email_subscription/ and allows FSIS customers to sign up for subscription options across eight categories. Options range from recalls to export information to regulations, directives and notices. Customers can add or delete subscriptions themselves and have the option to password protect their account.

List of Subjects in 9 CFR Part 327

Imported Products.

■ For the reasons set out in the preamble, FSIS is amending 9 CFR part 327 as follows:

PART 327—IMPORTED PRODUCTS

■ 1. The authority citation for part 327 continues to read as follows:

Authority: 21 U.S.C. 601–695; 7 CFR 2.18, 2.53.

§ 327.2 [Amended]

■ 2. Section 327.2 is amended by adding Chile in alphabetical order to the list of countries in paragraph (b).

Done at Washington, DC, on: October, 11, 2005.

Barbara I. Masters.

Administrator.

[FR Doc. 05–22980 Filed 11–18–05; 8:45 am] $\tt BILLING\ CODE\ 3410-DM-P$

FARM CREDIT ADMINISTRATION

12 CFR Part 627

RIN 3052-AC26

Title IV Conservators, Receivers, and Voluntary Liquidations; Receivership Repudiation Authorities; Effective Date

AGENCY: Farm Credit Administration. **ACTION:** Notice of effective date.

SUMMARY: The Farm Credit Administration (FCA) published a final rule under part 627 on September 22, 2005 (70 FR 55513). This final rule gives certainty to Farm Credit System institutions regarding how the Farm Credit System Insurance Corporation will treat qualifying participations and securitizations if the institution is subsequently placed in conservatorship or receivership. In accordance with 12 U.S.C. 2252, the effective date of the final rule is 30 days from the date of publication in the Federal Register during which either or both Houses of Congress are in session. Based on the records of the sessions of Congress, the effective date of the regulations is November 14, 2005.

DATES: The regulation amending 12 CFR part 627, published on September 22,

2005 (70 FR 55513) is effective November 14, 2005.

FOR FURTHER INFORMATION CONTACT:

Robert E. Donnelly, Senior Accountant, Office of Policy and Analysis, Farm Credit Administration, McLean, VA 22102–5090, (703) 883–4498, TTY (703) 883–4434; or Rebecca S. Orlich, Senior Attorney, Office of General Counsel, Farm Credit Administration, McLean, VA 22102–5090, (703) 883–4020, TTY (703) 883–4020.

(12 U.S.C. 2252(a)(9) and (10))

Dated: November 15, 2005.

Jeanette C. Brinkley,

Secretary, Farm Credit Administration Board. [FR Doc. 05–23001 Filed 11–18–05; 8:45 am] BILLING CODE 6705–01–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2005-23005; Directorate Identifier 2003-NM-110-AD; Amendment 39-14379; AD 2005-23-21]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A340–200 and A340–300 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Airbus Model A340–200 and A340–300 series airplanes. This AD requires repetitive inspections for cracking of the junction between the thrust reverser forward frame and the 12 o'clock and 6 o'clock beams; the thrust reverser common fittings; and the thrust reverser J-ring structure at the 12 o'clock and 6 o'clock positions; and related investigative and corrective actions. This AD results from fatigue and damage tolerance testing of the engine thrust reversers which revealed fatigue cracking of the junction between the thrust reverser forward frame and the 12 o'clock beam, and of the thrust reverser J-ring structure. We are issuing this AD to detect and correct such fatigue cracking, which could result in rupture of the J-ring, forward frame, or common fittings; loss of the thrust reverser system; or inadvertent deployment of a thrust reverser in flight and consequent reduced controllability of the airplane.

DATES: This AD becomes effective December 6, 2005.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of December 6, 2005.

We must receive comments on this AD by January 20, 2006.

ADDRESSES: Use one of the following addresses to submit comments on this AD

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, room PL-401, Washington, DC 20590.
 - Fax: (202) 493–2251.
- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; or Rohr Inc., 850 Lagoon Drive, Chula Vista, California 91912; for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, notified us that an unsafe condition may exist on all Airbus Model A340-200 and A340-300 series airplanes. The DGAC advises that fatigue and damage tolerance testing of the engine thrust reversers revealed cracking of the junction between the thrust reverser forward frame and the 12 o'clock beam. The cracking of the junction was found after the accumulation of approximately 20,000 total flight cycles. In addition to the fatigue cracking that was found during fatigue tests, fatigue cracking of the J-ring was also found on in-service airplanes that had accumulated fewer than 6,666 total flight cycles. These conditions, if not corrected, could result in rupture of the J-ring, forward frame, or common fittings; loss of the thrust reverser system; or inadvertent deployment of a thrust reverser in flight