

**PART 39—AIRWORTHINESS
DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2014–22–09 The Boeing Company:

Amendment 39–18014 ; Docket No. FAA–2014–0430; Directorate Identifier 2014–NM–083–AD.

(a) Effective Date

This AD is effective December 17, 2014.

(b) Affected ADs

None.

(c) Applicability

(1) This AD applies to all The Boeing Company Model 767–200, –300, –300F, and –400ER series airplanes, certificated in any category.

(2) Installation of Supplemental Type Certificate (STC) ST01920SE ([http://rgl.faa.gov/Regulatory_and_Guidance_Library/rstc.nsf/0/59027f43b9a7486e86257b1d006591ee/\\$FILE/ST01920SE.pdf](http://rgl.faa.gov/Regulatory_and_Guidance_Library/rstc.nsf/0/59027f43b9a7486e86257b1d006591ee/$FILE/ST01920SE.pdf)) does not affect the ability to accomplish the actions required by this AD. Therefore, for airplanes on which STC ST01920SE is installed, a “change in product” alternative method of compliance (AMOC) approval request is not necessary to comply with the requirements of 14 CFR 39.17.

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight Controls.

(e) Unsafe Condition

This AD was prompted by a report of a trailing edge (TE) flap rotary actuator that had slipped relative to its mating reaction ring, which is attached to the flap support rib. We are issuing this AD to detect and correct flap rotary actuator gear disengagement from its mating reaction ring. This disengagement with flaps extended could cause an uncommanded roll due to flap blowback, overload, or flap departure from the airplane, which could compromise safe flight and landing of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Repetitive Inspections, Related Investigative Actions, and Corrective Actions

Except as provided by paragraph (h) of this AD, at the applicable time specified in paragraph 1.E., “Compliance,” of Boeing Alert Service Bulletin 767–27A0229, dated March 4, 2014: Do a detailed inspection for corrosion of the rotary actuator assembly fixed ring gear and reaction ring splines for each support position; and do all applicable

related investigative and corrective actions if necessary; in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 767–27A0229, dated March 4, 2014. Do all applicable related investigative and corrective actions before further flight. Repeat the inspection of the rotary actuator assembly fixed ring gear and reaction ring splines for each support position thereafter at the applicable intervals specified in paragraph 1.E., “Compliance,” of Boeing Alert Service Bulletin 767–27A0229, dated March 4, 2014.

(h) Exception to the Requirements of Paragraph (g) of this AD

Where paragraph 1.E., “Compliance,” of Boeing Alert Service Bulletin 767–27A0229, dated March 4, 2014, specifies a compliance time “after the original issue date of this service bulletin,” this AD requires compliance within the specified compliance time “after the effective date of this AD.”

(i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD if it is approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane and the approval must specifically refer to this AD.

(4) If the service information contains steps that are labeled as RC (Required for Compliance), those steps must be done to comply with this AD; any steps that are not labeled as RC are recommended. Those steps that are not labeled as RC may be deviated from, done as part of other actions, or done using accepted methods different from those identified in the specified service information without obtaining approval of an AMOC, provided the steps labeled as RC can be done and the airplane can be put back in a serviceable condition. Any substitutions or changes to steps labeled as RC require approval of an AMOC.

(j) Related Information

For more information about this AD, contact Allen Rauschendorfer, Aerospace Engineer, Airframe Branch, ANM–120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, WA 98057–

3356; phone: 425–917–6487; fax: 425–917–6590; email: Allen.Rauschendorfer@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Service Bulletin 767–27A0229, dated March 4, 2014.

(ii) Reserved.

(3) For Boeing service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P. O. Box 3707, MC 2H–65, Seattle, WA 98124–2207; telephone 206–544–5000, extension 1; fax 206–766–5680; Internet <https://www.myboeingfleet.com>.

(4) You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on October 28, 2014.

Jeffrey E. Duven

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2014–26440 Filed 11–10–14; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 43**

[Docket No. FAA–2000–8017, Amdt. No. 43–38A]

RIN 2120–AH11

Disposition of Life Limited Aircraft Parts; Technical Amendment

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; technical amendment.

SUMMARY: The FAA is correcting a final rule published in the **Federal Register** on January 15, 2002 (67 FR 2110). In that rule, the FAA amended its regulations to require that all persons who remove any life-limited aircraft part safely control that part, to deter the installation of that part after it has reached its life limit. The rule reduced the risk of life-limited parts being used

beyond their life limits. The rule also required that type certificate and design approval holders of life-limited parts provide instructions on how to mark a part indicating its current status, when requested by persons removing such a part. This document corrects error in the codified text of that document.

DATES: This amendment becomes effective: November 12, 2014.

FOR FURTHER INFORMATION CONTACT: For technical questions concerning this action, contact Marcus Cunningham, Flight Standards Service, AFS-300, Federal Aviation Administration, 800 Independence Avenue SW., Washington, DC 20591; telephone: (202) 267-1694, facsimile (202) 267-1736, or email: marcus.cunningham@faa.gov.

SUPPLEMENTARY INFORMATION:

Good Cause for Immediate Adoption Without Prior Notice

Section 553(b)(3)(B) of the Administrative Procedure Act (APA) (5 U.S.C. 551 et seq.) authorizes agencies to dispense with notice and comment procedures for rules when the agency for “good cause” finds that those procedures are “impracticable, unnecessary, or contrary to the public interest.” Under this section, an agency, upon finding good cause, may issue a final rule without seeking comment prior to the rulemaking.

Section 553(d)(3) of the Administrative Procedure Act requires that agencies publish a rule not less than 30 days before its effective date, except as otherwise provided by the agency for good cause found and published with the rule.

This document is correcting an error that is in 14 CFR 43.10(c)(6). This correction will not impose any additional restrictions on the persons affected by these regulations. Furthermore, any additional delay in making the regulations correct would be contrary to the public interest. Accordingly, the FAA finds that (i) public comment on these standards prior to promulgation is unnecessary, and (ii) good cause exists to make this rule effective in less than 30 days.

Background

On January 15, 2002, the FAA published a final rule entitled, “Safe Disposition of Life Limited Aircraft Parts” (67 FR 2110).

In that final rule, the FAA revised the regulations to require that all persons who remove any life-limited aircraft part to safely control that part to deter the installation of that part after it has reached its life limit. When published, a typographical error was created and

the word “product” was inadvertently replaced with the word “produce” in 14 CFR 43.10(c)(6).

Technical Amendment

The technical amendment will replace the word “produce” with the word “product” in § 43.10(c)(6).

Because the changes in this technical amendment result in no substantive change, we find good cause exists under 5 U.S.C. 553(d)(3) to make the amendment effective in less than 30 days.

List of Subjects in 14 CFR Part 43

Aircraft, Aviation safety, Life-limited parts, Reporting and recordkeeping requirements.

The Amendment

In consideration of the foregoing, the Federal Aviation Administration amends chapter I of title 14, Code of Federal Regulations as follows:

PART 43—MAINTENANCE, PREVENTIVE MAINTENANCE, REBUILDING, AND ALTERATION

- 1. The authority citation for part 43 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44703, 44705, 44707, 44711, 44713, 44717, 44725.

- 2. Revise § 43.10(c)(6) to read as follows:

§ 43.10 Disposition of life-limited aircraft parts.

* * * * *

(c) * * *

(6) *Mutilation.* The part may be mutilated to deter its installation in a type certificated product. The mutilation must render the part beyond repair and incapable of being reworked to appear to be airworthy.

* * * * *

Issued under authority of 49 U.S.C. 106(f), 44701(a), and 44707 in Washington, DC, on November 5, 2014.

Lirio Liu,

Director, Office of Rulemaking.

[FR Doc. 2014-26626 Filed 11-10-14; 8:45 am]

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DEPARTMENT OF COMMERCE

Bureau of Industry and Security

15 CFR Parts 734, 740, 748, 758, and 774

[Docket No. 130110030-4928-03]

RIN 0694-AF87

Clarifications and Corrections to the Export Administration Regulations (EAR): Control of Spacecraft Systems and Related Items the President Determines No Longer Warrant Control Under the United States Munitions List (USML)

AGENCY: Bureau of Industry and Security, Department of Commerce.

ACTION: Final rule.

SUMMARY: This final rule makes clarifications and corrections to an interim final rule that was published on May 13, 2014. The May 13 rule added controls to the Export Administration Regulations (EAR) for spacecraft and related items that the President has determined no longer warrant control under United States Munitions List (USML) Category XV—spacecraft and related items. New Export Control Classification Numbers (ECCNs) 9A515, 9B515, 9D515, and 9E515 created by the May 13 rule and existing ECCNs on the Commerce Control List (CCL) will control such items. The May 13 rule also revised various sections of the EAR to provide the proper level of control for the new ECCNs. The vast majority of the changes included in the May 13 rule have been implemented as published in the interim final rule, so those changes are not republished in this final rule. A full description of those changes can be found in the Background section and the regulatory text of the May 13 rule. The changes included in this final rule are limited to corrections and clarifications to what was included in the interim final rule. These corrections and clarifications were also informed by comments received in response to the May 13 rule that included a request for comments.

DATES: *Effective Date:* This rule is effective November 10, 2014.

FOR FURTHER INFORMATION CONTACT: For questions about the ECCNs included in this rule, contact Dennis Krepp, Office of National Security and Technology Transfer Controls, Bureau of Industry and Security, U.S. Department of Commerce, Telephone: 202-482-1309, email: Dennis.Krepp@bis.doc.gov. For general questions about the regulatory changes pertaining to satellites, spacecraft, and related items, contact