Actions	Compliance	Procedures	
 (1) Replace any non-strengthened rudder pedals with reinforced rudder pedals: (i) The left-hand reinforced rudder pedal is part number (P/N) \$Y185A; and (ii) The right-hand reinforced rudder pedal is P/N \$Y196A. (2) Do not install any non-strengthened rudder pedal as specified in paragraphs (e)(1)(i) and (e)(1)(ii) of this AD. 	Within the next 25 hours time-in-service (TIS) after the effective date of this AD, unless already done. As of the effective date of this AD	Follow Société Nouvelle Centrair Service Bulletin No. 101–24, dated March 5, 2003 (this is the date of the French AD 2003–095(a) that transmitted the service bulletin). The applicable glider maintenance manual also addresses this issue. Not Applicable.	

May I Request an Alternative Method of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.19. Unless FAA authorizes otherwise, send your request to your principal inspector. The principal inspector may add comments and will send your request to the Manager, Standards Office, Small Airplane Directorate, FAA. For information on any already approved alternative methods of compliance, contact Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; facsimile: (816) 329-4090.

Is There Other Information That Relates to This Subject?

(g) French AD 2003-095(A), dated March 5, 2003, also addresses the subject of this AD.

May I Get Copies of the Documents Referenced in This AD?

(h) To get copies of the documents referenced in this AD, contact CENTRAIR, Aerodome B.P.N. 44, 36300 Le Blanc, France; telephone: 02.54.37.07.96; facsimile: 02.54.37.48.64. To view the AD docket, go to the Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC, or on the Internet at http:// /dms.dot.gov. This is docket number FAA-2004-19522.

Issued in Kansas City, Missouri, on December 6, 2004.

Scott L. Sedgwick,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-27197 Filed 12-10-04; 8:45 am] BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2004-19616; Directorate Identifier 2004-CE-38-AD1

RIN 2120-AA64

Airworthiness Directives; CENTRAIR 101 Series Gliders

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for all CENTRAIR 101 series gliders with other than elevator or aileron part number (P/ N) SY991A hinge pins installed. This proposed AD would require you to replace any installed elevator or aileron hinge pins that are not P/N SY991A hinge pins with P/N SY991A pins. This proposed AD results from mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for France. We are issuing this proposed AD to replace incorrectly heattreated elevator or aileron hinge pins, which could result in failure of the elevator or ailerons. Such failure during takeoff, landing, or flight operations could lead to loss of glider control. **DATES:** We must receive any comments

on this proposed AD by January 14, 2005.

ADDRESSES: Use one of the following to submit comments on this proposed AD:

- DOT Docket Web site: Go to http: //dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590-
 - Fax: 1-202-493-2251.
- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

To get the service information identified in this proposed AD, contact CENTRAIR, Aerodome B.P.N. 44, 36300 Le Blanc, France; telephone: 02.54.37.07.96; facsimile: 02.54.37.48.64.

To view the comments to this proposed AD, go to http://dms.dot.gov. This is docket number FAA-2004-19616.

FOR FURTHER INFORMATION CONTACT: Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; facsimile: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Comments Invited

How do I comment on this proposed AD? We invite you to submit any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under ADDRESSES. Include the docket number, "FAA-2004-19616; Directorate Identifier 2004-CE-38-AD" at the beginning of your comments. We will post all comments we receive, without change, to http://dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed rulemaking. Using the search function of our docket Web site, anyone can find and read the comments received into any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). This is docket number FAA-2004-19616. You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78) or you may visit http:// dms.dot.gov.

Are there any specific portions of this proposed AD I should pay attention to? We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. If you contact us through a nonwritten communication and that contact relates to a substantive part of this proposed AD, we will summarize the contact and place the summary in the docket. We will consider all comments received by the closing date and may amend this proposed AD in light of those comments and contacts.

Docket Information

Where can I go to view the docket information? You may view the AD docket that contains the proposal, any comments received, and any final disposition in person at the DMS Docket Offices between 9 a.m. and 5 p.m. (eastern standard time), Monday through Friday, except Federal holidays. The Docket Office (telephone 1-800-647-5227) is located on the plaza level of the Department of Transportation NASSIF Building at the street address stated in ADDRESSES. You may also view the AD docket on the Internet at http://dms.dot.gov. The comments will be available in the AD docket shortly after the DMS receives them.

Discussion

What events have caused this proposed AD? The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, recently notified FAA that an unsafe condition may exist on all CENTRAIR 101 series gliders. The DGAC reports occurrences of improperly heat-treated aileron and elevator hinge pins installed on the CENTRAIR 101 series gliders. Incorrectly heat-treated elevator or aileron hinge pins could result in longitudinal cracks that cause failure of the elevator or ailerons. CENTRAIR has made available new hinge pins (part number (P/N) SY991A) to replace any incorrectly heat-treated elevator or aileron hinge pins or hinge pins with longitudinal cracks.

What is the potential impact if FAA took no action? Failure of the elevator or ailerons during takeoff, landing, or flight operations could lead to loss of

glider control.

Is there service information that applies to this subject? CENTRAIR has issued Société Nouvelle Centrair Service Bulletin No. 101–22, dated March 13, 2001.

What are the provisions of this service information? The service bulletin includes procedures for:

- Immediately inspecting (visually and with dye penetrant) the aileron and elevator hinge pins for cracks;
- Immediately replacing any hinge pins found with longitudinal cracks as a result of the above inspection; and
- —Eventually replacing any installed elevator or aileron hinge pins that are not P/N SY991A hinge pins with P/ N SY991A pins.

What action did the DGAC take? The DGAC classified this service bulletin as mandatory and issued French AD Number 2001–247(A), dated June 27, 2001, to ensure the continued airworthiness of these gliders in France.

Did the DGAC inform the United States under the bilateral airworthiness agreement? These CENTRAIR 101 series gliders are manufactured in France and are type-certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement.

Under this bilateral airworthiness agreement, the DGAC has kept us informed of the situation described above.

FAA's Determination and Requirements of This Proposed AD

What has FAA decided? We have examined the DGAC's findings, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Since the unsafe condition described previously is likely to exist or develop

on other CENTRAIR 101 series gliders of the same type design that are registered in the United States, we are proposing AD action to replace incorrectly heattreated elevator or aileron hinge pins, which could result in failure of the elevator or ailerons. Such failure during takeoff, landing, or flight operations could lead to loss of glider control.

What would this proposed AD require? This proposed AD would require you to replace with P/N SY991A hinge pins any installed elevator or aileron hinge pins that are not P/N SY991A hinge pins.

How does the revision to 14 CFR part 39 affect this proposed AD? On July 10, 2002, we published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs FAA's AD system. This regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Costs of Compliance

How many gliders would this proposed AD impact? We estimate that this proposed AD affects 57 gliders in the U.S. registry.

What would be the cost impact of this proposed AD on owners/operators of the affected gliders? We estimate the following costs to do this proposed elevator and aileron hinge pin replacement. We have no way of determining the number of gliders that may need this hinge pin replacement. However, we have presented the costs to reflect all 57 gliders needing the mandatory replacement:

Labor cost	Parts cost	Total cost per	Total cost on U.S. operators
4 workhours × \$65 per hour = \$260	\$1	\$261	\$261 × 57 = \$14,877

Regulatory Findings

Would this proposed AD impact various entities? We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the National Government and the States, or on the distribution of power and responsibilities among the various levels of government.

Would this proposed AD involve a significant rule or regulatory action? For

the reasons discussed above, I certify that this proposed AD:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this proposed AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under ADDRESSES. Include "AD Docket FAA—2004—19616; Directorate Identifier 2004—CE—38—AD" in your request.

This proposed rulemaking is promulgated under the authority in subtitle VII, part A, subpart III, section 44701, General requirements. Under that section, the FAA is charged with prescribing minimum standards required in the interest of safety for the design of aircraft. This proposed regulation is within the scope of that authority since it corrects an unsafe

condition in the design of the aircraft caused by incorrectly heat-treated elevator or aileron hinge pins, which could result in failure of the elevator or ailerons. Such failure during takeoff, landing, or flight operations could lead to loss of glider control.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

CENTRAIR: Docket No. FAA-2004-19616; Directorate Identifier 2004-CE-38-AD

When Is the Last Date I Can Submit Comments on This Proposed AD?

(a) We must receive comments on this proposed airworthiness directive (AD) by January 14, 2005.

What Other ADs Are Affected by This Action?

(b) None.

What Gliders Are Affected by This AD?

(c) This AD affects Models 101, 101A, 101AP, and 101P gliders, all serial numbers, without elevator and aileron part number SY991A hinge pins installed, certificated in any category.

What Is the Unsafe Condition Presented in This AD?

(d) This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for France. The actions specified in this AD are intended to replace incorrectly heat-treated elevator or aileron hinge pins, which could result in failure of the elevator or ailerons. Such failure during takeoff, landing, or flight operations could lead to loss of glider control.

What Must I Do To Address This Problem?

(e) To address this problem, you must do the following:

Actions	Compliance	Procedures	
 (1) Replace any installed elevator and aileron hinge pins that are not part number (P/N) SY991A hinge pins. (2) Do not install any elevator and aileron hinge pins that are not P/N SY991A hinge pins as specified in paragraph (e)(1) of this AD. 	after the effective date of this AD, unless already done. As of the effective date of this AD	Follow Société Nouvelle Centrair Service Bulletin No. 101–22, dated March 13, 2001. Not Applicable.	

May I Request an Alternative Method of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.19. Unless FAA authorizes otherwise, send your request to your principal inspector. The principal inspector may add comments and will send your request to the Manager, Standards Office, Small Airplane Directorate, FAA. For information on any already approved alternative methods of compliance, contact Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4130; facsimile: (816) 329–4090.

Is There Other Information That Relates to This Subject?

(g) French AD Number 2001–247(A), dated June 27, 2001, also addresses the subject of this AD.

May I Get Copies of the Documents Referenced in This AD?

(h) To get copies of the documents referenced in this AD, contact CENTRAIR, Aerodome B.P.N. 44, 36300 Le Blanc, France; telephone: 02.54.37.07.96; facsimile: 02.54.37.48.64. To view the AD docket, go to the Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL–401, Washington, DC, or on the Internet at http://dms.dot.gov. This is docket number FAA–2004–19616.

Issued in Kansas City, Missouri, on December 6, 2004.

Scott L. Sedgwick,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04–27196 Filed 12–10–04; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF HOMELAND SECURITY

Coast Guard

33 CFR Part 117

[CGD01-04-127]

RIN 2115-AE47

Drawbridge Operation Regulations; Shrewsbury River, NJ

AGENCY: Coast Guard, DHS.

ACTION: Notice of proposed rulemaking.

SUMMARY: The Coast Guard proposes to change the drawbridge operating regulations governing the operation of the Route 36 Bridge, mile 1.8, across the Shrewsbury River at Highlands, New Jersey. This proposed change to the drawbridge operation regulations would allow the bridge owner to require an advance notice for bridge openings during periods the bridge has received

few requests to open from 11 p.m. to 7 a.m., each day, and all day during the winter months December 1 through March 31. This action is expected to help relieve the bridge owner from the burden of crewing the bridge at all times while continuing to meet the present needs of navigation.

DATES: Comments must reach the Coast Guard on or before February 11, 2005. ADDRESSES: You may mail comments to Commander (obr), First Coast Guard District, Bridge Branch, One South Street, Battery Park Building, New York, New York, 10004, or deliver them to the same address between 7 a.m. and 3 p.m., Monday through Friday, except, Federal holidays. The telephone number is (212) 668-7165. The First Coast Guard District, Bridge Branch, maintains the public docket for this rulemaking. Comments and material received from the public, as well as documents indicated in this preamble as being available in the docket, will become part of this docket and will be available for inspection or copying at the First Coast Guard District, Bridge Branch, 7 a.m. to 3 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Joe Arca, Project Officer, First Coast Guard District, (212) 668–7165.

SUPPLEMENTARY INFORMATION: