

using the procedures found in 14 CFR 39.19. Send information to ATTN: Wahib Mina, ANM-120L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5324; fax (562) 627-5210.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office. The AMOC approval letter must specifically reference this AD.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Los Angeles ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

Material Incorporated by Reference

(o) You must use Boeing Alert Service Bulletin DC9-32A350, Revision 2, dated March 20, 2009, as applicable, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, 3855 Lakewood Boulevard, MC D800-0019, Long Beach, California 90846-0001; telephone 206-544-5000, extension 2; fax 206-766-5683; e-mail dse.boecom@boeing.com; Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on November 19, 2009.

Stephen P. Boyd,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E9-28564 Filed 12-3-09; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2009-0797; Directorate Identifier 2009-CE-032-AD; Amendment 39-16118; AD 2009-25-01]

RIN 2120-AA64

Airworthiness Directives; Hawker Beechcraft Corporation Models 58, 58A, 58P, 58PA, 58TC, 58TCA, 95-B55, 95-B55A, A36, A36TC, B36TC, E55, E55A, F33A, and V35B Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) to supersede AD 91-18-19, which applies to certain Hawker Beechcraft Corporation (Hawker) (Type Certificate Numbers 3A15, 3A16, and A23CE formerly held by Raytheon Aircraft Company; formerly held by Beech Aircraft Corporation) Models 58, 58A, 58P, 58PA, 58TC, 58TCA, 95-B55, 95-B55A, A36, A36TC, B36TC, E55, E55A, F33A, and V35B airplanes. AD 91-18-19 currently requires you to do a one-time inspection of the pilot and copilot shoulder harnesses for an incorrect washer and replace any incorrect washer with the correct washer. Since we issued AD 91-18-19, we have found that the applicability of AD 91-18-19 was incorrectly stated when the Model A36TC airplane was omitted from the Applicability section. Consequently, this AD would retain the actions and the serial number (SN) applicability of AD 91-18-19 and realign the SN applicability for Models A36TC and B36TC airplanes. We are issuing this AD to detect and correct an incorrect washer installed in the pilot and copilot shoulder harnesses. This incorrect part could result in a malfunctioning shoulder harness. Such a malfunction could lead to occupant injury.

DATES: This AD becomes effective on January 8, 2010.

As of October 21, 1991 (56 FR 42224, August 27, 1991), the Director of the Federal Register approved the incorporation by reference of Beechcraft Mandatory Service Bulletin No. 2394, dated December 1990, listed in this AD.

ADDRESSES: For service information identified in this AD, contact Hawker Beechcraft Corporation, P.O. Box 85, Wichita, Kansas 67201-0085; telephone: (800) 429-5372 or (316) 676-3140;

Internet: <http://pubs.hawkerbeechcraft.com>.

To view the AD docket, go to U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590, or on the Internet at <http://www.regulations.gov>. The docket number is FAA-2009-0797; Directorate Identifier 2009-CE-032-AD.

FOR FURTHER INFORMATION CONTACT:

Steve Potter, Aerospace Engineer, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: (316) 946-4124; fax: (316) 946-4107.

SUPPLEMENTARY INFORMATION:

Discussion

On August 20, 2009, we issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Hawker (Type Certificate Numbers 3A15, 3A16, and A23CE formerly held by Raytheon Aircraft Company; formerly held by Beech Aircraft Corporation) Models 58, 58A, 58P, 58PA, 58TC, 58TCA, 95-B55, 95-B55A, A36, A36TC, B36TC, E55, E55A, F33A, and V35B airplanes. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on August 28, 2009 (74 FR 44311). The NPRM proposed to supersede AD 91-18-19 (56 FR 42224, August 27, 1991) with a new AD that would retain the actions and the SN applicability of AD 91-18-19 and realign the SN applicability for Models A36TC and B36TC airplanes.

Comments

We provided the public the opportunity to participate in developing this AD. We received no comments on the proposal or on the determination of the cost to the public.

Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial corrections. We have determined that these minor corrections:

- Are consistent with the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Costs of Compliance

We estimate that this AD affects 4,792 airplanes in the U.S. registry.

We estimate the following costs to do the inspection:

| Labor cost | Parts cost | Total cost per airplane | Total cost on U.S. operators |
|--|----------------------|-------------------------|------------------------------|
| 1 work-hour × \$80 per hour = \$80 | Not applicable | \$80 | \$383,360 |

We estimate the following costs to do any necessary replacements that would be required based on the results of the inspection. We have no way of determining the number of airplanes that may need this replacement:

| Labor cost | Parts cost | Total cost per airplane |
|--|------------|-------------------------|
| 1 work-hour × \$80 per hour = \$80 | \$5 | \$85 |

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106 describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this AD.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD (and other information as included in the Regulatory Evaluation) and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "Docket No. FAA-2009-0797; Directorate Identifier 2009-CE-032-AD" in your request.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 91-18-19, Amendment 39-8022 (56 FR 42224, August 27, 1991), and adding the following new AD:

2009-25-01 Hawker Beechcraft

Corporation: Amendment 39-16118; Docket No. FAA-2009-0797; Directorate Identifier 2009-CE-032-AD.

Effective Date

(a) This AD becomes effective on January 8, 2010.

Affected ADs

(b) This AD supersedes AD 91-18-19, Amendment 39-8022.

Applicability

(c) This AD applies to the following airplane models and serial numbers that are certificated in any category:

(1) *Group 1 Airplanes* (retains the actions and applicability from AD 91-18-19):

| Model | Serial Nos. (SNs) |
|-----------------------|-----------------------------------|
| 58, 58A | TH-733 through TH-1609. |
| 58P, 58PA | TJ-3 through TJ-497. |
| 58TC, 58TCA | TK-1 through TK-151. |
| 95-B55, 95-B55A | TC-1947 through TC-2456. |
| A36 | E-825 through E-2578. |
| B36TC | EA-242 and EA-273 through EA-509. |
| E55, E55A | TE-1078 through TE-1201. |
| F33A | CE-634 through CE-1536. |
| V35B | D-9862 through D-10403. |

(2) *Group 2 Airplanes* (aligns certain SNs applicability to Models A36TC airplanes):

| Model | SNs |
|-------|--|
| A36TC | EA-1 through EA-241 and EA-243 through EA-272. |

Unsafe Condition

(d) This AD results from reports of incorrect washers installed in the pilot and copilot shoulder harnesses on certain Beech 33, 35, 36, 55, 58, and 95 series airplanes. We are issuing this AD to detect and correct an incorrect washer installed in the pilot and copilot shoulder harnesses. This incorrect part could result in a malfunctioning shoulder harness. Such a malfunction could lead to occupant injury.

Compliance

(e) To address this problem, you must do the following, unless already done:

| Actions | Compliance | Procedures |
|---|--|---|
| (1) Inspect the washers on the "D" ring of the pilot and copilot shoulder harnesses for correct metal, inner and outer diameter, and thickness. | (i) <i>For Group 1 Airplanes:</i> Within the next 100 hours time-in-service (TIS) after October 21, 1991 (the effective date of AD 91-18-19). (ii) <i>For Group 2 Airplanes:</i> Within the next 100 hours TIS after January 8, 2010 (the effective date of this AD). | Follow Beechcraft Mandatory Service Bulletin No. 2394, dated December 1990. |

| Actions | Compliance | Procedures |
|--|--|---|
| (2) If you find, as a result of the inspection required by paragraph (e)(1) of this AD, any washer does not meet the criteria for correct metal, inner and outer diameter, and thickness, replace the incorrect washer with part number 100951X060YA washer. | Before further flight, after the inspection required by paragraph (e)(1) of this AD. | Follow Beechcraft Mandatory Service Bulletin No. 2394, dated December 1990. |

Alternative Methods of Compliance (AMOCs)

(f) The Manager, Wichita Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to *Attn:* Steve Potter, Aerospace Engineer, ACE-118W, Wichita Aircraft Certification Office (ACO), 1801 Airport Road, Room 100, Wichita, Kansas 67209; *telephone:* (316) 946-4124; *fax:* (316) 946-4107. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(g) In reviewing the docket and project files, we found no AMOCs submitted for AD 91-18-19. Since there are no AMOCs approved for AD 91-18-19 to approve for this AD, transfer of AMOCs to this AD does not apply.

Material Incorporated by Reference

(h) You must use Beechcraft Mandatory Service Bulletin No. 2394, dated December 1990, to do the actions required by this AD, unless the AD specifies otherwise. AD 91-18-19 (56 FR 42224; August 27, 1991), which is superseded by this airworthiness directive, incorporated this service information by reference as Beech Service Bulletin No. 2394, dated December 1990.

(1) On October 21, 1991 (56 FR 42224, August 27, 1991), the Director of the Federal Register approved the incorporation by reference of Beechcraft Mandatory Service Bulletin No. 2394, dated December 1990, under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Hawker Beechcraft Corporation, P.O. Box 85, Wichita, Kansas 67201-0085; *telephone:* (800) 429-5372 or (316) 676-3140; *Internet:* <http://pubs.hawkerbeechcraft.com>.

(3) You may review copies of the service information incorporated by reference for this AD at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the Central Region, call (816) 329-3768.

(4) You may also review copies of the service information incorporated by reference for this AD at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Kansas City, Missouri, on November 20, 2009.

Margaret Kline,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E9-28565 Filed 12-3-09; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2009-0868; Directorate Identifier 2009-CE-047-AD; Amendment 39-16120; AD 2009-25-03]

RIN 2120-AA64

Airworthiness Directives; ZLT Zeppelin Luftschifftechnik GmbH & Co KG Model LZ N07-100 Airships

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

The manufacturer has advised of receiving a report that during start up on ground a RH propeller gear box (PGB) on the airship has failed resulting in free rotation of the propeller. Investigation performed by the manufacturer revealed that the bevel gear in the propeller gearbox had cracked near the hub area.

During an extensive metallurgical investigation of the cracked bevel gear some different manufacturing deviations outside of the specifications were detected. Deviations in the heat treatment, wall thickness of the bevel gear near the hub area, and score marks caused during the production process have been established as causal factors for this failure.

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective January 8, 2010.

On January 8, 2010, the Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; *telephone:* (816) 329-4146; *fax:* (816) 329-4090; *e-mail:* karl.schletzbaum@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on September 21, 2009 (74 FR 48019). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

The manufacturer has advised of receiving a report that during start up on ground a RH propeller gear box (PGB) on the airship has failed resulting in free rotation of the propeller. Investigation performed by the manufacturer revealed that the bevel gear in the propeller gearbox had cracked near the hub area.

During an extensive metallurgical investigation of the cracked bevel gear some different manufacturing deviations outside of the specifications were detected. Deviations in the heat treatment, wall thickness of the bevel gear near the hub area, and score marks caused during the production process have been established as causal factors for this failure.

For the reasons described above, this new AD mandates the replacement of the affected bevel gears, and limits, as a temporary measure, their service-life to 1,000 Flight Hours (for non-refurbished PGBs) and to 1,600 Flight Hours (for refurbished PGBs).

You may obtain further information by examining the MCAI in the AD docket.