

necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2003-12-04—EMPRESA BRASILEIRA DE AERONAUTICA S.A. (EMBRAER): Amendment 39-13189. Docket 2002-NM-88-AD.

Applicability: Model EMB-135 and -145 series airplanes, as listed in EMBRAER Service Bulletin 145-36-0024, Change 02, dated December 13, 2002; excluding those airplanes listed in "In-production effectivity" in paragraph 1.A., "Effectivity," of the service bulletin; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent hot air leaks from the bleed air duct due to disconnection of the duct joint, which could result in heat damage to components near the duct, and consequent increased risk of fire in the rear baggage compartment, accomplish the following:

Replacement

(a) Within 1,000 flight hours after the effective date of this AD, do the actions specified in paragraphs (a)(1) and (a)(2) of this AD, as applicable, per EMBRAER Service Bulletin 145-36-0024, Change 02, dated December 13, 2002. Accomplishment of those actions in accordance with EMBRAER Service Bulletin 145-36-0024, dated May 31, 2001; or Change 01, dated August 7, 2002; are acceptable for compliance with the requirements of this paragraph.

(1) For all airplanes: Replace the four Gamah clamp/sleeve joints from the bleed line at the baggage compartment between frames 68 and 69 with new threaded coupling assemblies (including re-identifying, cleaning, and lubricating the bleed ducts; and installing protection sleeves).

(2) For airplanes having serial numbers listed in paragraph 3.G. of the Accomplishment Instructions of the service bulletin: Replace the two supports for the engine bleed air duct with two new supports, having part number 145-35923-007.

Parts Installation

(b) As of the effective date of this AD, no person shall install parts listed in paragraphs (b)(1) and (b)(2) of this AD, as applicable.

(1) For all airplanes: Gamah clamp/sleeve joints, from the bleed line at the baggage compartment between frames 68 and 69, having part number G30020CD, G30020TD, G30020C, or G30020T.

(2) For airplanes having serial numbers listed in paragraph 3.G. of the Accomplishment Instructions of EMBRAER Service Bulletin 145-36-0024, Change 02, dated December 13, 2002: Supports for the engine bleed air duct, with part number 145-35923-007.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport

Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) Unless otherwise specified in this AD, the actions must be done in accordance with EMBRAER Service Bulletin 145-36-0024, Change 02, dated December 13, 2002. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Empresa Brasileira de Aeronautica S.A. (EMBRAER), PO Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in Brazilian airworthiness directive 2001-09-03, dated October 2, 2001.

Effective Date

(f) This amendment becomes effective on July 17, 2003.

Issued in Renton, Washington, on June 4, 2003.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-14524 Filed 6-11-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NM-01-AD; Amendment 39-13188; AD 2003-12-03]

RIN 2120-AA64

Airworthiness Directives; Israel Aircraft Industries, Ltd., Model 1124 and 1124A Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD),

applicable to all Israel Aircraft Industries, Ltd., Model 1124 and 1124A series airplanes, that requires revising the airplane flight manual to advise the flightcrew to don oxygen masks as a first and immediate step following a cabin altitude alert. This action is necessary to prevent incapacitation of the flightcrew due to lack of oxygen. This action is intended to address the identified unsafe condition.

DATES: Effective July 17, 2003.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 17, 2003.

ADDRESSES: The service information referenced in this AD may be obtained from Gulfstream Aerospace Corporation, PO Box 2206, Mail Station D25, Savannah, Georgia 31402. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Tim Dulin, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2141; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Israel Aircraft Industries, Ltd., Model 1124 and 1124A series airplanes was published in the **Federal Register** on March 25, 2003 (68 FR 14353). That action proposed to require revising the airplane flight manual to advise the flightcrew to don oxygen masks as a first and immediate step following a cabin altitude alert.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Changes to 14 CFR Part 39/Effect on the AD

On July 10, 2002, the FAA issued a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the

FAA's airworthiness directives system. The regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. However, for clarity and consistency in this final rule, we have retained the language of the NPRM regarding that material.

Cost Impact

The FAA estimates that 198 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$11,880, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2003-12-03 ISRAEL AIRCRAFT INDUSTRIES, LTD.: Amendment 39-13188. Docket 2003-NM-01-AD.

Applicability: All Model 1124 and 1124A series airplanes, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent incapacitation of the flightcrew due to lack of oxygen, accomplish the following:

Revision to Airplane Flight Manual (AFM)

(a) Within 1 month after the effective date of this AD, revise the Emergency Procedures section of the AFM, as specified in paragraph (a)(1) or (a)(2) of this AD, as applicable.

(1) For Model 1124 series airplanes: Insert Israel Aircraft Industries, Ltd. 1124-Westwind Temporary Revision 3, dated January 16, 2001, into the 1124 Westwind AFM.

(2) For Model 1124A series airplanes: Insert Israel Aircraft Industries, Ltd. 1124A-Westwind Temporary Revision 5, dated January 16, 2001, into the 1124A Westwind AFM.

(b) When the information in the temporary revisions identified in paragraph (a) of this AD has been incorporated into the general revisions of the respective AFM, the general revisions may be incorporated into the AFMs, and these temporary revisions may be removed from the AFM.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Operations Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 1: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR

21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) The actions shall be done in accordance with Israel Aircraft Industries, Ltd. 1124-Westwind, Airplane Flight Manual, Temporary Revision 3, dated January 16, 2001; or Israel Aircraft Industries, Ltd. 1124A-Westwind Temporary Revision 5, dated January 16, 2001; as applicable. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Gulfstream Aerospace Corporation, PO Box 2206, Mail Station D25, Savannah, Georgia 31402. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 2: The subject of this AD is addressed in Israeli airworthiness directive 21-02-07-01, dated July 22, 2002.

Effective Date

(f) This amendment becomes effective on July 17, 2003.

Issued in Renton, Washington, on June 4, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-14523 Filed 6-11-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-64-AD; Amendment 39-13186; AD 2003-12-01]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 777 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all Boeing Model 777 series airplanes, that requires either a one-time inspection or a review of the airplane maintenance records for both stabilizer trim control modules (STCM) of the trim system of the horizontal stabilizer to determine if STCMs having certain serial numbers are installed; and follow-on corrective actions, if necessary. This amendment also requires eventual replacement of affected STCMs with new or reworked STCMs, which would terminate the

follow-on actions. The actions specified by this AD are intended to prevent an uncommanded stabilizer trim due to simultaneous failure of two static seals on one STCM, combined with failure of the automatic shutdown function of the stabilizer trim system. Such failures could result in loss of pitch control and consequent loss of control of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective July 17, 2003. The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 17, 2003.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Kenneth J. Fairhurst, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6456; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Boeing Model 777 series airplanes was published in the *Federal Register* on August 30, 2002 (67 FR 55737). That action proposed to require either a one-time inspection or a review of the airplane maintenance records for both stabilizer trim control modules (STCM) of the trim system of the horizontal stabilizer to determine if STCMs having certain serial numbers (S/N) are installed; and follow-on corrective actions, if necessary. That action also proposed to require eventual replacement of affected STCMs with new or reworked STCMs, which would terminate the follow-on actions.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

Requests To Change Paragraphs (b) and (d)

Three commenters ask that certain wording in paragraphs (b) and (d) of the

proposed AD be changed. Two commenters ask that the wording be changed to specify allowing installation of STCMs having S/Ns 006 through 556 inclusive, if the part has been reworked and marked with an "R" on the nameplate, or if MOOG Aircraft Group Service Bulletin Number 160300-27-124 is marked on the modification plate. The third commenter asks that the information phrase (modified and marked with an "R" suffix) be removed from paragraph (b) of the proposed AD, and that "unless reworked per Part 2 of the Work Instructions of Boeing Service Bulletin 777-27A0047, Revision 2, dated October 11, 2001," be added to paragraph (d) of the proposed AD.

The FAA agrees with the commenters because Part 2 of the Work Instructions of the referenced service bulletin specifies procedures for the installation of STCMs that have been reworked and marked with an "R" on the nameplate, or that include MOOG Aircraft Group Service Bulletin Number 160300-27-124 on the modification plate. We also agree to remove the information phrase (modified and marked with an "R" suffix) from paragraph (b) of the final rule, because the STCM also can be marked with the MOOG service bulletin number. Paragraphs (b) and (d) of this final rule have been changed accordingly.

Requests To Change STCM Serial Numbers

Two commenters state that the range of STCM S/Ns specified in the proposed AD section titled "Explanation of Relevant Service Information" is incorrect and should be changed. The first commenter states that the S/Ns in that section should be corrected to specify 006 through 556 inclusive. The second commenter states that Part 2 of the Work Instructions of the referenced service bulletin references S/Ns 006 through 549 inclusive (however, it actually specifies 006 through 556 inclusive). The commenter states that S/Ns 006 through 556 are the correct S/Ns and recommends those numbers be specified throughout the proposed AD to eliminate any confusion. The commenter also states that listing airplanes having S/Ns 2 through 266 and 273, excluding line numbers 256, 258, and 260 through 263 inclusive, as being subject to the actions specified in service bulletin, could be misinterpreted. The commenter recommends that the S/Ns in the applicability of the proposed AD should match the S/Ns (006 through 556 inclusive) listed in MOOG Aircraft Group Service Bulletin Number 160300-27-124.