

commerce on the day after the date of acceptance of a substantially complete notice by the appropriate regional director. The notificant(s) may consummate the proposed acquisition after the expiration of the 60-day notice period, unless the FDIC disapproves the proposed acquisition or extends the notice period.

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■ 12. Section 303.86 is amended by revising paragraph (c) to read as follows:

§ 303.86 Public Notice requirements.

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(c) *Shortening or waiving public comment period, waiving publications; acting before close of public comment period.* The FDIC may shorten the public comment period to a period of not less than 10 days, or waive the public comment or newspaper publication requirements of paragraph (a) of this section, or act on a notice before the expiration of a public comment period, if it determines in writing either that an emergency exists or that disclosure of the notice, solicitation of public comment, or delay until expiration of the public comment period would seriously threaten the safety and soundness of the bank to be acquired.

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■ 13. In section 303.244, paragraphs (c)(4) and (c)(5) are revised and new paragraph (c)(6) is added to read as follows:

§ 303.244 Golden parachute and severance plan payments.

* * * * *

(c) * * *

(4) The cost of the proposed payment and its impact on the institution's capital and earnings;

(5) The reasons why the consent to the payment should be granted; and

(6) Certification and documentation as to each of the points cited in § 359.4(a)(4).

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PART 333—EXTENSION OF CORPORATE POWERS

■ 14. The authority citation for part 333 continues to read as follows:

Authority: 12 U.S.C. 1816, 1818, 1819 (“Seventh”, “Eighth” and “Tenth”), 1828, 1828(m), 1831p–1(c).

§ 333.4 [Amended]

■ 15. In § 333.4, paragraphs (a) and (c) are amended by removing the words “§ 303.15 of this chapter” and adding in their place the words “subpart I of part 303 of this chapter.”

PART 347—INTERNATIONAL BANKING

■ 16. The authority citation for part 347 continues to read as follows:

Authority: 12 U.S.C. 1813, 1815, 1817, 1819, 1820, 1828, 3103, 3104, 3105, 3108; Title IX, Pub. L. 98–181, 97 Stat. 1153.

■ 17. Section 347.108 is amended by revising paragraph (f) to read as follows:

§ 347.108 Obtaining FDIC approval to invest in foreign organizations.

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(f) *Procedures.* Procedures for applications and notices under this section are set out in subpart J of part 303 of this chapter.

PART 348—MANAGEMENT OFFICIAL INTERLOCKS

■ 18. The authority citation for part 348 continues to read as follows:

Authority: 12 U.S.C. 1823(k), 3207.

■ 19. In § 348.2, paragraph (j)(1)(iii) is revised to read as follows:

§ 348.2 Definitions.

* * * * *

(j) * * *

(iii) A senior executive officer as that term is defined in 12 CFR 303.101(b).

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PART 359—GOLDEN PARACHUTE AND INDEMNIFICATION PAYMENTS

■ 20. The authority citation for part 359 continues to read as follows:

Authority: 12 U.S.C. 1828(k).

§ 359.1 [Amended]

■ 21. In § 359.1(f)(1)(ii)(C) remove the reference to “§ 303.14(a)(4)” and add in its place, “§ 303.101(c)”.

Dated at Washington, DC, this 4th day of August, 2003.

By order of the Board of Directors.

Federal Deposit Insurance Corporation.

Valerie J. Best,

Assistant Executive Secretary.

[FR Doc. 03–20451 Filed 8–20–03; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000–NE–13–AD; Amendment 39–13200; AD 2003–12–15]

RIN 2120–AA64

Airworthiness Directives; Rolls-Royce RB211 Series Turbofan Engines; Correction

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2003–12–15 that applies to Rolls-Royce (RR) plc RB211–535E4–37, RB211–535E4–B–37, and RB211–535E4–B–75 series turbofan engines that was published in the **Federal Register** on June 25, 2003. A service bulletin was incorrectly identified by revision number and revision date in the Compliance section, paragraph (a) and the Optional Terminating Action section, paragraph (f). This document corrects these items. In all other respects, the original document remains the same.

EFFECTIVE DATE: Effective June 25, 2003.

FOR FURTHER INFORMATION CONTACT: James Lawrence, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7176; fax (781) 238–7199.

SUPPLEMENTARY INFORMATION: A final rule AD, FR Doc 03–15449, that applies to Rolls-Royce (RR) plc RB211–535E4–37, RB211–535E4–B–37, and RB211–535E4–B–75 series turbofan engines, was published in the **Federal Register** on June 25, 2003 (68 FR 37735). The following corrections are needed:

§ 39.13 [Corrected]

■ On page 37736, in the third column, in the Compliance section, paragraph (a), in the third line, “dated August 6, 2002,” is corrected to read “Revision 2, dated September 26, 2002,”.

■ On page 37738, in the first column, in the Optional Terminating Action section, paragraph (f) in the third line, “Revision 1, dated August 6, 2002,” is corrected to read “Revision 2, dated September 26, 2002,”.

Issued in Burlington, MA, on August 14, 2003.

Marc J. Bouthillier,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 03-21412 Filed 8-20-03; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NE-32-AD; Amendment 39-13285; AD 2003-17-10]

RIN 2120-AA64

Airworthiness Directives; McCauley Propeller Systems, Inc. Propeller Hub Models B5JFR36C1101, C5JFR36C1102, B5JFR36C1103, and C5JFR36C1104

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is superseding an existing airworthiness directive (AD) for McCauley Systems, Inc. propellers that are installed on, but not limited to, BAE Systems (Operations) Limited Jetstream Model 4101 airplanes. That AD currently requires a one-time fluorescent penetrant inspection (FPI) of propeller blade retention areas for cracks. This AD requires initial and repetitive FPI or Ultrasonic Inspection (UT) of propeller blade retention areas for cracks, replacement of high time propeller blades, and a one-time inspection of propeller hubs. This AD is prompted by four reports of significant cracks found in propeller blade shanks since the issuance of AD 2003-15-01. We are issuing this AD to prevent propeller blade failure or hub failure due to cracking, which could result in failure of the propeller and loss of control of the airplane.

DATES: Effective August 21, 2003. The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of August 21, 2003.

We must receive any comments on this AD by October 20, 2003.

ADDRESSES: Use one of the following addresses to submit comments on this AD:

- By mail: Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003-NE-32-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

- By fax: (781) 238-7055.

- By e-mail: 9-ane-adcomment@faa.gov.

You can get the service information referenced in this AD from McCauley Propeller Systems, 3535 McCauley Drive, Vandalia, OH 45377.

You may examine the AD docket by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA. You may examine the service information at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Timothy Smyth, Aerospace Engineer, Chicago Aircraft Certification Office, FAA, Small Airplane Directorate, 2300 East Devon Avenue, Room 107, Des Plaines, IL 60018; telephone: (847) 294-7132; fax: (847) 294-7834.

SUPPLEMENTARY INFORMATION: On July 14, 2003, the FAA issued AD 2003-15-01, Amendment 39-13243 (68 FR 42244, July 17, 2003). That AD applies to the following McCauley Systems, Inc. propeller assemblies that are installed on, but not limited to, BAE Systems (Operations) Limited Jetstream Model 4101 airplanes:

- Hub Model B5JFR36C1101, with Model 114GC series propeller blades.
- Hub Model C5JFR36C1102, with Model L114GC series propeller blades.
- Hub Model B5JFR36C1103, with Model 114HC series propeller blades.
- Hub Model C5JFR36C1104, with Model L114HC series propeller blades.

That AD requires a one-time FPI of propeller blade retention areas for cracks. That AD was prompted by a report of a significant crack in a propeller blade shank and two reports of cracks in the hubs of the same propeller model. That condition, if not corrected, could result in a failure of the propeller blade or hub and loss of control of the airplane.

Actions Since AD 2003-15-01 Was Issued

Since that AD was issued, four additional propeller blade cracks have been reported. Based on examination of these cracked propeller blades, a repetitive inspection interval has been established.

Relevant Service Information

We have reviewed and approved the technical contents of McCauley Alert Service Bulletin (ASB) 246C, Revision 3, dated August 12, 2003, that describes procedures for FPI or UT of propeller

blades. We have also reviewed and approved the technical contents of McCauley ASB245A, Revision 1, dated August 13, 2003, that describes procedures for a one-time eddy current inspection of propeller hubs.

Differences Between This AD and the Service Information

McCauley ASB246C, Revision 3, dated August 12, 2003, requires the operator to perform a blade shake check at 72-hour intervals. This AD does not require the blade shake check. McCauley ASB245 A, Revision 1, dated August 13, 2003, requires initial and repetitive hub inspections. This AD does not require repetitive hub inspections, but does require a different compliance interval.

FAA's Determination and Requirements of This AD

The unsafe condition described previously is likely to exist or develop on other McCauley Systems, Inc. propeller hub Models B5JFR36C1101, C5JFR36C1102, B5JFR36C1103, and C5JFR36C1104, of the same type design. We are issuing this AD to prevent propeller blade failure or hub failure due to cracking, which could result in failure of the propeller and loss of control of the airplane. You must use the service information described previously to perform these actions.

FAA's Determination of the Effective Date

Since an unsafe condition exists that requires the immediate adoption of this AD, we have found that notice and opportunity for public comment before issuing this AD are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Changes to 14 CFR Part 39—Effect on the AD

On July 10, 2002, we issued a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs our AD system. This regulation now includes material that relates to special flight permits, alternative methods of compliance, and altered products. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Interim Action

These actions are interim actions and we may take further rulemaking actions in the future.