

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

Airbus Helicopters: Docket No. FAA–2021–0016; Project Identifier 2019–SW–114–AD.

(a) Comments Due Date

The FAA must receive comments by April 29, 2021.

(b) Affected Airworthiness Directives (ADs)

None.

(c) Applicability

This AD applies to all Airbus Helicopters Model EC225LP helicopters, certificated in any category.

(d) Subject

Joint Aircraft System Component (JASC) Code 6397, Main Rotor Drive System Wiring.

(e) Reason

This AD was prompted by reports of oil leaks during engine starting, originating from the main gearbox (MGB). The FAA is issuing this AD to address the inadvertent opening of the P 2.4 valve of the MGB emergency lubrication (EMLUB) system, which results from MGB pressurization by compressed air produced by the engine during starting in response to a signal from the EMLUB electronic control card. This condition could result in loss of the MGB lubrication system and a reduced ability of the crew to manage adverse operating conditions.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with European Union Aviation Safety Agency AD No. 2016–0232R1, dated December 12, 2019 (EASA AD 2016–0232R1).

(h) Exceptions to EASA AD 2016–0232R1

(1) Where EASA AD 2016–0232R1 refers December 6, 2016 (the effective date of European Aviation Safety Agency AD No. 2016–0232, dated November 22, 2016), this AD requires using the effective date of this AD.

(2) Where EASA AD 2016–0232R1 refers to flight hours (FH), this AD requires using hours time-in-service (TIS).

(3) Where paragraph (2) of EASA AD 2016–0232R1 allows an additional interval margin of 225 FH, this AD does not. This AD requires accomplishing the functional tests within 600 hours TIS, and thereafter at intervals not to exceed 600 hours TIS.

(4) Where the service information referenced in EASA AD 2016–0232R1 requires contacting Airbus Helicopters technical support, this AD requires that the corrective action be accomplished using a method approved by the Manager, Strategic Policy Rotorcraft Section, FAA. The Manager's approval letter must specifically refer to this AD.

(5) The “Remarks” section of EASA AD 2016–0232R1 does not apply to this AD.

(i) Special Flight Permit

Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the helicopter can be modified (if the operator elects to do so), provided the helicopter is operated under visual flight rules and without passengers only.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Strategic Policy Rotorcraft Section, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the Strategic Policy Rotorcraft Section, send it to: Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 470 L'Enfant Plaza SW, Washington, DC 20024; telephone 202–267–9167; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Related Information

(1) For EASA AD 2016–0232R1, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110. This material may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0016.

(2) For more information about this AD, contact: Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 470 L'Enfant Plaza SW, Washington, DC 20024; telephone 202–267–9167; email hal.jensen@faa.gov.

Issued on January 27, 2021.

Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–05196 Filed 3–12–21; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2021–0138; Project Identifier MCAI–2020–01466–T]

RIN 2120–AA64

Airworthiness Directives; BAE Systems (Operations) Limited Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for all BAE Systems (Operations) Limited Model BAe 146 and Model Avro 146–RJ series airplanes. This proposed AD was prompted by a report indicating that during a routine battery capacity check on the emergency light power units, the printed circuit boards (PCBs) for certain power units were found to show signs of burning. This proposed AD would require replacing each Honeywell emergency light power unit having a certain part number with a serviceable emergency light power unit. The FAA is proposing this AD to address the unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by April 29, 2021.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0138; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, any comments received, and other information. The street address for Docket Operations is listed above.

FOR FURTHER INFORMATION CONTACT:

Todd Thompson, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax: 206-231-3228; email Todd.Thompson@faa.gov.

SUPPLEMENTARY INFORMATION:**Comments Invited**

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2021-0138; Project Identifier MCAI-2020-01466-T" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend the proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to <https://www.regulations.gov>, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this NPRM.

Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your

comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Todd Thompson, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax: 206-231-3228; email Todd.Thompson@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

Background

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2020-0237, dated October 28, 2020 (EASA AD 2020-0237) (also referred to after this as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for all BAE Systems (Operations) Limited Model BAe 146 and Model Avro 146-RJ series airplanes. You may examine the MCAI in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0138.

This proposed AD was prompted by a report indicating that during a routine battery capacity check on the emergency

light power units, the PCBs for power units LE 10 and LE 22 (Illustrated Parts Catalog (IPC) 33-50-00) were found to show signs of burning. The FAA is proposing this AD to address heat damage of the PCBs, which could lead to battery discharge and possibly result in lack of power supply to the emergency light units when needed. See the MCAI for additional background information.

FAA's Determination

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI and service information referenced above. The FAA is proposing this AD because the FAA evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop on other products of the same type design.

Proposed AD Requirements in This NPRM

This proposed AD would require replacing each Honeywell emergency light power unit having a certain part number with a serviceable emergency light power unit.

Costs of Compliance

The FAA estimates that this proposed AD affects 30 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
2 work-hours × \$85 per hour = \$170	\$1,800	\$1,970	\$59,100

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing

regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

The FAA determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the

national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Would not affect intrastate aviation in Alaska, and

(3) Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

BAE Systems (Operations) Limited: Docket No. FAA–2021–0138; Project Identifier MCAI–2020–01466–T.

(a) Comments Due Date

The FAA must receive comments on this airworthiness directive (AD) action by April 29, 2021.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all BAE Systems (Operations) Limited airplanes specified in paragraphs (c)(1) and (2) of this AD, certificated in any category.

(1) Model BAe 146–100A, –200A, and –300A airplanes.

(2) Model Avro 146–RJ70A, 146–RJ85A, and 146–RJ100A airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 33, Lights.

(e) Unsafe Condition

This AD was prompted by a report indicating that during a routine battery capacity check on the emergency light power units, the printed circuit boards (PCBs) for power units LE 10 and LE 22 (Illustrated Parts Catalog (IPC) 33–50–00) were found to show signs of burning. The FAA is issuing this AD to address heat damage of the PCBs, which could lead to battery discharge and possibly result in lack of power supply to the emergency light units when needed.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Definitions

(1) An affected part is defined as a Honeywell emergency light power unit, having part number 60–3550–1, except for those modified and marked using the instructions specified in Honeywell Service Bulletin 60–3550–33–0001, Revision 1, dated September 3, 2013.

(2) A serviceable part is defined as an emergency light power unit that is not an affected part.

(3) Group 1 airplanes are those that have an affected part installed.

(4) Group 2 airplanes are those that do not have an affected part installed.

(h) Replacement

Within two months after the effective date of this AD: Replace each affected part with a serviceable part.

Note 1 to paragraph (h): BAE Systems (Operations) Limited Service Bulletin ISB.33–081, dated November 4, 2019, contains information related to the replacement specified in paragraph (h) of this AD.

(i) Parts Installation Prohibition

At the applicable compliance times specified in paragraphs (i)(1) or (2) of this AD, do not install an affected part on any airplane.

(1) For Group 1 airplanes: After replacement of each affected part on an airplane as specified in paragraph (h) of this AD.

(2) For Group 2 airplanes: As of the effective date of this AD.

(j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Large Aircraft Section, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the Large Aircraft Section, International Validation Branch, send it to the attention of the person identified in paragraph (k)(2) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, Large Aircraft Section, International Validation Branch, FAA; or the European Union Aviation Safety Agency (EASA); or BAE Systems (Operations) Limited's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(k) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2020–0237, dated October 28, 2020, for related information. This MCAI may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0138.

(2) For more information about this AD, contact Todd Thompson, Aerospace

Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax: 206–231–3228; email Todd.Thompson@faa.gov.

Issued on March 3, 2021.

Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–05204 Filed 3–12–21; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2006–26107; Project Identifier 2004–SW–30–AD]

RIN 2120–AA64

Airworthiness Directives; Carson Helicopters, Inc.; Croman Corporation; Sikorsky Aircraft Corporation; and Siller Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Supplemental notice of proposed rulemaking (SNPRM); reopening of comment period.

SUMMARY: The FAA is revising an earlier proposal for all Sikorsky Aircraft Corporation Model S–61 A, D, E, and V helicopters; Croman Corporation Model SH–3H helicopters, Carson Helicopters, Inc. Model S–61L helicopters; and Siller Helicopters Model CH–3E and SH–3A helicopters. This action revises the notice of proposed rulemaking (NPRM) by adding camshaft and gear housing part numbers that need to be marked and clarifying the applicability and certain compliance times. The FAA is proposing this airworthiness directive (AD) to address the unsafe condition on these products. Since these actions would impose an additional burden over that in the NPRM, the FAA is reopening the comment period to allow the public the chance to comment on these changes.

DATES: The comment period for the NPRM published in the **Federal Register** on October 30, 2006 (71 FR 63272), is reopened.

The FAA must receive comments on this SNPRM by April 29, 2021.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.