

Issued in Burlington, Massachusetts, on March 7, 2007.

Peter A. White,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2006-26497; Directorate Identifier 2006-CE-082-AD; Amendment 39-14989; AD 2007-06-08]

RIN 2120-AA64

Airworthiness Directives; Przedsiębiorstwo Doswiadczalno- Produkcyjne Szybownictwa "PZL- Bielsko" Model SZD-50-3 "Puchacz" Gliders

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Some cases of turnbuckle adjusting screws fatigue failure have occurred, due to lateral load component applied by pilot's foot. Such events may lead to rudder and pedals disconnection.

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective April 19, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of April 19, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Gregory Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; *telephone:* (816) 329-4130; *fax:* (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Streamlined Issuance of AD

The FAA is implementing a new process for streamlining the issuance of ADs related to MCAI. The streamlined process will allow us to adopt MCAI safety requirements in a more efficient manner and will reduce safety risks to the public. This process continues to follow all FAA AD issuance processes to meet legal, economic, Administrative Procedure Act, and **Federal Register** requirements. We also continue to meet our technical decision-making responsibilities to identify and correct unsafe conditions on U.S.-certificated products.

This AD references the MCAI and related service information that we considered in forming the engineering basis to correct the unsafe condition. The AD contains text copied from the MCAI and for this reason might not follow our plain language principles.

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on January 5, 2007 (72 FR 485). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states that some cases of turnbuckle adjusting screws fatigue failure have occurred, due to lateral load component applied by pilot's foot. Such events may lead to rudder and pedals disconnection.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies.

Any such differences are highlighted in a NOTE within the AD.

Costs of Compliance

We estimate that this AD will affect 8 products of U.S. registry. We also estimate that it will take about 2 work-hours per product to comply with basic requirements of this AD. The average labor rate is \$80 per work-hour. Required parts will cost about \$100 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$2,080, or \$260 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD Docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5227) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2007-06-08 Przedsiębiorstwo Doswiadczalno-Produkcyjne Szybownictwa "PZL-Bielsko":
Amendment 39-14989; Docket No. FAA-2006-26497; Directorate Identifier 2006-CE-082-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective April 19, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Model SZD-50-3 "Puchacz" Gliders, all serial numbers, certificated in any category.

Reason

(d) The mandatory continuing airworthiness information (MCAI) states:

Some cases of turnbuckle adjusting screws fatigue failure have occurred, due to lateral load component applied by pilot's foot. Such events may lead to rudder and pedals disconnection.

Actions and Compliance

(e) Unless already done, within the next 3 calendar months after April 19, 2007 (the

effective date of this AD), install the extra pull rod between the rear pedals and turnbuckle adjusting screws following Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, except as specified in paragraphs (e)(1) through (e)(4) of this AD. For owners/operators that have installed an additional short cable between the rear seat pedal and turnbuckle prior to Allstar PZL's issuance of Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, this additional short cable assembly must comply with the requirements of Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006. Upon completion, a logbook entry is required.

(1) Paragraph 1 of Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, describes the dimension length of the extra segment pull rod to be 140 mm. Modify this to read: "140 mm (5.5118 inches)."

(2) Paragraph 4 of Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, describes the dimensions of the short pull rod to be 3 mm diameter core and approximately 140 mm. Modify this to read: "3 mm (0.1181 inch) and 140 mm (5.5118 inches)."

(3) Paragraph 4.4 of Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, describes a 1 mm diameter cotter pin. Modify this to read: "1 mm (0.03937 inch)."

(4) Paragraph 5 of Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, reads, "The parts necessary for modification are available at Allstar PZL Glider, or substitute aircraft parts may be used—capable to withstand a load of 6100N at minimum." Change this to read: "The parts necessary for modification are available at Allstar PZL Glider, or substitute aircraft parts may be used—capable to withstand a load of 6100N (1,372 lbs) at minimum. If a substitute part is used, the hole diameter specified in Figure 1 of the service bulletin as 'Ø 6 Hg' means a 6 mm (0.2362 inch) diameter hole with a dimensional tolerance of +0.03 mm (+0.0012 inch). Contact the manufacturer for further details."

FAA AD Differences

Note: This AD differs from the MCAI and/or service information as follows: Paragraphs (e)(1) through (e)(4) of this AD have been added to clarify certain procedures in the service bulletin.

Other FAA AD Provisions

(f) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Staff, FAA, ATTN: Gregory Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; fax: (816) 329-4090, has the authority to approve

AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Material Incorporated by Reference

(g) You must use Allstar PZL Glider Sp. Z o.o. Mandatory Service Bulletin No. BE-057/SZD-50-3/2006 "PUCHACZ", dated October 16, 2006, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Allstar PZL Glider Sp. z o.o., ul. Cieszyńska 325, 43 300 Bielsko-Biala; telephone: +48 (0)33 8125021; fax: +48 (0)33 8123739; e-mail: office@szd.com.pl.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Kansas City, Missouri, on March 7, 2007.

David R. Showers,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

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