

The petitioner explains that the portals and gates are in place to ensure that personnel who gain access to the controlled access area have the proper clearance or are under escort and ensuring that prohibited articles are not allowed into the controlled area. The petitioner believes that the missing element of security is whether the fence line, which the petitioner believes does minimize the unauthorized removal of special nuclear material of 10 and 20 ton cylinders, adequately protects against the unauthorized removal of restricted information, equipment, and other materials or the unauthorized access to these types of materials.

The petitioner asserts that other facilities that possess Category III quantities of special nuclear material regulated by the NRC do not share the level of concern for classified matter, equipment, and technology that exists at the gaseous diffusion plants. The petitioner suggests that the regulations concerning security programs at the gaseous diffusion plants, such as escort requirements and physical security measures, should be amended to be made more stringent to protect this technology.

Sabotage Events

According to the petitioner, the NRC typically relies on local law enforcement agencies to respond to incidents of workplace violence or sabotage at material licensee facilities. The petitioner states that the scope and complexity of a gaseous diffusion plant makes it far different from other types of NRC licensed materials facilities. Furthermore, the petitioner believes that these differences result in unique problems in relying on local law enforcement agencies to protect such a facility from violent incidents. The petitioner indicates that local law enforcement agencies in the vicinity of the Paducah plant have stated, for the record, that they should not be viewed as a replacement for on-site security because of their lack of knowledge of the plant site, the types of hazards contained in the plant, and their limited resources. The petitioner presents two letters, attached to the petition, from law enforcement agencies in the vicinity of the Paducah plant that support this contention.

Because of the unique nature of gaseous diffusion plants and the importance of their operation, the petitioner believes that a violent incident or an act of sabotage would affect national security. The petitioner also asserts that, because of the many radiological and toxicological hazards associated with these plants, an act of

sabotage could adversely affect the safety of plant workers and the public.

The petitioner believes that these dangers were not addressed as part of the certification process. According to the petitioner, current NRC standards do not require a security force that is capable of preventing a sabotage event. The petitioner requests that the regulations be amended to require that security forces at the gaseous diffusion plants be able to detect, respond to, and mitigate violent incidents or acts of sabotage.

The petitioner also notes that current regulations do not require that the security force be armed or empowered to enforce the Atomic Energy Act. The petitioner requests that security officers at the gaseous diffusion plants be armed and empowered to make arrests in limited situations, such as for violations of the Atomic Energy Act.

Dated at Rockville, Maryland, this 4th day of May, 2000.

For the Nuclear Regulatory Commission.

Annette Vietti-Cook,

Secretary of the Commission.

[FR Doc. 00-11662 Filed 5-9-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-103-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737-200, -300, -400, and -500 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Boeing Model 737-200, -300, -400, and -500 series airplanes. This proposal would require replacement of existing door handle mounting hub assemblies with new, improved hub assemblies. This proposal is prompted by reports of cracked or broken mounting hub assemblies for the interior door handles on the cabin doors. The actions specified by the proposed AD are intended to prevent cracking or breaking of the door handle mounting hub, which could result in the interior door handle breaking off while the door is being opened. In an

emergency situation, this could impede evacuation of the airplane.

DATES: Comments must be received by June 26, 2000.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-103-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Keith Ladderud, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2780; fax (425) 227-1181.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2000-NM-103-AD."

The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-103-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received reports indicating that cracked or broken mounting hub assemblies for the interior door handles on the cabin doors have been found on certain Boeing Model 737-200, -300, -400, and -500 series airplanes. The primary use of the interior door handle is to be turned to latch the door after the door is shut using the assist handles. If the interior door handle is also used to close the door, the moment arm of the door handle puts too much force on the existing aluminum door handle mounting hub, which causes the mounting hub to crack or break. This condition, if not corrected, could result in the interior door handle breaking off while the door is being opened. In an emergency situation, this could impede evacuation of the airplane.

Explanation of Relevant Service Information

The FAA has reviewed and approved Boeing Service Bulletin 737-25-1322, Revision 2, dated February 19, 1998. That service bulletin describes procedures for replacement of existing door handle mounting hub assemblies in the forward and aft entry doors, forward galley door, and aft service door, with new, improved hub assemblies. The new mounting hub assemblies are made of stainless steel and are stronger than the existing aluminum mounting hub assemblies. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously, except as discussed below.

Differences Between the Proposed Rule and the Service Information

Operators should note that the proposed AD would require

replacement of existing door handle mounting hub assemblies with new, improved hub assemblies within 18 months after the effective date of this AD. The service bulletin recommends that the mounting hub in the forward entry door be replaced at the next "A" check, and the mounting hub assemblies in the aft entry door, forward galley door, and aft service door be replaced at the next "C" check. In developing an appropriate compliance time for this proposed AD, the FAA considered not only the manufacturer's recommendation, but the degree of urgency associated with addressing the subject unsafe condition, the average utilization of the affected fleet, and the time necessary to replace the mounting hub assemblies (approximately 3 work hours per door). In light of all of these factors, the FAA finds an 18-month compliance time for initiating the proposed actions to be warranted, in that it represents an appropriate interval of time allowable for affected airplanes to continue to operate without compromising safety.

Cost Impact

There are approximately 1,575 airplanes of the affected design in the worldwide fleet. The FAA estimates that 632 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 12 work hours per airplane (3 work hours per door) to accomplish the proposed replacement, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$2,150 per airplane. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$1,813,840, or \$2,870 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT

Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 2000-NM-103-AD.

Applicability: Model 737-200, -300, -400, and -500 series airplanes; as listed in Boeing Service Bulletin 737-25-1322, Revision 2, dated February 19, 1998; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent cracking or breaking of the door handle mounting hub, which could result in the interior door handle breaking off while the door is being opened, and, in an emergency situation, could impede evacuation of the airplane, accomplish the following:

Replacement

(a) Within 18 months after the effective date of this AD, replace existing door handle mounting hub assemblies in the forward and aft entry doors, forward galley door, and aft service door, with new, improved hub assemblies, in accordance with Boeing Service Bulletin 737-25-1322, Revision 2, dated February 19, 1998.

Note 2: Replacements accomplished prior to the effective date of this AD in accordance with Boeing Service Bulletin 737-25-1322, dated January 19, 1995, or Revision 1, dated December 19, 1996, are considered acceptable for compliance with paragraph (a) of this AD.

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on May 4, 2000.

Vi L. Lipski,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 00-11725 Filed 5-9-00; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2000-NM-50-AD]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model DC-10-10, -15, -30, -30F (KC-10A Military), and -40 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model DC-

10-10, -15, -30, -30F (KC-10A military), and -40 series airplanes. This proposal would require performing repetitive ultrasonic inspections of the attaching bolts on the inboard and outboard support on the inboard and outboard flap assembly to detect failed bolts, or verifying the torque of the attaching bolts on the inboard support on the outboard flap; and follow-on actions. This proposal also would require replacing all bolts with bolts made from Inconel, which would constitute terminating action for the repetitive inspection requirements. This proposal is prompted by a report of an in-flight loss of the inboard flap assembly on an airplane during approach for landing. The actions specified by the proposed AD are intended to prevent in-flight loss of inboard and outboard flap assemblies due to failure of H-11 attaching bolts, which could result in reduced controllability of the airplane.

DATES: Comments must be received by June 26, 2000.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-50-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1-L51 (2-60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: Ron Atmur, Aerospace Engineer, Airframe Branch, ANM-120L, FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5224; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as

they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2000-NM-50-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-50-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received a report of an in-flight loss of the left inboard flap assembly on a McDonnell Douglas Model DC-10 series airplane during approach for landing. Investigation revealed that bolts made from H-11 steel, which attach the outboard hinge to the lower surface of the flap, had failed. Analysis of the bolts determined the cause of failure to be stress corrosion. The FAA has received no damage or failure reports about the outboard flaps. However, the inboard and outboard hinges are attached to the lower surface of the flap using similar type design and the same material as the installation of the inboard flap outboard hinge. Failure of H-11 attaching bolts could result in an in-flight loss of inboard and outboard flap assemblies, and consequent reduced controllability of the airplane.

Explanation of Relevant Service Information

The FAA has reviewed and approved McDonnell Douglas Alert Service Bulletin DC10-57A143, dated December