

**§ 39.13 [Amended]**

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2021-07-17 The Boeing Company:**

Amendment 39-21494 ; Docket No. FAA-2019-1071; Product Identifier 2019-NM-165-AD.

**(a) Effective Date**

This airworthiness directive (AD) is effective May 25, 2021.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to all The Boeing Company Model 737-900ER series airplanes, certificated in any category.

**(d) Subject**

Air Transport Association (ATA) of America Code 24, Electrical power.

**(e) Unsafe Condition**

This AD was prompted by reports of significant corrosion of electrical connectors located in the main landing gear (MLG) wheel well. The FAA is issuing this AD to address corrosion and subsequent moisture ingress that may lead to electrical shorting of the connectors and incorrect functioning of critical systems necessary for safe flight and landing.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

Within 24 months after the date of issuance of the original airworthiness certificate or the date of issuance of the original export certificate of airworthiness or 12 months after the effective date of this AD, whichever is later: Do the actions required by paragraph (g)(1) or (2) of this AD.

(1) Determine airplane exposure to runway deicing fluids containing potassium formate or potassium acetate by reviewing airport data on the types of components in the deicing fluid used at airports that support airplane operations.

(i) If the airplane has not been exposed: Repeat the requirements specified in paragraph (g)(1) of this AD thereafter at intervals not to exceed 24 months.

(ii) If the airplane has been exposed: Within 90 days after that determination is made, do the inspection required by paragraph (g)(2) of this AD. Repeat the inspection thereafter at intervals not to exceed 24 months.

(2) Do a detailed inspection of the electrical connectors, including the contacts and backshells of the line replaceable unit (LRU) in the wheel well of the MLG, for corrosion in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 737-24A1148, Revision 2, dated September 14, 2020. Perform applicable corrective actions at the applicable times, as specified in paragraphs (g)(2)(i) through (iii) of this AD, in

accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 737-24A1148, Revision 2, dated September 14, 2020. Repeat the inspection thereafter at intervals not to exceed 24 months. For the purposes of this AD, a detailed inspection is defined as an intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required.

(i) If the total backshell surface area corrosion is 10 percent or less, clean the backshell(s) before further flight.

(ii) If the total backshell surface area corrosion is greater than 10 percent but less than 20 percent, replace the connectors and backshells within 30 days after the detailed inspection.

(iii) If the total backshell surface area corrosion is 20 percent or more, replace the connectors and backshells before further flight.

**(h) Credit for Previous Actions**

This paragraph provides credit for the initial detailed inspection and applicable corrective actions specified in paragraph (g)(2) of this AD, if those actions were performed before the effective date of this AD using Boeing Alert Service Bulletin 737-24A1148, dated December 6, 2001, or Boeing Alert Service Bulletin 737-24A1148, Revision 1, dated July 10, 2003.

**(i) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Seattle ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: [9-ANM-Seattle-ACO-AMOC-Requests@faa.gov](mailto:9-ANM-Seattle-ACO-AMOC-Requests@faa.gov).

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by The Boeing Company Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO Branch, FAA, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

(4) FAA AMOC Letter 130S-09-9, dated April 28, 2009, and AMOCs approved previously for AD 2005-18-23, Amendment 39-14264 (70 FR 54253, September 14, 2005), are approved as AMOCs for the corresponding provisions of this AD.

**(j) Related Information**

For more information about this AD, contact Julio C. Alvarez, Aerospace Engineer, Systems and Equipment Section, FAA, Seattle ACO Branch, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3657; email: [julio.c.alvarez@faa.gov](mailto:julio.c.alvarez@faa.gov).

**(k) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Service Bulletin 737-24A1148, Revision 2, dated September 14, 2020.

(ii) [Reserved]

(3) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740-5600; telephone 562-797-1717; internet <https://www.myboeingfleet.com>.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on March 29, 2021.

**Lance T. Gant,**

*Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2021-08058 Filed 4-19-21; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2020-1167; Project Identifier AD-2020-01007-T; Amendment 39-21504; AD 2021-08-10]

**RIN 2120-AA64**

**Airworthiness Directives; The Boeing Company Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all The Boeing Company Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, DC-10-30F (KC-10A and KDC-10), DC-10-40

and DC-10-40F airplanes; and Model MD-10-10F and MD-10-30F airplanes. This AD was prompted by a report that an operator found a crack in the upper flange of the pylon aft bulkhead bracket. This AD requires a general visual inspection of the left and right wing pylon at the aft bulkhead bracket for any lockbolt and collar; repetitive surface and open hole eddy current high frequency (ETHF) inspections of the left and right wing pylon at the aft bulkhead bracket for any cracking; and applicable on-condition actions. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective May 25, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of May 25, 2021.

**ADDRESSES:** For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740-5600; telephone 562-797-1717; internet <https://www.myboeingfleet.com>. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1167.

#### Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1167; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal

holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

#### FOR FURTHER INFORMATION CONTACT:

Manuel Hernandez, Aerospace Engineer, Airframe Section, FAA, Los Angeles ACO Branch, 3960 Paramount Boulevard, Lakewood, CA 90712-4137; phone: 562-627-5256; fax: 562-627-5210; email: [Manuel.F.Hernandez@faa.gov](mailto:Manuel.F.Hernandez@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all The Boeing Company Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, DC-10-30F (KC-10A and KDC-10), DC-10-40 and DC-10-40F airplanes; and Model MD-10-10F and MD-10-30F airplanes. The NPRM published in the **Federal Register** on January 15, 2021 (86 FR 3885). The NPRM was prompted by a report that an operator found a crack in the upper flange of the pylon aft bulkhead bracket. In the NPRM, the FAA proposed to require a general visual inspection of the left and right wing pylon at the aft bulkhead bracket for any lockbolt and collar; repetitive surface and open hole ETHF inspections of the left and right wing pylon at the aft bulkhead bracket for any cracking; and applicable on-condition actions. The FAA is issuing this AD to address possible cracking of the wing pylon at the aft bulkhead bracket, which could result in the inability of the pylon to sustain limit load and adversely affect the structural integrity of the airplane.

#### Discussion of Final Airworthiness Directive

##### Comments

The FAA received a comment from The Air Line Pilots Association, International (ALPA) who supported the NPRM without change.

##### Conclusion

The FAA reviewed the relevant data, considered any comments received, and determined that air safety requires adopting this AD as proposed. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

#### Related Service Information Under 1 CFR Part 51

The FAA reviewed Boeing Alert Requirements Bulletin DC10-54A111 RB, dated June 26, 2020. The service information describes procedures for a general visual inspection of the left and right wing pylon at the aft bulkhead bracket for any lockbolt and collar; repetitive surface and open hole ETHF inspections of the left and right wing pylon at the aft bulkhead bracket for any cracking; and applicable on-condition actions. On-condition actions include modifying any aft bulkhead bracket that has a lockbolt and collar, and repair or replacement of the aft bulkhead bracket.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in **ADDRESSES**.

#### Costs of Compliance

The FAA estimates that this AD affects 103 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

#### ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
General visual inspection .....	2 work-hours × \$85 per hour = \$170 .....	\$0	\$170 .....	\$17,510.
Surface and open hole ETHF inspections.	5 work-hours × \$85 per hour = \$425 per inspection cycle.	0	\$425 per inspection cycle.	\$8,755 per inspection cycle.

The FAA estimates the following costs to do any necessary on-condition

modifications that would be required. The FAA has no way of determining the

number of aircraft that might need these on-condition modifications:

#### ESTIMATED COSTS OF ON-CONDITION ACTIONS

Labor cost	Parts cost	Cost per product
1 work-hour × \$85 per hour = \$85 per lockbolt/collar (maximum of 8 lockbolt/collars) .....	\$100 per lockbolt/collar ..	\$185 per lockbolt/collar.

The FAA has received no definitive data on which to base the cost estimates for the on-condition repairs and replacements specified in this AD.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Will not affect intrastate aviation in Alaska, and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2021–08–10 The Boeing Company:**  
Amendment 39–21504; Docket No. FAA–2020–1167; Project Identifier AD–2020–01007–T.

##### (a) Effective Date

This airworthiness directive (AD) is effective May 25, 2021.

##### (b) Affected ADs

None.

##### (c) Applicability

This AD applies to all The Boeing Company airplanes specified in paragraphs (c)(1) through (5) of this AD, certificated in any category.

(1) Model DC–10–10 and DC–10–10F airplanes.

(2) Model DC–10–15 airplanes.

(3) Model DC–10–30 and DC–10–30F (KC–10A and KDC–10) airplanes.

(4) Model DC–10–40 and DC–10–40F airplanes.

(5) Model MD–10–10F and MD–10–30F airplanes.

##### (d) Subject

Air Transport Association (ATA) of America Code 54, Nacelles/pylons.

##### (e) Unsafe Condition

This AD was prompted by a report that an operator found a crack in the upper flange of the pylon aft bulkhead bracket. The FAA is issuing this AD to address possible cracking of the wing pylon at the aft bulkhead bracket, which could result in the inability of the pylon to sustain limit load and adversely affect the structural integrity of the airplane.

##### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

##### (g) Required Actions

Except as specified by paragraph (h) of this AD: At the applicable times specified in the "Compliance" paragraph of Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020, do all applicable actions identified in, and in accordance with, the Accomplishment Instructions of Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020.

**Note 1 to paragraph (g):** Guidance for accomplishing the actions required by this AD can be found in Boeing Alert Service Bulletin DC10–54A111, dated June 26, 2020, which is referred to in Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020.

##### (h) Exceptions to Service Information Specifications

(1) Where Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020, uses the phrase "the original issue date of Requirements Bulletin DC10–54A111 RB," this AD requires using "the effective date of this AD."

(2) Where Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020, specifies contacting Boeing for repair, modification, or replacement instructions: This AD requires doing the repair, modification, or replacement using a method approved in accordance with the procedures specified in paragraph (i) of this AD.

##### (i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Los Angeles ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j)(1) of this AD. Information may be emailed to: 9-ANM-LAACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by The Boeing Company Organization Designation Authorization (ODA) that has been authorized by the Manager, Los Angeles ACO Branch, FAA, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

##### (j) Related Information

(1) For more information about this AD, contact Manuel Hernandez, Aerospace Engineer, Airframe Section, FAA, Los Angeles ACO Branch, 3960 Paramount Boulevard, Lakewood, CA 90712–4137; phone: 562–627–5256; fax: 562–627–5210; email: Manuel.F.Hernandez@faa.gov.

(2) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (k)(3) and (4) of this AD.

##### (k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Requirements Bulletin DC10–54A111 RB, dated June 26, 2020.

(ii) [Reserved]

(3) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110–SK57, Seal Beach, CA 90740–5600; telephone 562–797–1717; internet <https://www.myboeingfleet.com>.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the

availability of this material at the FAA, call 206–231–3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on April 1, 2021.

**Lance T. Gant,**

*Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2021–08055 Filed 4–19–21; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2020–0911; Product Identifier 2020–NM–075–AD; Amendment 39–21497; AD 2021–08–03]

**RIN 2120–AA64**

#### **Airworthiness Directives; MHI RJ Aviation ULC (Type Certificate Previously Held by Bombardier, Inc.) Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all MHI RJ Aviation ULC Model CL–600–2C10 (Regional Jet Series 700, 701 & 702) airplanes; Model CL–600–2C11 (Regional Jet Series 550) airplanes; Model CL–600–2D15 (Regional Jet Series 705) airplanes; Model CL–600–2D24 (Regional Jet Series 900) airplanes; and Model CL–600–2E25 (Regional Jet Series 1000) airplanes. This AD was prompted by a determination that a new or more restrictive airworthiness limitation is necessary. This AD requires revising the existing maintenance or inspection program, as applicable, to incorporate a new or more restrictive airworthiness limitation. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective May 25, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of May 25, 2021.

**ADDRESSES:** For service information identified in this final rule, contact MHI RJ Aviation ULC, 12655 Henri-Fabre Blvd., Mirabel, Québec J7N 1E1, Canada; Widebody Customer Response

Center North America toll-free telephone +1 844 272 2720 or direct-dial telephone +1 514 855 8500; fax +1–514 855–8501; email [thd.crj@mhjr.com](mailto:thd.crj@mhjr.com); internet <https://mhjr.com>. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0911.

#### **Examining the AD Docket**

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0911; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Siddeeq Bacchus, Aerospace Engineer, Mechanical Systems and Administrative Services Section, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7362; fax 516–794–5531; email [9-avs-nyaco-cos@faa.gov](mailto:9-avs-nyaco-cos@faa.gov).

#### **SUPPLEMENTARY INFORMATION:**

##### **Discussion**

Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued TCCA AD CF–2020–08, dated April 6, 2020 (also referred to as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for all MHI RJ Aviation ULC Model CL–600–2C10 (Regional Jet Series 700, 701 & 702) airplanes; Model CL–600–2C11 (Regional Jet Series 550) airplanes; Model CL–600–2D15 (Regional Jet Series 705) airplanes; Model CL–600–2D24 (Regional Jet Series 900) airplanes; and Model CL–600–2E25 (Regional Jet Series 1000) airplanes. You may examine the MCAI in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0911.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all MHI RJ Aviation ULC Model

CL–600–2C10 (Regional Jet Series 700, 701 & 702) airplanes; Model CL–600–2C11 (Regional Jet Series 550) airplanes; Model CL–600–2D15 (Regional Jet Series 705) airplanes; Model CL–600–2D24 (Regional Jet Series 900) airplanes; and Model CL–600–2E25 (Regional Jet Series 1000) airplanes. The NPRM published in the **Federal Register** on October 13, 2020 (85 FR 64417). The NPRM was prompted by a determination that a new or more restrictive airworthiness limitation is necessary. The NPRM proposed to require revising the existing maintenance or inspection program, as applicable, to incorporate a new or more restrictive airworthiness limitation. The FAA is issuing this AD to address failed telescopic ducts in the wing anti-ice system, which could result in loss of the wing anti-ice system function, slat skew, slat jam, structural damage to the slat panel, and loss of the slat panel, possibly resulting in reduced control of the airplane. See the MCAI for additional background information.

#### **Comments**

The FAA gave the public the opportunity to participate in developing this final rule. The following presents the comments received on the NPRM and the FAA's response to each comment.

#### **Support for the NPRM**

An anonymous commenter and Endeavor Air indicated support for the NPRM.

#### **Request To Extend the 60-Day Compliance Time for Initial Actions**

Endeavor Air requested a revision to the initial compliance times proposed in the NPRM for accomplishing the tasks. Endeavor Air suggested removing the phrase “or within 60 days after the effective date of the AD, whichever is later” and replacing it with “for aircraft with more than 15,200 FH [flight hours], phase in within 8,800 FH from the effective date of the AD.” Endeavor Air reasoned that the change would allow the work to be scheduled during C-check maintenance, and align with the original intent.

The FAA disagrees with replacing the 60-day grace period with the compliance time suggested by the commenter. The FAA reviewed the compliance times, typical fleet usage, and the TCCA AD, and found that the compliance time specified in paragraph (g) of this AD adequately addresses the unsafe condition. Most airplanes average 8.5 flight hours per day and will not reach another 8,800 flight hours for 32 months; therefore, an extension is