

herein, as may be requested by FSA. Such records and accounts must be retained for 2 years after the date of the final payment to the producer under this program.

(b) At all times during regular business hours, authorized representatives of FSA, the United States Department of Agriculture, or the Comptroller General of the United States shall have access to the premises of the producer in order to inspect, examine, and make copies of the books, records, and accounts, and other written data as specified in paragraph (a) of this section.

(c) Audits of certifications of average adjusted gross income may be conducted as necessary to determine compliance with the requirements of this subpart. As a part of this audit, income tax forms may be requested and if requested, must be supplied. If a producer has submitted information to FSA, including a certification from a certified public accountant or attorney, that relied upon information from a form previously filed with the Internal Revenue Service, such producer shall provide FSA a copy of any amended form filed with the Internal Revenue Service with 30 days of the filing.

(d) If requested in writing by FSA, the United States Department of Agriculture, or the Comptroller General of the United States, the producer shall provide all information and documentation the reviewing authority determines necessary to verify any information or certification provided under this subpart, including all documents referred to in § 1580.301(c), within 30 days. Acceptable production documentation may be submitted by facsimile, in person, or by mail and may include copies of receipts, ledgers, income statements, deposit slips, register tapes, invoices for custom harvesting, records to verify production costs, contemporaneous measurements, truck scale tickets, and contemporaneous diaries that are determined acceptable by the county committee. Failure to provide necessary and accurate information to verify compliance, or failure to comply with this subpart's requirements, will result in ineligibility for all program benefits subject to this subpart for the year or years subject to the request.

(e) All information provided to FSA for the purposes of determining compliance with this part will remain confidential and not be subject to any request submitted under the Freedom of Information Act.

§ 1580.503 Debarment and suspension.

The Government-wide Debarment and Suspension (Nonprocurement) regulations and Government Requirements for Drug-Free Workplace (Grants), 7 CFR part 3017—subparts A through E, apply to this part.

§ 1580.504 Fraud and recovery of overpayments.

(a) If the Administrator, FSA or a court of competent jurisdiction, determines that any person has received any payment under this program to which the person was not entitled, such person will be liable to repay such amount to the Administrator, FSA. The Administrator, FSA may waive such repayment if it is determined that:

(1) The payment was made without fault on the part of the person; and
(2) Requiring such repayment would be contrary to equity and good conscience.

(b) Unless an overpayment is otherwise recovered, or waived under paragraph (a), the Administrator, FSA shall recover the overpayment by deductions from any sums payable to such person.

(c) If the Administrator, FSA, or a court of competent jurisdiction, determines that a person:

(1) Knowingly has made, or caused another to make, a false statement or representation of a material fact, or
(2) Knowingly has failed, or caused another to fail, to disclose a material fact, and, as a result of such false statement or representation, or of such nondisclosure, such person has received any payment under this program to which the person was not entitled, such person shall, in addition to any other penalty provided by law, be ineligible for any further payment under this program.

(d) Except for overpayments determined by a court of competent jurisdiction, no repayment may be required, and no deduction may be made, under this section until a determination and an opportunity for a fair hearing has been given to the person concerned, and the determination has become final.

(e) Whoever makes a false statement of a material fact knowing it to be false, or knowingly fails to disclose a material fact, for the purpose of obtaining or increasing for himself or for any other person any payments authorized to be furnished under this program shall be fined not more than \$10,000 or imprisoned for not more than 1 year, or both.

§ 1580.505 Appeals.

Any person may obtain reconsideration and review of

determinations made with respect to applications for program benefits under this part in accordance with appeal regulations of the 7 CFR part 780.

§ 1580.601 Implementation.

Trade adjustment assistance is available for the most recent marketing year for which prices were available on February 3, 2003.

§ 1580.602 Paperwork Reduction Act assigned number.

The information collection requirements contained in these regulations (7 CFR part 1580) have been approved by the Office of Management and Budget under the provisions of 44 U.S.C. Chapter 35 and been assigned OMB control number xxxx-xxxx.

Dated: June 27, 2003.

A. Ellen Terpstra,

Administrator, Foreign Agricultural Service.

[FR Doc. 03-16812 Filed 7-1-03; 8:45 am]

BILLING CODE 3410-10-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-149-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747-100, -200B, and -200F Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Boeing Model 747-100, -200B, and -200F series airplanes. This proposal would require initial and repetitive inspections to find discrepancies in the upper and lower skins of the fuselage lap joints, and repair if necessary. This action is necessary to find and fix such discrepancies, which could result in sudden fracture and failure of a lap joint and rapid in-flight decompression of the airplane fuselage. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by August 18, 2003.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-

149-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address:

9-anm-nprmcomment@faa.gov.

Comments sent via fax or the Internet must contain "Docket No. 2002-NM-149-AD" in the subject line and need not be submitted in triplicate.

Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Rick Kawaguchi, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6434; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received. Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by

interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002-NM-149-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-149-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received a report indicating rapid in-flight decompression of a Boeing Model 737 series airplane. Investigation revealed that the skin above the forward entry door was separated at the stringer S-4R lap joint, with a 28-inch tear running along the lap joint. The skin was bent back at the upper edge of the stringer at S-5R and formed a rectangular opening that progressed from body station (BS) 328 to BS 300. Further investigation revealed that numerous scratches on the skin of the lap joint had initiated fatigue cracks and subsequent tearing of the skin. We also have received reports of similar damage (corrosion and fatigue cracking) to certain lap joints on other Model 737 series airplanes. These discrepancies have been attributed to the manufacturing process, which includes a cold-bonded adhesive of the lap joint configuration. Such conditions, if not corrected, could result in sudden fracture and failure of a lap joint and rapid in-flight decompression of the airplane fuselage.

The subject area on certain Model 747-100, -200B, and -200F series airplanes is manufactured using a process similar to that used on the affected Model 737 series airplanes. Therefore, those Model 747-100, -200B, and -200F series airplanes may be subject to the same unsafe condition revealed on the Model 737 series airplanes.

Related AD

This AD is related to AD 2000-17-04, amendment 39-11878, (65 FR 51750, August 25, 2000), applicable to certain Boeing Model 737-100, -200, and -200C series airplanes. That AD requires repetitive inspections to detect

discrepancies in the upper and lower skins of the fuselage lap joint, and repair if necessary.

Explanation of Relevant Service Information

The FAA has reviewed and approved Boeing Alert Service Bulletin 747-53A2463, including Appendices A, B, and C, and the Evaluation Form, dated March 7, 2002, which describes procedures for repetitive medium and low frequency eddy current inspections for discrepancies (cracking and/or corrosion), and repair of any discrepancies found. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously, except as discussed below.

Difference Between Alert Service Bulletin and This Proposed AD

Although the alert service bulletin specifies that the manufacturer may be contacted for disposition of certain repair conditions or inspection procedures, this proposed AD requires the repair and inspection procedures be accomplished in accordance with a method approved by the FAA, or in accordance with data meeting the type certification basis of the airplane approved by a Boeing Company Designated Engineering Representative who has been authorized by the FAA to make such findings.

Changes to 14 CFR Part 39/Effect on the Proposed AD

On July 10, 2002, the FAA issued a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the FAA's airworthiness directives system. The regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. Because we have now included this material in part 39, we no longer need to include it in each individual AD; however, this AD identifies the office authorized to approve alternative methods of compliance.

Cost Impact

There are approximately 86 airplanes of the affected design in the worldwide fleet. The FAA estimates that 55

airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 5,334 work hours per airplane to accomplish the proposed inspections, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$17,602,200, or \$320,040 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 2002–NM–149–AD.

Applicability: Model 747–100, –200B, and –200F series airplanes, as listed in Boeing Alert Service Bulletin 747–53A2463, dated March 7, 2002; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To find and fix discrepancies in the upper and lower skins of the fuselage lap joints, which could result in sudden fracture and failure of a lap joint and rapid in-flight decompression of the airplane fuselage, accomplish the following:

Initial and Repetitive Inspections

(a) Do the applicable (initial and repetitive) inspections as specified in Figures 2 through 8, as applicable, of the Accomplishment Instructions of Boeing Alert Service Bulletin 747–53A2463, including Appendices A, B, and C, dated March 7, 2002, to find discrepancies (cracking and corrosion) in the upper and lower skins of the fuselage lap joints. Do the inspections at the applicable times specified in Figure 1 of the Accomplishment Instructions of the alert service bulletin, in accordance with the alert service bulletin; except that where Figure 1 specifies a compliance time of "after the release date of this service bulletin," this AD requires a compliance time of "after the effective date of this AD." Where Figure 1 specifies a compliance time of "flight cycles" this AD requires a compliance time of "total flight cycles."

(b) Where Boeing Alert Service Bulletin 747–53A2463, including Appendices A, B, and C, dated March 7, 2002, specifies that the manufacturer may be contacted for certain inspection procedures, inspect per a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA; or per data meeting the type certification basis of the airplane approved by a Boeing Company Designated Engineering Representative (DER) who has been authorized by the Manager, Seattle ACO, to make such findings.

Adjustments to Compliance Time: Cabin Differential Pressure

(c) For the purposes of calculating the compliance threshold and repetitive interval for the inspections required by paragraph (a) of this AD: Flight cycles in which cabin differential pressure is at 2.0 pounds per square inch (psi) or less need not be counted when determining the number of flight cycles that have occurred on the airplane, provided that flight cycles with momentary spikes in cabin differential pressure above 2.0 psi are included as full pressure flight cycles. For this provision to apply, all cabin pressure records must be maintained for each

airplane. No fleet-averaging of cabin pressure is allowed.

Repair

(d) Before further flight, repair any discrepancy (cracking or corrosion) found during any inspection required by paragraph (a) of this AD, per the Accomplishment Instructions of Boeing Alert Service Bulletin 747–53A2463, including Appendices A, B, and C, dated March 7, 2002. If any discrepancy is found and the alert service bulletin specifies that the manufacturer may be contacted for disposition of certain repairs, before further flight, repair per a method approved by the Manager, Seattle ACO; or per data meeting the type certification basis of the airplane approved by a Boeing Company DER who has been authorized by the Manager, Seattle ACO, to make such findings.

Alternative Methods of Compliance

(e) In accordance with 14 CFR 39.19, the Manager, Seattle ACO, is authorized to approve alternative methods of compliance for this AD.

Issued in Renton, Washington, on June 26, 2003.

Michael J. Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03–16694 Filed 7–1–03; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 98–NM–11–AD]

RIN 2120–AA64

Airworthiness Directives; Boeing Model 737–100, –200, –200C, –300, –400, and –500 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Supplemental notice of proposed rulemaking; reopening of comment period.

SUMMARY: This document revises an earlier proposed airworthiness directive (AD), applicable to certain Boeing Model 737 series airplanes, that would have required inspections of certain bonded skin panels to detect delamination of the skin doublers (tear straps) from the skin panels, and follow-on corrective actions if necessary. This new action revises the proposed rule by revising certain inspection methods and expanding the area of certain inspections. This new action also proposes to extend the compliance time for certain inspections. The actions specified by this new proposed AD are intended to prevent skin doublers from