

Frequency	Field strength (volts per meter)	
	Peak	Average
8 GHz–12 GHz	3000	300
12 GHz–18 GHz	2000	200
18 GHz–40 GHz	600	200

The field strengths are expressed in terms of peak of the root-mean-square (rms) over the complete modulation period.

The threat levels identified above are the result of an FAA review of existing studies on the subject of HIRF, in light of the ongoing work of the Electromagnetic Effects Harmonization Working Group of the Aviation Rulemaking Advisory Committee.

Applicability

As discussed above, these special conditions are applicable to Israel Aircraft Industries Model 1124 airplanes modified by Avionics Certification Services. Should Avionics Certification Services apply at a later date for a supplemental type certificate to modify any other model included on Type Certificate No. A2SW to incorporate the same or similar novel or unusual design feature, these special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features on Israel Aircraft Industries Model 1124 airplanes modified by Avionics Certification Services. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the airplane.

The substance of these special conditions has been subjected to the notice and comment procedure in several prior instances and has been derived without substantive change from those previously issued. Because a delay would significantly affect the certification of the airplane, which is imminent, the FAA has determined that prior public notice and comment are unnecessary and impracticable, and good cause exists for adopting these special conditions upon issuance. The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior opportunities for comment described above.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

■ The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the supplemental type certification basis for the Israel Aircraft Industries Model 1124 airplanes modified by Avionics Certification Services.

1. *Protection from Unwanted Effects of High-Intensity Radiated Fields (HIRF).* Each electrical and electronic system that performs critical functions must be designed and installed to ensure that the operation and operational capability of these systems to perform critical functions are not adversely affected when the airplane is exposed to high-intensity radiated fields.

2. For the purpose of these special conditions, the following definition applies: *Critical Functions:* Functions whose failure would contribute to or cause a failure condition that would prevent the continued safe flight and landing of the airplane.

Issued in Renton, Washington, on August 7, 2003.

Neil D. Schalekamp,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-21106 Filed 8-15-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-CE-20-AD; Amendment 39-13270; AD 2003-16-17]

RIN 2120-AA64

Airworthiness Directives; Dornier Luftfahrt GMBH Models 228-100, 228-101, 228-200, 228-201, 228-202, and 228-212 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Dornier Luftfahrt

GMBH (Dornier) Models 228-100, 228-101, 228-200, 228-201, 228-202, and 228-212 airplanes that have electrical cabin/cockpit heater option P05 or option P09 installed. This AD requires you to modify the cockpit and cabin auxiliary heating wiring. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions specified by this AD are intended to correct problems with the current design of the heater wiring, which could result in failure of the auxiliary cabin heater. Such failure could lead to overheating and smoke in the cockpit.

DATES: This AD becomes effective on October 6, 2003.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of October 6, 2003.

ADDRESSES: You may get the service information referenced in this AD from Dornier Luftfahrt GmbH, Customer Support, P.O. Box 1103, D-82230 Wessling, Federal Republic of Germany; telephone: (08153) 300; facsimile: (08153) 304463. You may view this information at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003-CE-20-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Discussion

What events have caused this AD? The Luftfahrt-Bundesamt (LBA), which is the airworthiness authority for Germany, recently notified FAA that an unsafe condition may exist on certain Dornier Models 228-100, 228-101, 228-200 and 228-201, 228-202, and 228-212 airplanes. The LBA reports an occurrence of stuck contacts of the power relay of the heating circuit to the

auxiliary cabin heater, Dornier option P05 or P09.

What is the potential impact if FAA took no action? Failure of the auxiliary cabin heater could lead to overheating and smoke in the cockpit.

Has FAA taken any action to this point? We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Dornier Models 228–100, 228–101, 228–200, 228–201, 228–202, and 228–212 airplanes that have electrical cabin/cockpit heater option P05 or option P09 installed. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on May 15, 2003 (68 FR 26242). The NPRM proposed to require you to modify the cockpit and cabin auxiliary heating wiring.

Was the public invited to comment? The FAA encouraged interested persons to participate in the making of this

amendment. We did not receive any comments on the proposed rule or on our determination of the cost to the public.

FAA's Determination

What is FAA's final determination on this issue? After careful review of all available information related to the subject presented above, we have determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. We have determined that these minor corrections:

- Provide the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

How does the revision to 14 CFR part 39 affect this AD? On July 10, 2002,

FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs FAA's AD system. This regulation now includes material that relates to special flight permits, alternative methods of compliance, and altered products. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Cost Impact

How many airplanes does this AD impact? We estimate that this AD affects 14 airplanes in the U.S. registry.

What is the cost impact of this AD on owners/operators of the affected airplanes? We estimate the following costs to accomplish this modification. We have no way of determining the number of airplanes that may need such modification:

	Labor cost	Parts cost	Total cost per airplane
3 workhours × \$60 per hour = \$180		\$95	\$275

Regulatory Impact

Does this AD impact various entities? The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

Does this AD involve a significant rule or regulatory action? For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is

contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. FAA amends § 39.13 by adding a new AD to read as follows:

2003–16–17 Dornier Luftfahrt GMBH:

Amendment 39–13270; Docket No. 2003–CE–20–AD.

(a) *What airplanes are affected by this AD?* This AD affects Models 228–100, 228–101, 228–200, 228–201, 228–202, and 228–212 airplanes, all serial numbers, that are:

- (1) Certificated in any category; and
- (2) Equipped with electrical cabin/cockpit heater option P05 or option P09 auxiliary cabin heater(s) (32HA/35HA or 51HA/52HA).

(b) *Who must comply with this AD?*

Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?*

The actions specified by this AD are intended to correct problems with the current design of the heater wiring, which could result in failure of the auxiliary cabin heater. Such failure could lead to overheating and smoke in the cockpit.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Modify any installed cockpit and cabin auxiliary cabin heater (32HA/35HA or 51HA/52HA) heating wiring.	Within the next 50 hours time-in-service (TIS) after October 6, 2003 (the effective date of this AD), unless already accomplished. Removal from the airplane of any unmodified auxiliary cabin heater (32HA/35HA or 51HA/52HA) is terminating action for this AD.	In accordance with Fairchild Dornier Dornier 228 Service Bulletin No. SB–228–249, Revision No. 1, dated October 14, 2002, and following standard practices.

Actions	Compliance	Procedures
(2) Do not install any auxiliary cabin heater (32HA/35HA or 51HA/52HA) (or FAA-approved equivalent part number) unless it has been modified as required in paragraph (d)(1) of effective this AD.	As of October 6, 2003 (the effective date of this AD).	Not applicable.

(e) *Can I comply with this AD in any other way?* To use an alternative method of compliance or adjust the compliance time, follow the procedures in 14 CFR 39.19. Send these requests to the Manager, Standards Office, Small Airplane Directorate. For information on any already approved alternative methods of compliance, contact Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

(f) *Are any service bulletins incorporated into this AD by reference?* Actions required by this AD must be done in accordance with Fairchild Dornier Dornier 228 Service Bulletin No. SB-228-249, Revision No. 1, dated October 14, 2002. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You may get copies from Dornier Luftfahrt GmbH, Customer Support, P.O. Box 1103, D-82230 Wessling, Federal Republic of Germany; telephone: (08153) 300; facsimile: (08153) 304463. You may view copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note: The subject of this AD is addressed in German AD Number 2002-264, dated September 19, 2002.

(g) *When does this amendment become effective?* This amendment becomes effective on October 6, 2003.

Issued in Kansas City, Missouri, on August 7, 2003.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-20709 Filed 8-15-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NM-328-AD; Amendment 39-13266; AD 2003-16-13]

RIN 2120-AA64

Airworthiness Directives; Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes, that requires installing new vent tube assemblies for the main fuel tanks; and, on certain airplanes, inspecting to measure the clearance between the vent system tubing and the applicable wing ribs, and corrective action if necessary. This action is necessary to prevent a fire hazard due to fuel spillage. This action is intended to address the identified unsafe condition.

DATES: Effective September 22, 2003.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of September 22, 2003.

ADDRESSES: The service information referenced in this AD may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., Suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

James Delisio, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York 11581; telephone (516) 256-7521; fax (516) 568-2716.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes was published in the **Federal Register** on June 18, 2003 (68 FR 36513). That action proposed to require installing new vent tube assemblies for the main fuel tanks; and, on certain airplanes, inspecting to measure the clearance between the vent

system tubing and the applicable wing ribs, and corrective action if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Change to Labor Rate Estimate

We have reviewed the figures we have used over the past several years to calculate AD costs to operators. To account for various inflationary costs in the airline industry, we find it necessary to increase the labor rate used in these calculations from \$60 per work hour to \$65 per work hour. The cost impact information, below, reflects this increase in the specified hourly labor rate.

Cost Impact

The FAA estimates that the installation will be required to be accomplished on 45 Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes of U.S. registry. It will take approximately 15 work hours per airplane to accomplish the installation at an average labor rate of \$65 per work hour. Required parts will cost approximately \$10,273 per airplane. Based on these figures, the cost impact of the installation on U.S. operators is estimated to be \$506,160, or \$11,248 per airplane.

The FAA estimates that the inspection will be required to be accomplished on 43 Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes of U.S. registry. It will take approximately 1 work hour per airplane to accomplish the inspection at an average labor rate of \$65 per work hour. Based on these figures, the cost impact of the inspection on U.S. operators is estimated to be \$2,795, or \$65 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish