

report to Congress, as required by 5 U.S.C. 5757(d). Before February 15, 2006, each agency must submit a written report to OPM on—

(1) The agency's use of extended assignment incentives by providing the data required in paragraph (c) of this section;

(2) Whether the use of extended assignment incentives influenced employees to stay longer than their initial tour of duty at their current duty stations; and

(3) The agency's recommendations for changes necessary to improve the effectiveness of extended assignment incentives.

(c) Each agency report must contain the following data for the period from May 2, 2003, to December 31, 2005:

(1) The number of extended assignment service agreements that commenced in each fiscal year;

(2) The dollar amount expended on extended assignment incentives in each fiscal year;

(3) The number of employees who declined an extended assignment incentive, by occupational series and geographic location;

(4) The number of employees who signed an extended assignment incentive service agreement, the total amount of the planned incentives, and the total number of years of agreed-upon service, by occupational series and geographic location;

(5) The number of employees whose service agreements were terminated before completion of the agreed-upon service period, with subcounts showing the number covered by § 575.511, § 575.512, and § 575.513, respectively.

(6) The number of employees who incurred a repayment debt under § 575.513 (including any repayment penalty under § 575.513(e)) and the total amount of repayment debt incurred; and

(7) The portion of the repayment debt that, as of December 31, 2005—

(i) Has been recovered;

(ii) Is subject to ongoing collection efforts; and

(iii) Has been waived or written off.

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FEDERAL RESERVE SYSTEM

12 CFR Part 229

[Regulation CC; Docket No. R-1160]

Availability of Funds and Collection of Checks

AGENCY: Board of Governors of the Federal Reserve System.

ACTION: Final rule; correction.

SUMMARY: On September 2, 2003, the Board of Governors published in the **Federal Register** a final rule amending appendix A of Regulation CC. The rule removed the reference to the Pittsburgh check processing office of the Federal Reserve Bank of Cleveland and reassigned the Federal Reserve routing symbols currently listed under that office to the head office of the Federal Reserve Bank of Cleveland. The rule also replaced all references to Thomson Financial Publishing Inc., in appendices A and E with more general references to "an agent of the American Bankers Association." This document corrects the amendatory instructions for the removal of the reference to Thomson Financial Publishing Inc., in appendix E. The original amendatory instruction would have caused the amended sentence in section II.DD. to contain duplicative references to the American Bankers Association agency arrangement.

DATES: The correction is effective on November 1, 2003 (*i.e.*, the effective date of the final rule).

FOR FURTHER INFORMATION CONTACT: Adrienne G. Threatt, Counsel (202/452-3554), Legal Division. For users of Telecommunications Devices for the Deaf (TDD) only, contact 202/263-4869.

SUPPLEMENTARY INFORMATION: In the final rule, FR Doc. 03-22333 published on September 2, 2003, make the following correction:

Appendix E to Part 229—[Corrected]

■ On page 52078, in the first and second columns, correct amendatory language in amendment 3. to read as follows:

3. Appendix E is amended in section II.DD. by removing the phrase "Thomson Financial Publishing Inc., as agent for" and adding the phrase "an agent of" in its place, and in sections XVIII.A.2.b.ii. and XXII.B.2.b.i. by removing the phrase "Thomson Financial Publishing Inc." and adding the phrase "an agent of the American Bankers Association" in its place.

By order of the Board of Governors of the Federal Reserve System, September 8, 2003.

Jennifer J. Johnson,

Secretary of the Board.

[FR Doc. 03-23239 Filed 9-11-03; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. NM254; Special Conditions No. 25-246-SC]

Special Conditions: Cessna Model 680 Sovereign; Side-Facing Single-Occupant Seats

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions.

SUMMARY: These special conditions are issued for the Cessna Model 680 Sovereign airplane. This airplane will have a novel or unusual design feature(s) associated with side-facing single-occupant seats. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

EFFECTIVE DATE: October 14, 2003.

FOR FURTHER INFORMATION CONTACT: Mark Quam, FAA, Standardization Branch, ANM-113, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue SW., Renton, Washington, 98055-4056; telephone (425) 227-2145, facsimile (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Background

On November 24, 1999, Cessna Aircraft Company, One Cessna Boulevard, Wichita, KS 67277, applied for type certificate for their new Cessna Model 680 Sovereign airplane. The Model 680 Sovereign is a twin-engine pressurized executive jet airplane with standard seating provisions for 12 passenger/crew and allowance for baggage and optional equipment. This airplane will have a maximum takeoff weight of 30,000 pounds with a wingspan of 63.1 feet and will have two aft-mounted Pratt & Whitney 306C engines.

The Cessna Model 680 offers interior arrangements, which include single-occupant side-facing seat installations. These seats are installed on the LH and RH side of the cabin's forward section, forward of and opposite to the entry door respectively. Dynamic testing of all seats approved for occupancy during takeoff and landing is required by 14 CFR 25.562. The pass/fail criteria for the testing developed in Amendment 25-64

to § 25.562 focused primarily on fore/aft-facing seats. Side-facing seating installations were not adequately addressed for transport category airplanes in this Amendment.

These special conditions are applicable to single-occupant side-facing seats only. They are not intended to be used for multiple-occupant side-facing divans or sofas, as they do not account for possible interaction among the occupants.

Type Certification Basis

Under the provisions of 14 CFR 21.17, the Cessna Aircraft Company must show that the Model 680 Sovereign airplane meets the applicable provisions of 14 CFR part 25, effective February 1, 1965, as amended by Amendments 25–1 through 25–98; 14 CFR part 34, effective September 10, 1990, as amended by any amendment in effect on the date of certification.

If the Administrator finds that the applicable airworthiness regulations (*i.e.*, 14 CFR part 25) do not contain adequate or appropriate safety standards for the Cessna Model 680 Sovereign airplane because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

In addition to the applicable airworthiness regulations and special conditions, the Cessna Model 680 Sovereign must comply with the fuel vent and exhaust emission requirements of 14 CFR part 34 and the noise certification requirements of 14 CFR part 36, and the FAA must issue a finding of regulatory adequacy pursuant to § 611 of Public Law 92–574, the “Noise Control Act of 1972.”

Special conditions, as defined in 14 CFR 11.19, are issued in accordance with § 11.38 and become part of the type certification basis in accordance with § 21.17(a)(2).

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design feature, the special conditions would also apply to the other model.

Novel or Unusual Design Features

The Cessna Model 680 offers interior arrangements, which include single-occupant side-facing seat installations. These seats are installed on the LH and RH side of the cabin’s forward section, forward and opposite to the entry door respectively. Dynamic testing of all seats approved for occupancy during takeoff and landing is required by § 25.562. The pass/fail criteria for the testing

developed in Amendment 25–64 to § 25.562 focused primarily on fore/aft-facing seats. Side-facing seating installations were not adequately addressed for transport category airplanes in this Amendment.

Discussion

The following injury criteria and testing represent the minimum acceptable standards for certification of the Model 680 single-occupant side-facing seats, and are to be included as requirements in these special conditions. However, the existing requirements call for a “no yaw” test condition. Cessna will demonstrate values of thoracic trauma index (TTI) and lateral pelvic acceleration (LPA) for a “10 degree yaw” for which it has some test data. In this case, Cessna must show the “10 degree yaw” yields results that will only differ slightly from the “no yaw” condition and that these differences would not be of such magnitude as to exceed the maximum allowable.

Proposed Injury Criteria

(a) Existing Criteria. As referenced by § 25.785(b), all injury protection criteria of §§ 25.562(c)(1) through (c)(6) apply to the occupants of the single-occupant side-facing seats. Head injury criteria (HIC) assessments are only required for head contact with the seat and/or adjacent structures.

(b) Body-to-wall/furnishing contact. The seat must be installed aft of a structure such as an interior wall or furnishing that will contact the pelvis, upper arm, chest, or head of an occupant seated next to the structure. A conservative representation of the structure and its stiffness must be included in the tests. It is recommended, but not required, that the contact surface of this structure be covered with at least two inches of energy absorbing protective padding (foam or equivalent), such as Ensolite.

(c) Thoracic Trauma. Testing with a Side Impact Dummy (SID), as defined by 49 CFR Part 572, Subpart F, or its equivalent, must be conducted and TTI injury criteria acquired with the SID must be less than 85, as defined in 49 CFR Part 572, Subpart F. Side Impact Dummy TTI must be processed as defined in Federal Motor Vehicle Safety Standard (FMVSS) Part 571.214, section S6.13.5. Rational analysis, comparing an installation with another installation where TTI data were acquired and found acceptable, may also be viable.

(d) Pelvis. Pelvic lateral acceleration must not exceed 130g, pelvic acceleration data must be processed as

defined in FMVSS Part 571.214, section S6.13.5

(f) Shoulder Strap Loads. Where upper torso straps (shoulder straps) are used for sofa occupants, tension loads in individual straps must not exceed 1,750 pounds. If dual straps are used for restraining the upper torso, the total strap tension loads must not exceed 2,000 pounds.

Discussion of Comments

Notice of Proposed Special Conditions No. 25–03–03–SC for the Cessna Model 680 Sovereign airplane was published in the **Federal Register** on May 15, 2003 (68 FR 26237). No comments were received.

Applicability

As discussed above, these special conditions are applicable to the Cessna Model 680 Sovereign airplane. Should Cessna Aircraft Company apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design feature, these special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features on the Cessna Model 680 Sovereign airplane. It is not a rule of general applicability, and it affects only the applicant who applied to the FAA for approval of these features on the airplane.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

■ The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for the Cessna Aircraft Company Model 680 Sovereign airplane.

The minimum acceptable standards of injury criteria and testing requirements for dynamic certification of the Model 680 side-facing single-occupant seats are as follows:

(a) Existing Criteria. As referenced by § 25.785(b), all injury protection criteria of §§ 25.562(c)(1) through (c)(6) apply to the occupants of the side-facing seats. Head injury criteria (HIC) assessments are only required for head contact with the seat and/or adjacent structures.

(b) Body-to-wall/furnishing contact. The seat must be installed aft of a structure such as an interior wall or

furnishing that will contact the pelvis, upper arm, chest, or head of an occupant seated next to the structure. A conservative representation of the structure and its stiffness must be included in the tests. It is recommended, but not required, that the contact surface of this structure must be covered with at least two inches of energy absorbing protective padding (foam or equivalent), such as Ensolute.

(c) Thoracic Trauma. Testing with a Side Impact Dummy (SID), as defined by 49 CFR Part 572, Subpart F, or its equivalent, must be conducted and TTI injury criteria acquired with the SID must be less than 85, as defined in 49 CFR Part 572, Subpart F. SID TTI must be processed as defined in Federal Motor Vehicle Safety Standard (FMVSS) Part 571.214, section S6.13.5. Rational analysis, comparing an installation with another installation where TTI data were acquired and found acceptable, may also be viable.

(d) Pelvis. Pelvic lateral acceleration must not exceed 130g, pelvic acceleration data must be processed as defined in FMVSS Part 571.214, section S6.13.5.

(f) Shoulder Strap Loads. Where upper torso straps (shoulder straps) are used for occupants, tension loads in individual straps must not exceed 1,750 pounds. If dual straps are used for restraining the upper torso, the total strap tension loads must not exceed 2,000 pounds.

Issued in Renton, Washington, on September 3, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-23293 Filed 9-11-03; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2003-16026; Airspace Docket No. 03-ACE-70]

Modification of Class D Airspace; and Modification of Class E Airspace; St. Joseph, MO

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Direct final rule; request for comments.

SUMMARY: Area Navigation (RNAV) Global Positioning System (GPS) Standard Instrument Approach Procedures (SIAPs) have been developed to serve Rosecrans Memorial

Airport, St. Joseph, MO. Also, several existing SIAPs serving Rosecrans Memorial Airport have been amended. An examination of controlled airspace for St. Joseph, MO revealed discrepancies in the legal descriptions for the St. Joseph, MO Class D and Class E airspace areas.

The intended effect of this rule is to provide controlled airspace for appropriate dimensions to protect aircraft executing SIAPs to Rosecrans Memorial Airport. It also corrects discrepancies in the legal descriptions of St. Joseph, MO Class D and Class E airspace areas and brings the airspace areas and legal descriptions into compliance with FAA Orders.

DATES: This direct final rule is effective on 0901 UTC, December 25, 2003. Comments for inclusion in the Rules Docket must be received on or before October 23, 2003.

ADDRESSES: Send comments on this proposal to the Docket Management System, U.S. Department of Transportation, Room Plaza 401, 400 Seventh Street, SW., Washington, DC 20590-0001. You must identify the docket number FAA-2003-16026/Airspace Docket No. 03-ACE-70, at the beginning of your comments. You may also submit comments on the Internet at <http://dms.dot.gov>. You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1-800-647-5527) is on the plaza level of the Department of Transportation NASSIF Building at the above address.

FOR FURTHER INFORMATION CONTACT: Brenda Mumper, Air Traffic Division, Airspace Branch, ACE-520A, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329-2524.

SUPPLEMENTARY INFORMATION: This amendment to 14 CFR 71 modifies the Class D airspace area, the Class E airspace area designated as an extension to the Class D airspace area and the Class E airspace area extending upward from 700 feet above the surface of the earth at St. Joseph, MO. RNAV (GPS) RWY 35, ORIGINAL SIAP; RNAV (GPS) RWY 17, ORIGINAL SIAP; VOR/DME or TACAN RWY 35, ORIGINAL SIAP; ILS or LOC RWY 35, AMENDMENT 31 SIAP; VOR or TACAN RWY 17, AMENDMENT 14 SIAP; LOC BC RWY 17, AMENDMENT 9 SIAP and NDB RWY 35, AMENDMENT 28F SIAP have been developed to serve Rosecrans Memorial Airport, St. Joseph, MO. The

St. Joseph, MO controlled airspace areas must be tailored to contain aircraft executing the approach procedures. An examination of controlled airspace for St. Joseph, MO revealed discrepancies in the legal descriptions for the St. Joseph, MO Class D and Class E airspace areas. This action corrects the discrepancies and brings the airspace areas and their legal descriptions into compliance with FAA Order 7400.2E, Procedures for Handling Airspace Matters. The areas will be depicted on appropriate aeronautical charts. Class D airspace areas are published in paragraph 5000 of FAA Order 7400.9K, dated August 30, 2002, and effective September 16, 2002, which is incorporated by reference in 14 CFR 71.1. Class E airspace areas designated as an extension to a Class D or Class E surface area and Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in paragraphs 6004 and 6005 respectively of the same FAA Order. The Class D and Class E airspace designations listed in this document will be published subsequently in the Order.

The Direct Final Rule Procedure

The FAA anticipates that this regulation will not result in adverse or negative comment and, therefore, is issuing it as a direct final rule. Previous actions of this nature have not been controversial and have not resulted in adverse comments or objections. Unless a written adverse or negative comment, or a written notice of intent to submit an adverse or negative comment is received within the comment period, the regulation will become effective on the date specified above. After the close of the comment period, the FAA will publish a document in the **Federal Register** indicating that no adverse or negative comments were received and confirming the date on which the final rule will become effective. If the FAA does receive, within the comment period, an adverse or negative comment, or written notice of intent to submit such a comment, a document withdrawing the direct final rule will be published in the **Federal Register**, and a notice of proposed rulemaking may be published with a new comment period.

Comments Invited

Interested parties are invited to participate in this rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory