

AFM by doing the actions specified in paragraphs (h)(2)(i), (h)(2)(ii), (h)(2)(iii), and (h)(2)(iv) of this AD.

(i) Delete the procedure “WARNING HORN—CABIN ALTITUDE OR CONFIGURATION” added by AD 2006–13–13, Amendment 39–14666. If the title of this procedure has been changed according to FAA alternative method of compliance (AMOC) letter 130S–09–

134a, dated April 28, 2009, delete the procedure that was approved according to this AMOC letter.

(ii) Delete the procedure entitled “CABIN ALTITUDE WARNING OR RAPID DEPRESSURIZATION” added by AD 2003–14–08, Amendment 39–13227.

(iii) If the procedure entitled “CABIN ALTITUDE (Airplanes with the CABIN ALTITUDE lights installed)” is currently

contained in the applicable Boeing 737 AFM, delete the procedure entitled “CABIN ALTITUDE (Airplanes with the CABIN ALTITUDE lights installed).”

(iv) Add the following statement. This may be done by inserting a copy of this AD into the applicable AFM.

“CABIN ALTITUDE WARNING OR RAPID DEPRESSURIZATION

Condition: The CABIN ALTITUDE warning light illuminates or the intermittent warning horn sounds in flight above 10,000 ft MSL.

RECALL

Oxygen Masks and Regulators ..... ON, 100%.  
Crew Communications ..... ESTABLISH.

REFERENCE

Pressurization Mode Selector ..... MANUAL.  
Outflow Valve Switch ..... CLOSE.  
If Cabin Altitude is uncontrollable:  
Emergency Descent (If Required) ..... INITIATE.  
Passenger Oxygen Switch ..... ON.  
Thrust Levers ..... CLOSE.  
Speed Brakes ..... FLIGHT DETENT.  
Target Speed ..... VMO/MMO.”

(3) Revise the Normal Procedures Section of the applicable Boeing 737 AFM by doing the actions specified in paragraphs (h)(3)(i) and (h)(3)(ii) of this AD.

(i) Delete the “CABIN ALTITUDE WARNING TAKEOFF BRIEFING” procedure added by AD 2008–23–07.

(ii) Add the following statement. This may be done by inserting a copy of this AD into the applicable AFM.

“For normal operations, the pressurization mode selector should be in AUTO prior to takeoff.

**Note 1:** When statements identical to those specified in paragraphs (h)(2)(iv) and (h)(3)(ii) of this AD have been included in the general revisions of the AFM, the general revisions may be inserted into the AFM, and the copies of this AD may be removed from the AFM.

**Terminating Action for Affected ADs**

(i) Accomplishment of the requirements of this AD terminates the requirements of the ADs identified in paragraphs (i)(1), (i)(2), and (i)(3) of this AD for only the airplanes identified in paragraph (c) of this AD.

(1) AD 2003–14–08: The requirements specified in Table 1 and Figure 1 of that AD.

(2) AD 2008–23–07: All requirements of that AD.

(3) AD 2006–13–13: All requirements of that AD.

**Special Flight Permit**

(j) Special flight permits, as described in Section 21.197 and Section 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199), are not allowed.

**Alternative Methods of Compliance (AMOCs)**

(k)(1) The Manager, Seattle ACO, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD. Information may be e-mailed to: *9-ANM-Seattle-ACO-AMOC-Requests@faa.gov*.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

**Related Information**

(l) For more information about this AD, contact Jeffrey W. Palmer, Aerospace Engineer, Systems and Equipment Branch, ANM–130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; *phone:* (425) 917–6472; *fax:* (425) 917–6590; *e-mail:* *jeffrey.w.palmer@faa.gov*.

(m) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H–65, Seattle, Washington 98124–2207; telephone 206–544–5000, extension 1, fax 206–766–5680; e-mail *me.boecom@boeing.com*; Internet *https://www.myboeingfleet.com*. You may review copies of the referenced

service information at the FAA, Transport Airplane Directorate, the FAA, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on March 14, 2011.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2011–6931 Filed 3–23–11; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA–2011–0257; Directorate Identifier 2010–NM–122–AD]

**RIN 2120–AA64**

**Airworthiness Directives; Airbus Model A318, A319, A320, and A321 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above that would supersede an existing AD. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an

aviation product. The MCAI describes the unsafe condition as:

\* \* \* \* \*

The issue 10 of Airbus A318/A319/A320/A321 ALI [Airworthiness Limitation Items] Document and issue 2 of Airbus A319 Corporate Jet ALI Document introduce more restrictive maintenance requirements/airworthiness limitations. Failure to comply with this issue 10 constitutes an unsafe condition. \* \* \*

\* \* \* \* \*

The unsafe condition is fatigue cracking, accidental damage, or corrosion in principal structural elements and possible failure of certain life limited parts, which could result in reduced structural integrity of the airplane. The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** We must receive comments on this proposed AD by May 9, 2011.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* (202) 493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Airbus, Airworthiness Office—EAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The

street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tim Dulin, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2141; fax (425) 227-1149; [tim.dulin@faa.gov](mailto:tim.dulin@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2011-0257; Directorate Identifier 2010-NM-122-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

##### Discussion

On September 21, 2007, we issued AD 2007-20-05, Amendment 39-15215 (72 FR 56262, October 3, 2007). That AD required operators of Airbus Model A318, A319, A320, and A321 series airplanes to revise the Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness to incorporate Sub-part 1-2, “Life Limits,” and Sub-part 1-3, “Demonstrated Fatigue Lives,” of Airbus A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006, and for certain airplanes, Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006. These actions were intended to address an unsafe condition on the products listed above.

Since we issued AD 2007-20-05, we have determined that more restrictive limitations are necessary. We have also added Model A318-121 and -122 airplanes to the applicability. The European Aviation Safety Agency (EASA), which is the Technical Agent

for the Member States of the European Community, has issued EASA Airworthiness Directive 2010-0071R1, dated May 28, 2010 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

The airworthiness limitations are currently included in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS).

The airworthiness limitations applicable to the Damage Tolerant Airworthiness Limitation Items (DT ALI) are currently given in Airbus A318/A319/A320/A321 ALI Document reference AI/SE-M4/95A.0252/96 and Airbus A319 Corporate Jet ALI Document reference AI/SE-M2/95A.1038/99, which are approved by the European Aviation Safety Agency (EASA) and referenced in Airbus Airworthiness Limitations Section (ALS) Part 2.

The issue 10 of Airbus A318/A319/A320/A321 ALI Document and issue 2 of Airbus A319 Corporate Jet ALI Document introduce more restrictive maintenance requirements/airworthiness limitations. Failure to comply with this issue 10 constitutes an unsafe condition.

EASA AD 2010-0071 retains the requirements of EASA AD 2006-0165, which is superseded, and requires the implementation of more restrictive maintenance requirements/airworthiness limitations as specified in Airbus A318/A319/A320/A321 ALI Document reference AI/SE-M4/95A.0252/96 issue 10 and Airbus A319 Corporate Jet ALI Document reference AI/SE-M2/95A.1038/99.

This AD has been revised to clarify the special compliance times defined in Table 1 of this AD.

You may obtain further information by examining the MCAI in the AD docket.

#### Relevant Service Information

Airbus has issued A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009; and A319 Corporate Jet Airworthiness Limitation Items, Document reference AI/SE-M2/95A.1038/99, Issue 02, dated March 2009. A319 Corporate Jet Airworthiness Limitation Items, Document AI/SE-M2/95A.1038/99, Issue 02, dated March 2009, states that the limitations for the Model A319 corporate jets are specified in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

#### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another

country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.

#### Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 729 products of U.S. registry.

The actions that are required by AD 2007–20–05 and retained in this proposed AD take about 1 work-hour per product, at an average labor rate of \$85 per work hour. Based on these figures, the estimated cost of the currently required actions is \$85 per product.

We estimate that it would take about 1 work-hour per product to comply with the new basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$61,965, or \$85 per product.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with

promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
- (3) Will not affect intrastate aviation in Alaska, and
- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment 39–15215 (72 FR 56262, October 3, 2007) and adding the following new AD:

**Airbus:** Docket No. FAA–2011–0257;  
Directorate Identifier 2010–NM–122–AD.

##### Comments Due Date

- (a) We must receive comments by May 9, 2011.

##### Affected ADs

- (b) This AD supersedes AD 2007–20–05, Amendment 39–15215.

#### Applicability

(c) This AD applies to all Airbus Model A318–111, –112, –121, and –122 airplanes; A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes; Model A320–111, –211, –212, –214, –231, –232, and –233 airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes; certificated in any category.

**Note 1:** This AD requires revisions to certain operator maintenance documents to include new inspections. Compliance with these inspections is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these inspections, the operator may not be able to accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to paragraph (n) of this AD. The request should include a description of changes to the required inspections that will ensure the continued damage tolerance of the affected structure. The FAA has provided guidance for this determination in Advisory Circular (AC) 25.1529–1.

#### Subject

(d) Air Transport Association (ATA) of America Code 05: Wings.

#### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

\* \* \* \* \*

The issue 10 of Airbus A318/A319/A320/A321 ALI [Airworthiness Limitation Items] Document and issue 2 of Airbus A319 Corporate Jet ALI Document introduce more restrictive maintenance requirements/airworthiness limitations. Failure to comply with this issue 10 constitutes an unsafe condition. \* \* \*

\* \* \* \* \*

The unsafe condition is fatigue cracking, accidental damage, or corrosion in principal structural elements and possible failure of certain life limited parts, which could result in reduced structural integrity of the airplane.

#### Compliance

(f) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Restatement of Requirements of AD 2007–20–05

#### Revise Airworthiness Limitations Section (ALS) To Incorporate Safe Life ALIs

(g) For Airbus Model A318–111, and –112 airplanes; A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes; Model A320–111, –211, –212, –214, –231, –232, and –233 airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes: Within 3 months after November 7, 2007 (the effective date of AD 2007–20–05), revise the ALS of the Instructions for Continued Airworthiness to incorporate Subpart 1–2, "Life Limits," and Subpart 1–3, "Demonstrated Fatigue Lives," of Airbus

A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006. Accomplish the actions in Sub-part 1–2, “Life Limits,” and Sub-part 1–3, “Demonstrated Fatigue Lives,” of Airbus A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006, at the times specified in Sub-part 1–2, “Life Limits,” and Sub-part 1–3, “Demonstrated Fatigue Lives,” of Airbus A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006, except as provided by paragraph (i) of this AD.

#### Revise ALS To Incorporate Damage-Tolerant ALLs

(h) For Airbus Model A318–111, and –112 airplanes; A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes; Model A320–111, –211, –212, –214, –231, –232, and –233 airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes, except Model A319 airplanes on which Airbus Modifications 28238, 28162, and 28342 have been incorporated in production: Within 14 days after November 7, 2007, revise the ALS of the Instructions for Continued Airworthiness to incorporate Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 7, dated December 2005 (approved by the EASA on February 7, 2006); Issue 08, dated March 2006 (approved by the EASA on January 4, 2007); or Issue 09, dated November 2006 (approved by the EASA on May 21, 2007). Accomplish the actions in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96,

Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006; at the times specified in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006; as applicable; except as provided by paragraph (i) of this AD. Doing the actions required by paragraph (j) of this AD terminates the requirements of this paragraph.

#### Grace Period for New or More Restrictive Actions

(i) For Airbus Model A318–111, and –112 airplanes; A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes; Model A320–111, –211, –212, –214, –231, –232, and –233 airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes: For any new or more restrictive life limit introduced with Sub-part 1–2, “Life Limits,” and Sub-part 1–3, “Demonstrated Fatigue Lives,” of Airbus A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006, replace the part at the time specified in Sub-part 1–2, “Life Limits,” and Sub-part 1–3, “Demonstrated Fatigue Lives,” of Airbus A318/A319/A320/A321 ALS Part 1—Safe Life Airworthiness Limitation Items, dated February 28, 2006, or within 6 months after November 7, 2007, whichever is later. For any new or more restrictive inspection introduced with Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006; do

the inspection at the time specified Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006; as applicable; or within 6 months after November 7, 2007, whichever is later.

#### New Requirements of This AD

##### Revise ALS To Incorporate Damage-Tolerant ALLs With Revised Compliance Times

(j) Within 9 months after the effective date of this AD: Revise the maintenance program by incorporating all maintenance requirements and associated airworthiness limitations specified in the Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 10, dated October 2009. Comply with all applicable maintenance requirements and associated airworthiness limitations included in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 10, dated October 2009, except as provided by paragraph (k) of this AD. Doing the actions required by this paragraph terminates the requirements of paragraph (h) of this AD.

##### Special Compliance Times for Certain Tasks

(k) For new and more restrictive tasks introduced with Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 10, dated October 2009, as specified in Table 1 of this AD: The initial compliance time for doing the tasks is specified in Table 1 of this AD.

TABLE 1—COMPLIANCE TIMES FOR NEW TASKS

Task	Applicability (as specified in the applicability column of the task)	Compliance time, whichever occurs later	
545102–01–6 .....	Group 19–1A CFM, Group 19–1B CFM, and A320–200 CFM/IAE.	The threshold as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 10, dated October 2009.	Within 2,000 flight cycles or 5,500 flight hours, after the effective date of this AD, whichever occurs first.
545102–01–7 .....	A320–100 .....	The threshold as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 10, dated October 2009.	Within 2,000 flight cycles or 2,000 flight hours, after the effective date of this AD, whichever occurs first.
572050–01–1 or alternative task 572050–02–1.	Group 19–1A and Group 19–1B ..	At the time of the next due accomplishment of any one of the tasks 572004, 572020, or 572053 as currently described in the Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE–M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.	Within 6 months after the effective date of this AD.

TABLE 1—COMPLIANCE TIMES FOR NEW TASKS—Continued

572050-01-4 or alternative task 572050-02-4.	A320-200 .....	At the time of the next due accomplishment of any one of the tasks 572004, 572020, or 572053 as currently described in the Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.	Within 6 months after the effective date of this AD.
572050-01-5 or alternative task 572050-02-5.	Group 21-1A .....	At the time of the next due accomplishment of any one of the tasks 572004, 572020, or 572053 as currently described in the Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.	Within 6 months after the effective date of this AD.
572050-01-7 or alternative task 572050-02-7.	A320-100 .....	At the time of the next due accomplishment of any one of the tasks 572004, 572020, or 572053 as currently described in the Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.	Within 6 months after the effective date of this AD.
534132-01-1 .....	A320 PRE 30748 .....	The threshold/interval as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009.	Within 100 days after the effective date of this AD, without exceeding the previous threshold/interval as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.
531118-01-1 .....	A318 (except (A318-121 and -122), Group 19-1A, Group 19-1B, A320, A321.	The threshold/interval as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009.	Within 100 days after the effective date of this AD, without exceeding the previous threshold/interval as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Issue 08, dated March 2006; or Issue 09, dated November 2006.
531118-01-1 .....	A318-121 and -122 .....	The threshold/interval as defined in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009.	Within 100 days after the effective date of this AD.

**Note 2:** New ALI Task 572050 refers to the outer wing dry bay and is comprised of extracts from three ALI Tasks: 572004, 572020 and 572053. The threshold of ALI Task 572050 for the whole dry bay area is that of the lowest threshold of the source ALI tasks, i.e., that of ALI Task 572053.

**No Alternative Life Limits, Inspections, or Inspection Intervals**

(l) After the actions specified in paragraphs (g) and (h) of this AD have been

accomplished, no alternative life limits, inspections, or inspection intervals may be used, except as provided by paragraphs (i) and (m) of this AD, and except as required by paragraph (j) of this AD.

(m) After the actions specified in paragraph (j) of this AD have been accomplished, no alternative life limits, inspections, or inspection intervals may be used.

**FAA AD Differences**

**Note 3:** This AD differs from the MCAI and/or service information as follows: EASA AD requires operators to comply with the limitations specified in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009; or Airbus A319 Corporate Jet Airworthiness Limitation Items, Document AI/SE-M2/95A.1038/99, Issue 02, dated March 2009; as applicable. However,

this AD does not require incorporating Airbus A319 Corporate Jet Airworthiness Limitation Items, Document AI/SE-M2/95A.1038/99, Issue 02, dated March 2009, because that ALI only specifies compliance with the limitations specified in Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009.

#### Other FAA AD Provisions

(n) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch/ACO, send it to *Attn: Tim Dulin, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2141; fax (425) 227-1149. Information may be e-mailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov.* Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### Related Information

(o) Refer to MCAI European Aviation Safety Agency (EASA) Airworthiness Directive 2010-0071R1, dated May 28, 2010; Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 7, dated December 2005; Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 08, dated March 2006; Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 09, dated November 2006; and Airbus A318/A319/A320/A321 Airworthiness Limitation Items, Document AI/SE-M4/95A.0252/96, Issue 10, dated October 2009; for related information.

Issued in Renton, Washington, on March 15, 2011.

**Kalene C. Yanamura,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

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**BILLING CODE 4910-13-P**

## COMMODITY FUTURES TRADING COMMISSION

### 17 CFR Part 39

**RIN 3038-AC98**

### Risk Management Requirements for Derivatives Clearing Organizations

**AGENCY:** Commodity Futures Trading Commission.

**ACTION:** Reopening of comment period.

**SUMMARY:** The Commodity Futures Trading Commission (Commission) is reopening the comment period for a proposed rule that would require derivatives clearing organizations (DCOs) to report end-of-day positions for each clearing member, by customer origin and house origin, and for customer origin, separately, the gross positions of each beneficial owner.

**DATES:** Submit comments on or before April 25, 2011.

**ADDRESSES:** You may submit comments, identified by RIN number 3038-AC98, by any of the following methods:

- *Agency Web site, via its Comments Online process:* <http://comments.cftc.gov>. Follow the instructions for submitting comments through the Web site.
- *Mail:* David A. Stawick, Secretary of the Commission, Commodity Futures Trading Commission, Three Lafayette Centre, 1155 21st Street, NW., Washington, DC 20581.
- *Hand Delivery/Courier:* Same as mail above.
- *Federal eRulemaking Portal:* <http://www.Regulations.gov>. Follow the instructions for submitting comments. Please submit comments by only one method.

All comments must be submitted in English, or if not, accompanied by an English translation. Comments will be posted as received to <http://www.cftc.gov>. You should submit only information that you wish to make available publicly. If you wish the Commission to consider information that may be exempt from disclosure under the Freedom of Information Act (FOIA), a petition for confidential treatment of the exempt information may be submitted according to the procedures established in § 145.9 of the Commission's regulations.<sup>1</sup> The Commission reserves the right, but shall have no obligation, to review, pre-screen, filter, redact, refuse, or remove any or all of your submission from

<sup>1</sup> Commission regulations referred to herein are found at 17 CFR Ch. 1 (2010). They are accessible on the Commission's Web site at <http://www.cftc.gov>.

<http://www.cftc.gov> that it may deem to be inappropriate for publication, such as obscene language. All submissions that have been redacted or removed that contain comments on the merits of the rulemaking will be retained in the public comment file and will be considered as required under the Administrative Procedure Act and other applicable laws, and may be accessible under FOIA.

#### FOR FURTHER INFORMATION CONTACT:

Phyllis P. Dietz, Associate Director, 202-418-5449, [pdietz@cftc.gov](mailto:pdietz@cftc.gov), Jacob Preiserowicz, Attorney-Advisor, 202-418-5432, [jpreiserowicz@cftc.gov](mailto:jpreiserowicz@cftc.gov), Division of Clearing and Intermediary Oversight, Commodity Futures Trading Commission, Three Lafayette Centre, 1155 21st Street, NW., Washington, DC 20581; and Anne C. Polaski, Special Counsel, 312-596-0575, [apolaski@cftc.gov](mailto:apolaski@cftc.gov), Division of Clearing and Intermediary Oversight, Commodity Futures Trading Commission, 525 West Monroe Street, Chicago, Illinois 60661.

**SUPPLEMENTARY INFORMATION:** On July 21, 2010, President Obama signed the Dodd-Frank Act.<sup>2</sup> Title VII of the Dodd-Frank Act<sup>3</sup> amended the Commodity Exchange Act (CEA)<sup>4</sup> to establish a comprehensive regulatory framework to reduce risk, increase transparency, and promote market integrity within the financial system. Section 725(c) of the Dodd-Frank Act amended Section 5b(c)(2) of the CEA, which sets forth core principles with which a DCO must comply to be registered and to maintain registration as a DCO.<sup>5</sup>

On December 16, 2010, the Commission approved for publication in the **Federal Register** proposed regulations which, among other things, would implement DCO Core Principle D (Risk Management) and would establish a related reporting requirement under Core Principle J (Reporting). More specifically, the Commission proposed § 39.13(g)(8)(i) to establish customer "gross margin" requirements, and proposed § 39.19(c)(1)(iv) to establish related daily reporting requirements. The proposed regulations were published for comment in the January 20, 2011 issue of the **Federal Register**.<sup>6</sup>

The **Federal Register** preamble explained that proposed § 39.13(g)(8)(i)

<sup>2</sup> See Dodd-Frank Wall Street Reform and Consumer Protection Act, Public Law 111-203, 124 Stat. 1376 (2010). The text of the Dodd-Frank Act may be accessed at <http://www.cftc.gov/LawRegulation/OTCDERIVATIVES/index.htm>.

<sup>3</sup> Pursuant to section 701 of the Dodd-Frank Act, Title VII may be cited as the "Wall Street Transparency and Accountability Act of 2010."

<sup>4</sup> 7 U.S.C. 1 *et seq.*

<sup>5</sup> 7 U.S.C. 5b(c)(2).

<sup>6</sup> 76 FR 3698, Jan. 20, 2011.