07; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2007–28372; Directorate Identifier 2007–NM–080–AD; Amendment 39–15194; AD 2007–19–04]

RIN 2120–AA64

Airworthiness Directives; Airbus Model A300F4–605R and A300F4–622R Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of

another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Further to cases of parking brake loss at the gate, a pressure switch system had been introduced on some A300–600 aircraft. The aim of this modification was to recover pedals braking authority if parking brake is not efficient, without having to set the parking brake handle to OFF.

However, it appears that in case of failure of the pressure switch system, there is the risk of double (normal and alternate) pressurization of the brakes potentially leading to undetected residual braking, which may lead to a loss of performances of the aircraft at Take-Off.

The loss of performance could result in runway overrun or impact with obstacles or terrain during takeoff. We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective October 23, 2007.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 23, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT: Tom Stafford, Aerospace Engineer, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–1622; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on June 11, 2007 (72 FR 32025). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Further to cases of parking brake loss at the gate, a pressure switch system had been introduced on some A300–600 aircraft. The aim of this modification was to recover pedals braking authority if parking brake is not efficient, without having to set the parking brake handle to OFF.

However, it appears that in case of failure of the pressure switch system, there is the risk of double (normal and alternate) pressurization of the brakes potentially leading to undetected residual braking,

which may lead to a loss of performances of the aircraft at Take-Off.

This new AD requires accomplishment of a wiring modification that will inhibit the effect of modifications 12088 and 12403.

The loss of performance could result in runway overrun or impact with obstacles or terrain during takeoff. You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a Note within the AD.

Costs of Compliance

Based on the service information, we estimate that this AD affects about 51 products of U.S. registry. We also estimate that it takes about 3 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Where the service information lists required labor costs that are covered under warranty at the operator's agreed in-house warranty labor rate, we have assumed that there will be no charge for these costs. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD on U.S. operators to be \$12,240, or \$240 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of

the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2007-19-04 Airbus: Amendment 39-15194.
Docket No. FAA-2007-28372;
Directorate Identifier 2007-NM-080-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective October 23, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Airbus Model A300F4-605R and A300F4-622R airplanes; certificated in any category; all serial numbers; on which Airbus Modifications 12088 and 12403 have been embodied during production, or which incorporated Airbus Service Bulletin A300-32-6085 in service, except airplanes on which Airbus Modification 12618 has been embodied during production, or which incorporated Airbus Service Bulletin A300-32-6100 in service.

Subject

(d) Air Transport Association (ATA) of America Code 32: Landing Gear.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states: Further to cases of parking brake loss at the gate, a pressure switch system had been introduced on some A300-600 aircraft. The aim of this modification was to recover pedals braking authority if parking brake is not efficient, without having to set the parking brake handle to OFF.

However, it appears that in case of failure of the pressure switch system, there is the risk of double (normal and alternate) pressurization of the brakes potentially leading to undetected residual braking, which may lead to a loss of performances of the aircraft at Take-Off.

This new AD requires accomplishment of a wiring modification that will inhibit the effect of modifications 12088 and 12403. The loss of performance could result in runway overrun or impact with obstacles or terrain during takeoff.

Actions and Compliance

(f) Within 3 months after the effective date of this AD unless already done: Modify the wiring in the right electronics rack 90VU (volt unit), in accordance with the instructions of Airbus Service Bulletin A300-32-6100, dated September 18, 2006.

FAA AD Differences

Note: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tom Stafford, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1622; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2007-0068, dated March 14, 2007; and Airbus Service Bulletin A300-32-6100, dated September 18, 2006; for related information.

Material Incorporated by Reference

(i) You must use Airbus Service Bulletin A300-32-6100, dated September 18, 2006, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France.

(3) You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on August 31, 2007.

Stephen P. Boyd,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-18050 Filed 9-17-07; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-28308; Directorate Identifier 2007-NM-016-AD; Amendment 39-15195; AD 2007-19-05]

RIN 2120-AA64

Airworthiness Directives; Hawker Beechcraft Model 400, 400A, and 400T Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Hawker Beechcraft Model 400, 400A, and 400T series airplanes. This AD requires modifying the attachment fasteners on the engine cowlings panels. This AD results from several reports of loose attachment fasteners found on the engine cowlings panels, and subsequently the panels either peeling back or separating from the airplane during flight. We are issuing this AD to prevent failure of the attachment fasteners on the engine cowlings panels, which could result in separation of a panel from the airplane, and consequent damage to airplane structure. These conditions could adversely affect continued safe flight and landing of the airplane, or cause injury to people or damage to property on the ground.

DATES: This AD becomes effective October 23, 2007.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of October 23, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

Contact Hawker Beechcraft Corporation, 9709 East Central, Wichita, Kansas 67206, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: William Griffith, Aerospace Engineer,