

of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the front spar attachment and upper cap of the engine pylons due to fatigue cracking, and consequent reduced structural integrity of the airplane, accomplish the following:

Repetitive Inspections

(a) Prior to the accumulation of 9,800 total flight hours, or within the next 1,800 flight hours after August 23, 1977 (the effective date of AD 77-14-19, amendment 39-2971), whichever occurs later, unless accomplished previously within the last 1,800 flight hours, accomplish the radiographic and ultrasonic or eddy current inspections in accordance with the instructions in Douglas Service Bulletin 54-30, dated January 19, 1977. Repeat the inspection thereafter at intervals not to exceed 3,600 flight hours. For those operators who have conducted only the radiographic inspections in accordance with Douglas All Operators Letter AOL 9-835, dated October 30, 1974, perform the ultrasonic or eddy current inspections, and thereafter, the radiographic and ultrasonic or eddy current inspection in accordance with the requirements of this AD, as applicable.

Note 2: Inspections accomplished prior to the effective date of this AD in accordance with McDonnell Douglas Service Bulletin 54-30, Revision 1, dated June 29, 1977, Revision 2, dated October 27, 1978, Revision 3, dated April 30, 1986, or Revision 4, dated March 25, 1991; or McDonnell Douglas Service Bulletin DC9-54-030, Revision 05, dated August 26, 1999, or Revision 06, dated November 11, 1999; are considered acceptable for compliance with the inspections required by paragraph (a) of this AD.

Initial Inspections and Follow-On/Corrective Action

(b) For airplanes on which the modification specified in paragraph (e) of this AD has not been accomplished: Prior to the accumulation of 8,000 total flight hours or within 3,600 flight hours after the effective date of this AD, whichever occurs later, perform x-ray and eddy current inspections to detect cracks of the upper cap of the front spar of the left and right engine pylons, in accordance with McDonnell Douglas Service Bulletin DC9-54-030, Revision 06, dated November 11, 1999. Accomplishment of these inspections constitutes terminating action for the repetitive inspection requirements of paragraph (a) of this AD.

No Crack Detected: Repetitive Inspections

(c) If no crack is detected during any inspection required by paragraph (a) or (b) of this AD, repeat the inspections thereafter at intervals not to exceed 3,600 flight hours

until the modification required by paragraph (e) of this AD is accomplished.

Any Crack Detected: Modification

(d) If any crack is detected during any inspection required by paragraph (a) or (b) of this AD, prior to further flight, accomplish the modification specified in paragraph (e) of this AD.

Terminating Modification

(e) Except as provided by paragraph (d) of this AD, prior to the accumulation of 100,000 total landings, or within 6 months after the effective date of this AD, whichever occurs later, modify the upper cap of the front spar of the left and right engine pylons in accordance with McDonnell Douglas Service Bulletin DC9-54-030, Revision 06, dated November 11, 1999. Accomplishment of this modification constitutes terminating action for the requirements of this AD.

Note 3: Accomplishment of the modification of the upper cap of the front spar of the left and right engine pylons prior to the effective date of this AD in accordance with Douglas Service Bulletin 54-30, Revision 4, dated March 25, 1991, or McDonnell Douglas Service Bulletin DC9-54-030, Revision 5, dated August 26, 1999; is considered acceptable for compliance with the modification specified in paragraph (e) of this AD.

Alternative Methods of Compliance

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 4: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permits

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(h) The actions specified in paragraphs (b) and (e) of this AD shall be done in accordance with McDonnell Douglas Service Bulletin DC9-54-030, Revision 06, dated November 11, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1-L51 (2-60). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960

Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(i) This amendment becomes effective on December 13, 2000.

Issued in Renton, Washington, on October 30, 2000.

Donald L. Riggins,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-28234 Filed 11-7-00; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-152-AD; Amendment 39-11963; AD 2000-22-16]

RIN 2120-AA64

Airworthiness Directives; British Aerospace (Jetstream) Model 4101 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all British Aerospace (Jetstream) Model 4101 airplanes, that requires a functional check of the rudder pedals to ensure full and free movement at all rudder pedal positions, and modification of the forward rudder pedal boxes. The actions specified by this AD are intended to prevent restricted movement of the rudder pedals due to structural interference, which could result in reduced controllability of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective December 13, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of December 13, 2000.

ADDRESSES: The service information referenced in this AD may be obtained from British Aerospace Regional Aircraft American Support, 13850 Mclearen Road, Herndon, Virginia 20171. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all British Aerospace (Jetstream) Model 4101 airplanes was published in the **Federal Register** on August 29, 2000 (65 FR 52373). That action proposed to require a functional check of the rudder pedals to ensure full and free movement at all rudder pedal positions, and modification of the forward rudder pedal boxes.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 58 British Aerospace (Jetstream) Model 4101 airplanes of U.S. registry will be affected by this AD.

It will take approximately 1 work hour per airplane to accomplish the functional check, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$3,480, or \$60 per airplane.

It will take approximately 6 work hours per airplane to accomplish the modification, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$20,880, or \$360 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator will accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures

typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000-22-16 British Aerospace Regional Aircraft (Formerly Jetstream Aircraft Limited; British Aerospace (Commercial Aircraft) Limited): Amendment 39-11963. Docket 2000-NM-152-AD.

Applicability: All Model Jetstream 4101 airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent restricted movement of the rudder pedals due to structural interference, which could result in reduced controllability of the airplane, accomplish the following:

Functional Check

(a) Within 60 days after the effective date of this AD, perform a functional check of the left and right rudder pedals to ensure full and free movement at all rudder pedal positions, in accordance with Part 1 of the Accomplishment Instructions of Jetstream Alert Service Bulletin J41-A27-055, dated March 10, 2000. If any restriction in rudder pedal movement is found, prior to further flight, accomplish the modification required by paragraph (b) of this AD.

Modification

(b) Within 8 months after the effective date of this AD, modify the forward pedal boxes (including moving the rudder pedals and measuring clearances between the rod attachment bolt and the flange of the floor channel; correcting any incorrect clearances; and repeating the functional check of the rudder pedals specified in paragraph (a) of this AD) in accordance with Part 2 of the Accomplishment Instructions of Jetstream Alert Service Bulletin J41-A27-055, dated March 10, 2000.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) The actions shall be done in accordance with Jetstream Alert Service Bulletin J41–A27–055, dated March 10, 2000. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from British Aerospace Regional Aircraft American Support, 13850 Mclearen Road, Herndon, Virginia 20171. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in British airworthiness directive 002–03–2000.

Effective Date

(f) This amendment becomes effective on December 13, 2000.

Issued in Renton, Washington, on October 30, 2000.

Donald L. Rigglin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00–28232 Filed 11–7–00; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF LABOR**Mine Safety and Health Administration****30 CFR Part 62**

RIN 1219–AA53

Health Standards for Occupational Noise Exposure; Correction

AGENCY: Mine Safety and Health Administration (MSHA), Labor.

ACTION: Correcting amendment.

SUMMARY: The Mine Safety and Health Administration published in the **Federal Register** of Monday, September 13, 1999, (64 FR 49548) the final rule relating to the health standards for occupational noise exposure. This document contains a correction to that document.

DATES: Effective on November 8, 2000.

FOR FURTHER INFORMATION CONTACT: Carol J. Jones, Director, Office of Standards, Regulations, and Variances, MSHA, (703) 235–1910 (not a toll-free call).

SUPPLEMENTARY INFORMATION: As published, the final rule contains an error which needs to be corrected.

Accordingly, 30 CFR part 62 is corrected by making the following correcting amendment:

PART 62—OCCUPATIONAL NOISE EXPOSURE

1. The authority citation for part 62 is revised to read as follows:

Authority: 30 U.S.C. 811.

Appendix to Part 62 [Amended]

2. In the Appendix to part 62, in the note to Table 62–1, the formula for T is revised to read as follows:

$$T = 8/2(L-90)/5$$

Dated: November 1, 2000.

J. Davitt McAteer,

Assistant Secretary for Mine Safety and Health.

[FR Doc. 00–28586 Filed 11–7–00; 8:45 am]

BILLING CODE 4510–43–P

DEPARTMENT OF THE INTERIOR**Office of Surface Mining Reclamation and Enforcement****30 CFR Part 920**

[MD–047–FOR]

Maryland Regulatory Program

AGENCY: Office of Surface Mining Reclamation and Enforcement (OSM), Interior.

ACTION: Final rule.

SUMMARY: OSM is approving an amendment to the Maryland regulatory program under the Surface Mining Control and Reclamation Act of 1977 (SMCRA). Maryland proposed the amendment to make its program no less effective than the federal regulations regarding procedures for financing abandoned mine land reclamation projects that involve the incidental extraction of coal. The amendment is intended to revise the Maryland program to be consistent with the corresponding Federal regulations and SMCRA.

EFFECTIVE DATE: November 8, 2000.

FOR FURTHER INFORMATION CONTACT: George Rieger, Program Manager, OSM, Appalachian Regional Coordinating Center, 3 Parkway Center, Pittsburgh, PA 15220. Telephone: (412) 937–2153; E-mail: grieger@osmre.gov.

SUPPLEMENTARY INFORMATION:

- I. Background on the Maryland Program
- II. Submission of the Amendment
- III. Director's Findings
- IV. Summary and Disposition of Comments
- V. Director's Decision

VI. Procedural Determinations**I. Background on the Maryland Program**

On February 18, 1982, the Secretary of the Interior approved the Maryland program. You can find background information on the Maryland program, including the Secretary's findings, the disposition of comments, and the conditions of approval in the February 18, 1982, **Federal Register** (47 FR 7214). You can find subsequent actions concerning conditions of approval and program amendments at 30 CFR 920.12, 920.15, and 920.16.

II. Submission of the Amendment

By letter dated July 10, 2000 (Administrative Record No. MD–582–00), Maryland submitted the proposed amendment to its regulatory program pursuant to the federal regulations at 30 CFR 732.17(b). Maryland proposed the amendment to make its program no less effective than the federal regulations at 30 CFR 707.5, 707.10, 874.10, and 874.17. These sections of the federal regulations describe procedures for financing abandoned mine land reclamation projects that involve the incidental extraction of coal. Maryland proposed to change the definition of the term, “Government-Financed Construction” at Code of Maryland Regulation (COMAR) 26.20.12.02 B(1)(a) by adding the phrase, “Funding at less than 50 percent may qualify if the construction is undertaken as an approved reclamation project under Environment Article, Title 15, Subtitle 11 Annotated Code of Maryland and 30 CFR Subchapter R.”

Maryland also added new section .04 to COMAR 26.20.12. This section is titled, “Government Funded Reclamation Projects.” The proposed rulemaking was published in the August 14, 2000, **Federal Register** (65 FR 49524). The public comment period closed on September 13, 2000. No one requested an opportunity to speak at a public hearing, so no hearing was held.

III. Director's Findings

Set forth below, pursuant to SMCRA and the federal regulations at 30 CFR 732.15 and 732.17, are the Director's findings concerning the amendments to the Maryland regulatory program.

The first change Maryland is making to its program is the modification of the definition of the term, “Government-Financed Construction” at COMAR 26.20.12.02 B(1)(a). Maryland added the