TABLE 1—COMPLIANCE TIMES FOR DETAILED VISUAL INSPECTION—Continued	
Model	Compliance time (whichever occurs later)
(iii) 'LR' A310-304, A310-322, A310-324, and A310-325 long range airplanes.	<ul> <li>(A) Prior to accumulation of 12,800 flight cycles or 64,300 flight hours since first flight of the airplane, whichever occurs first; or</li> <li>(B) Within 300 flight cycles or 1,450 flight hours, whichever occurs first, after the effective date of this AD.</li> </ul>

(2) As of the effective date of this AD, if any repair has already been done as a result of finding skin cracks at rib 2 in the area to be inspected, the inspection requirements of this AD are not required for the repaired area. Instead, for previously repaired areas,

continue the inspection in accordance with the procedures specified in paragraph (g) of this AD. The rest of the rib 2 area not covered by the repair must be inspected in accordance with the requirements of this AD.

(3) If no crack is found, repeat the inspection required by paragraph (f)(1) of this AD thereafter at the intervals not to exceed those specified in Table 2 of this AD, as applicable.

### TABLE 2—COMPLIANCE TIMES FOR REPETITIVE INSPECTION INTERVAL

Model	Repetitive inspection interval
A310–203, A310–204, A310–221, and A310–222 airplanes	Within 1,700 flight cycles or 3,500 flight hours, whichever occurs first. Within 1,600 flight cycles or 4,600 flight hours, whichever occurs first.
'LR' A310-304, A310-322, A310-324, and A310-325 long range airplanes.	Within 1,200 flight cycles or 6,100 flight hours, whichever occurs first.

- (4) If any crack is found during any inspection required by paragraph (f)(1) or (f)(3) of this AD, before further flight, repair in accordance with the Accomplishment Instructions of Airbus Mandatory Service Bulletin A310-57-2096, dated May 6, 2008. For previously repaired areas, continue the inspection in accordance with the procedures specified in paragraph (g) of this AD.
- (5) After each inspection required by this AD, submit an inspection report in accordance with Airbus Mandatory Service Bulletin A310-57-2096, dated May 6, 2008; at the times specified in paragraphs (f)(5)(i) or (f)(5)(ii) of this AD, as applicable.
- (i) If the inspection is done on or after the effective date of this AD: Submit the report within 30 days after the inspection.
- (ii) If the inspection was accomplished prior to the effective date of this ÂD: Submit the report within 30 days after the effective date of this AD.

### FAA AD Differences

Note 2: This AD differs from the MCAI and/or service information as follows: No differences.

### Other FAA AD Provisions

- (g) The following provisions also apply to this AD:
- (1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or

- principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office.
- (2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.
- (3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

# **Related Information**

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2008-0211, dated December 4, 2008; and Airbus Mandatory Service Bulletin A310-57-2096, dated May 6, 2008; for related information.

# Material Incorporated by Reference

- (i) You must use Airbus Mandatory Service Bulletin A310-57-2096, including Appendix 1, dated May 6, 2008, to do the actions required by this AD, unless the AD specifies otherwise.
- (1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) For service information identified in this AD, contact Airbus, Airbus SAS-EAW (Airworthiness Office), 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail: account.airwortheas@airbus.com; Internet http:// www.airbus.com.

- (3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.
- (4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal\_register/ code\_of\_federal\_regulations/ibr\_ locations.html.

Issued in Renton, Washington, on January 28, 2010.

### Ali Bahrami.

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2010-2607 Filed 2-11-10; 8:45 am]

BILLING CODE 4910-13-P

# **DEPARTMENT OF TRANSPORTATION**

# **Federal Aviation Administration**

## 14 CFR Part 39

[Docket No. FAA-2009-0994; Directorate Identifier 2009-NM-108-AD; Amendment 39-16194; AD 2010-04-01]

### RIN 2120-AA64

Airworthiness Directives; Dassault-**Aviation Model Falcon 900EX Airplanes** 

**AGENCY:** Federal Aviation

Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

A quality control performed during completion of one Falcon 900EX aeroplane has shown that the crew and passenger Right-Hand (RH) oxygen lines may both interfere with the frame 8 of the aeroplane structure. A subsequent design review of the oxygen lines routing has confirmed that, on certain aeroplanes, equipped in RH mid-cabin with a 115 cu-ft oxygen cylinder, the installation of the line support assembly at frame 8 needs to be accomplished with precaution; otherwise, the oxygen lines might interfere with the structure, and this condition could lead to an oxygen leak.

\* \* \* \* \*

The unsafe condition is an oxygen leak, which would result in insufficient oxygen flow to passenger oxygen masks during a depressurization event. We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective March 19, 2010.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of March 19, 2010.

ADDRESSES: You may examine the AD docket on the Internet at http://www.regulations.gov or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149.

### SUPPLEMENTARY INFORMATION:

# Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on October 28, 2009 (74 FR 55488). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

A quality control performed during completion of one Falcon 900EX aeroplane has shown that the crew and passenger RightHand (RH) oxygen lines may both interfere with the frame 8 of the aeroplane structure. A subsequent design review of the oxygen lines routing has confirmed that, on certain aeroplanes, equipped in RH mid-cabin with a 115 cu-ft oxygen cylinder, the installation of the line support assembly at frame 8 needs to be accomplished with precaution; otherwise, the oxygen lines might interfere with the structure, and this condition could lead to an oxygen leak.

As a result, [EASA] Airworthiness Directive 2009–0104 was issued to require inspection of the oxygen lines [for signs of interference or chafing damage], replacement of any damaged lines and modification of their support assembly. Since then, it has been found that the applicability of the AD had not been correctly defined.

This [new EASA] AD retains the requirements of AD 2009–0104 which is superseded and corrects the applicability.

The unsafe condition is an oxygen leak, which would result in insufficient oxygen flow to passenger oxygen masks during a depressurization event. Modifying the support assembly of the oxygen lines includes drilling holes to install improved support bracket assemblies at frame 8, stringers 11 and 13, and installing the improved assemblies. You may obtain further information by examining the MCAI in the AD docket.

#### Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

# **Explanation of Change Made to This AD**

We have revised this AD to identify the legal name of the manufacturer as published in the most recent type certificate data sheet for the affected airplane models.

# **Explanation of Change to Costs of Compliance**

Since issuance of the NPRM, we have increased the labor rate used in the Costs of Compliance from \$80 per workhour to \$85 per work-hour. The Costs of Compliance information, below, reflects this increase in the specified hourly labor rate.

## Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD with the changes described previously. We determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

# Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a Note within the AD.

### **Costs of Compliance**

We estimate that this AD will affect about 23 products of U.S. registry. We also estimate that it will take about 4 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts will cost about \$0 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$7,820, or \$340 per product.

### **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- 1. Is not a "significant regulatory action" under Executive Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- 3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

### **Examining the AD Docket**

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

# 2010–04–01 Dassault-Aviation:

Amendment 39–16194. Docket No. FAA–2009–0994; Directorate Identifier 2009–NM–108–AD.

### **Effective Date**

(a) This airworthiness directive (AD) becomes effective March 19, 2010.

### Affected ADs

(b) None.

### Applicability

(c) This AD applies to Dassault-Aviation Model Falcon 900EX airplanes, certificated in any category, with serial numbers 120 through 123 inclusive, 125 through 127 inclusive, 129, 132, 134 through 145 inclusive, 147, 151, 153, 155, 157 through 159 inclusive, 163, 165, 168 through 170 inclusive, 172, 174, 178, 182, 183, 194, 196, 197, 199, and 206.

### Subject

(d) Air Transport Association (ATA) of America Code 35: Oxygen.

#### Reasor

(e) The mandatory continuing airworthiness information (MCAI) states:

A quality control performed during completion of one Falcon 900EX aeroplane has shown that the crew and passenger Right-Hand (RH) oxygen lines may both interfere with the frame 8 of the aeroplane structure. A subsequent design review of the oxygen lines routing has confirmed that, on certain aeroplanes, equipped in RH mid-cabin with a 115 cu-ft oxygen cylinder, the installation of the line support assembly at frame 8 needs to be accomplished with precaution; otherwise, the oxygen lines might interfere with the structure, and this condition could lead to an oxygen leak.

As a result, [European Aviation Safety Agency (EASA)] Airworthiness Directive 2009–0104 was issued to require inspection of the oxygen lines [for signs of interference and chafing damage], replacement of any damaged lines and modification of their support assembly. Since then, it has been found that the applicability of the AD had not been correctly defined.

This [EASA] AD retains the requirements of AD 2009–0104 which is superseded and corrects the applicability.

The unsafe condition is an oxygen leak, which would result in insufficient oxygen flow to passenger oxygen masks during a depressurization event. Modifying the support assembly of the oxygen lines includes drilling holes to install improved support bracket assemblies at frame 8, stringers 11 and 13, and installing the improved assemblies.

### **Actions and Compliance**

- (f) Unless already done, do the following actions.
- (1) Within 2 months after the effective date of this AD, inspect the oxygen lines in accordance with Part 1 of the Accomplishment Instructions of Dassault Mandatory Service Bulletin F900EX–347, Revision 1, dated May 18, 2009. If any interference or damage is found, before further flight, replace the oxygen lines and install improved brackets, in accordance with Part 2 of the Accomplishment Instructions of Dassault Mandatory Service Bulletin F900EX–347, Revision 1, dated May 18, 2009.
- (2) If no interference and no damage are found during the inspection required by paragraph (f)(1) of this AD: Within 72 months after the effective date of this AD, replace the oxygen line support assemblies, in accordance with Part 2 of the

Accomplishment Instructions of Dassault Mandatory Service Bulletin F900EX–347, Revision 1, dated May 18, 2009.

(3) Actions accomplished before the effective date of this AD in accordance with Dassault Mandatory Service Bulletin F900EX–347, dated March 19, 2009, are acceptable for compliance with corresponding actions specified in this AD.

### **FAA AD Differences**

**Note 1:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

- (g) The following provisions also apply to this AD:  $\,$
- (1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tom Rodriguez, Aerospace Engineer, International Branch. ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office. The AMOC approval letter must specifically reference this AD.
- (2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.
- (3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

### **Related Information**

(h) Refer to MCAI EASA Airworthiness Directive 2009–0126, dated June 18, 2009; and Dassault Mandatory Service Bulletin F900EX–347, Revision 1, dated May 18, 2009; for related information.

### Material Incorporated by Reference

- (i) You must use Dassault Mandatory Service Bulletin F900EX–347, Revision 1, dated May 18, 2009, to do the actions required by this AD, unless the AD specifies otherwise.
- (1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) For service information identified in this AD, contact Dassault Falcon Jet, P.O. Box 2000, South Hackensack, New Jersey 07606;

telephone 201–440–6700; Internet http://www.dassaultfalcon.com.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221 or 425–227–1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http://www.archives.gov/federal\_register/code\_of\_federal\_regulations/ibr locations.html.

Issued in Renton, Washington, on January 29, 2010.

### Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2010-2609 Filed 2-11-10; 8:45 am]

BILLING CODE 4910-13-P

# PENSION BENEFIT GUARANTY CORPORATION

### 29 CFR Part 4022

Benefits Payable in Terminated Single-Employer Plans; Interest Assumptions for Valuing and Paying Benefits

**AGENCY: Pension Benefit Guaranty** 

Corporation. **ACTION:** Final rule.

SUMMARY: Pension Benefit Guaranty Corporation's regulation on Benefits Payable in Terminated Single-Employer Plans prescribes interest assumptions for valuing and paying certain benefits under terminating single-employer plans. This final rule amends the benefit payments regulation to adopt interest assumptions for plans with valuation dates in March 2010. Interest assumptions are also published on PBGC's Web site (http://www.pbgc.gov). DATES: Effective March 1, 2010.

### FOR FURTHER INFORMATION CONTACT:

Catherine B. Klion, Manager, Regulatory and Policy Division, Legislative and Regulatory Department, Pension Benefit Guaranty Corporation, 1200 K Street, NW., Washington, DC 20005, 202–326–4024. (TTY/TDD users may call the Federal relay service toll-free at 1–800–

877–8339 and ask to be connected to 202–326–4024.)

supplementary information: PBGC's regulations prescribe actuarial assumptions—including interest assumptions—for valuing and paying plan benefits of terminating single-employer plans covered by title IV of the Employee Retirement Income Security Act of 1974. The interest assumptions are intended to reflect current conditions in the financial and annuity markets.

These interest assumptions are found in two PBGC regulations: the regulation on Benefits Payable in Terminated Single-Employer Plans (29 CFR part 4022) and the regulation on Allocation of Assets in Single-Employer Plans (29 CFR part 4044). Assumptions under the asset allocation regulation are updated quarterly; assumptions under the benefit payments regulation are updated monthly. This final rule updates only the assumptions under the benefit payments regulation.

Two sets of interest assumptions are prescribed under the benefit payments regulation: (1) a set for PBGC to use to determine whether a benefit is payable as a lump sum and to determine lump-sum amounts to be paid by PBGC (found in Appendix B to Part 4022), and (2) a set for private-sector pension practitioners to refer to if they wish to use lump-sum interest rates determined using PBGC's historical methodology (found in Appendix C to Part 4022).

This amendment (1) adds to Appendix B to Part 4022 the interest assumptions for PBGC to use for its own lump-sum payments in plans with valuation dates during March 2010, and (2) adds to Appendix C to Part 4022 the interest assumptions for private-sector pension practitioners to refer to if they wish to use lump-sum interest rates determined using PBGC's historical methodology for valuation dates during March 2010.

The interest assumptions that PBGC will use for its own lump-sum payments (set forth in Appendix B to part 4022) will be 2.75 percent for the period during which a benefit is in pay status and 4.00 percent during any years preceding the benefit's placement in pay status. In comparison with the interest

assumptions in effect for February 2010, these interest assumptions are unchanged. For private-sector payments, the interest assumptions (set forth in Appendix C to part 4022) will be the same as those used by PBGC for determining and paying lump sums (set forth in Appendix B to part 4022).

PBGC has determined that notice and public comment on this amendment are impracticable and contrary to the public interest. This finding is based on the need to determine and issue new interest assumptions promptly so that the assumptions can reflect current market conditions as accurately as possible.

Because of the need to provide immediate guidance for the valuation and payment of benefits in plans with valuation dates during March 2010, PBGC finds that good cause exists for making the assumptions set forth in this amendment effective less than 30 days after publication.

PBGC has determined that this action is not a "significant regulatory action" under the criteria set forth in Executive Order 12866.

Because no general notice of proposed rulemaking is required for this amendment, the Regulatory Flexibility Act of 1980 does not apply. See 5 U.S.C. 601(2).

# List of Subjects in 29 CFR Part 4022

Employee benefit plans, Pension insurance, Pensions, Reporting and recordkeeping requirements.

■ In consideration of the foregoing, 29 CFR part 4022 is amended as follows:

### PART 4022—BENEFITS PAYABLE IN TERMINATED SINGLE-EMPLOYER PLANS

■ 1. The authority citation for part 4022 continues to read as follows:

**Authority:** 29 U.S.C. 1302, 1322, 1322b, 1341(c)(3)(D), and 1344.

■ 2. In appendix B to part 4022, Rate Set 197, as set forth below, is added to the table.

# Appendix B to Part 4022—Lump Sum Interest Rates for PBGC Payments

For plans with a valuation Deferred annuities (percent) Immediate date Rate set annuity rate  $i_1$ (percent)  $i_2$ ĺз  $n_2$ On or after Before 197 3-1-10 2.75 4.00 4.00 4.00 7 8 4-1-10