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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-0048; Directorate Identifier 2007-NM-181-AD; Amendment 39-15503; AD 2008-09-22]

RIN 2120-AA64

Airworthiness Directives; Construcciones Aeronauticas, S.A. (CASA), Model CN-235, CN-235-100, CN-235-200, CN-235-300, and C-295 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Subsequent to accidents involving Fuel Tank System explosions in flight * * * and on ground, * * * Special Federal Aviation Regulation 88 (SFAR88) * * * required a safety review of the aircraft Fuel Tank System * * *.

* * * * *

Fuel Airworthiness Limitations are items arising from a systems safety analysis that have been shown to have failure mode(s) associated with an 'unsafe condition' * * *. These are identified in Failure Conditions for which an unacceptable probability of ignition risk could exist if specific tasks and/or practices are not performed in accordance with the manufacturers' requirements.

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective June 5, 2008.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Shahram Daneshmandi, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1112; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on October 17, 2007 (72 FR 58770). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Subsequent to accidents involving Fuel Tank System explosions in flight * * * and on ground, the FAA published Special Federal Aviation Regulation 88 (SFAR 88) in June 2001. SFAR 88 required a safety review of the aircraft Fuel Tank System to determine that the design meets the requirements of FAR (Federal Aviation Regulation) § 25.901 and § 25.981(a) and (b).

A similar regulation has been recommended by the JAA (Joint Aviation Authorities) to the European National Aviation Authorities in JAA letter 04/00/02/07/03-L024 of 3 February 2003. The review was requested to be mandated by NAA's (National Aviation Authorities) using JAR (Joint Aviation Regulation) § 25.901(c), § 25.1309.

In August 2005 EASA published a policy statement on the process for developing instructions for maintenance and inspection of Fuel Tank System ignition source prevention (EASA D 2005/CPRO, www.easa.eu.int/home/cert_policy_statements_en.html) that also included the EASA expectations with regard to compliance times of the corrective actions on the unsafe and the not unsafe part of the harmonised design review results. On a global scale the TC (type certificate) holders committed themselves to the EASA published compliance dates (see EASA policy statement). The EASA policy statement has been revised in March 2006:

the date of 31-12-2005 for the unsafe related actions was set at 01-07-2006.

Fuel Airworthiness Limitations are items arising from a systems safety analysis that have been shown to have failure mode(s) associated with an 'unsafe condition' as defined in FAA's memo 2003-112-15 'SFAR 88—Mandatory Action Decision Criteria'. These are identified in Failure Conditions for which an unacceptable probability of ignition risk could exist if specific tasks and/or practices are not performed in accordance with the manufacturers' requirements.

This EASA Airworthiness Directive mandates the Fuel System Airworthiness Limitations (comprising maintenance/inspection tasks and Critical Design Configuration Control Limitations (CDCCL)) for the type of aircraft, that resulted from the design reviews and the JAA recommendation and EASA policy statement mentioned above.

The corrective action is revising the Airworthiness Limitations Section of the Instructions for Continued Airworthiness to include CDCCL data. You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Explanation of Changes to the AD

EASA Airworthiness Directive 2007-0007, dated January 9, 2007, specifies to incorporate into the aircraft maintenance program the fuel airworthiness limitation (FAL) maintenance and inspection tasks defined in EADS CASA CN-235/C-295 Technical Document DT-0-C00-05001, Issue C, dated October 2006. We cited both these documents in the NPRM and specified the FAL tasks in paragraph (f)(1) of the NPRM. However, further examination of the EADS CASA Technical Document has shown that it does not contain FAL maintenance and inspection tasks. Therefore, we have removed paragraph (f)(1) of the NPRM and re-identified the remaining paragraphs in this final rule. We have also removed Note 1 of the NPRM, which gave instructions regarding maintenance documents and new inspections. That note is no longer relevant in this final rule. We have also added a difference in Note 2 of this final rule to specify that we have not included the FAL action in this final

rule. If EADS CASA CN-235/C-295 Technical Document DT-0-C00-05001, Issue C, dated October 2006, is revised in the future to include the FAL tasks, or if these tasks are included in another document, we might consider additional rulemaking then.

This AD requires operators to revise the Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness to include CDCCL data. Operators must comply with the terms of the ALS, as specified in sections 43.16 (for persons maintaining products) and 91.403 (for operators) of the Federal Aviation Regulations (14 CFR 43.16 and 14 CFR 91.403). However, for the FAA to require compliance with any new or more restrictive life limits and inspections that the manufacturer or the FAA might impose in the ALS, we must engage in rulemaking; if we do not engage in rulemaking, the revised limitations in the maintenance manual cannot be made mandatory. We have, however, determined that it is appropriate in this case to allow accomplishing the ALS revision by incorporating the CDCCLs as defined in later revisions of EADS CASA CN-235/C-295 Technical Document DT-0-C00-05001, Issue C, dated October 2006. Therefore, we have revised paragraph (f)(3) of the NPRM (paragraph (f)(2) of this final rule) to allow later revisions of the EADS CASA technical document as acceptable methods of compliance if they are approved by the Manager, ANM-116, International Branch, Transport Airplane Directorate, FAA, or the European Aviation Safety Agency (EASA) (or its delegated agent).

We have also added Table 1 and a new Note 1 to this final rule to provide information about the EADS CASA Component Maintenance Manuals (CMMs) that are given as references in EADS CASA CN-235/C-295 Technical Document.

Conclusion

We reviewed the relevant data and determined that air safety and the public interest require adopting the AD with the changes described previously. We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S.

operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a NOTE within the AD.

Costs of Compliance

We estimate that this AD will affect about 8 products of U.S. registry. We also estimate that it will take about 1 work-hour per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$640, or \$80 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2008-09-22 Construcciones Aeronauticas, S.A. (CASA): Amendment 39-15503. Docket No. FAA-2007-0048; Directorate Identifier 2007-NM-181-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective June 5, 2008.

Affected ADs

(b) None.

Applicability

(c) This AD applies to all CASA Model CN-235, CN-235-100, CN-235-200, CN-235-300, and C-295 airplanes; certificated in any category.

Subject

(d) Air Transport Association (ATA) of America Code 28: Fuel.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

Subsequent to accidents involving Fuel Tank System explosions in flight * * * and on ground, the FAA published Special Federal Aviation Regulation 88 (SFAR 88) in June 2001. SFAR 88 required a safety review of the aircraft Fuel Tank System to determine that the design meets the requirements of FAR (Federal Aviation Regulation) § 25.901 and § 25.981(a) and (b).

A similar regulation has been recommended by the JAA (Joint Aviation Authorities) to the European National Aviation Authorities in JAA letter 04/00/02/07/03–L024 of 3 February 2003. The review was requested to be mandated by NAA's (National Aviation Authorities) using JAR (Joint Aviation Regulation) § 25.901(c), § 25.1309.

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statement has been revised in March 2006: The date of 31–12–2005 for the unsafe related actions was set at 01–07–2006.

Fuel Airworthiness Limitations are items arising from a systems safety analysis that have been shown to have failure mode(s) associated with an 'unsafe condition' as defined in FAA's memo 2003–112–15 'SFAR 88—Mandatory Action Decision Criteria'. These are identified in Failure Conditions for which an unacceptable probability of ignition risk could exist if specific tasks and/or practices are not performed in accordance with the manufacturers' requirements.

This EASA Airworthiness Directive mandates the Fuel System Airworthiness Limitations (comprising maintenance/inspection tasks and Critical Design Configuration Control Limitations (CDCCL)) for the type of aircraft, that resulted from the design reviews and the JAA recommendation and EASA policy statement mentioned above.

The corrective action is revising the Airworthiness Limitations Section of the Instructions for Continued Airworthiness to include CDCCL data.

Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 6 months after the effective date of this AD, or before December 16, 2008, whichever occurs first, revise the Airworthiness Limitations section of the Instructions for Continued Airworthiness to include the CDCCL data using a method approved in accordance with the procedures specified in paragraph (g)(1) of this AD. One approved method is EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006. Where the EADS CASA technical document refers to an EADS CASA component maintenance manual (CMM), use the applicable CMM specified in Table 1 of this AD.

TABLE 1.—APPROVED METHODS OF COMPLIANCE

CDCCL No.	CDCCL description	CMM	Revision	Date
8	Fuel pumps	Parker Hannifin CMM with Illustrated Parts List 28–22–12 (replaces CM 1C12–34).	5	January 10, 2008.
8	Centrifugal fuel boost pump ...	Parker Hannifin CMM with Illustrated Parts List CM 1C7–20, –21 (replaces CMM RR54170).	B	November 20, 2006.
9	Low level sensor	EADS CASA CMM with Illustrated Parts List 28–21–12	002	June 15, 2007.
10	3/4" shutoff motorized valve	Eaton CMM with Illustrated Parts List 28–20–81	2	June 20, 2006.
11	2" motorized spherical plug pressure relief valve.	Eaton CMM with Illustrated Parts List 28–10–63	3	June 20, 2006.
12	Signal conditioner	Gull CMM with Illustrated Parts List 28–40–61	3	June 28, 2007.
13	Fuel control unit	Zodiac Inter technique CMM 28–41–05	3	September 25, 2006.

Note 1: Table 1 does not include CMM 28–22–15, CE400150–E01, and C 17MQ0020–005SE, which are listed in EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006. These CMM document numbers no longer apply. In addition, CMM document number 28–21–81 in EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006, should be CMM document number 28–20–81.

(2) After accomplishing the actions specified in paragraph (f)(1) of this AD, no alternative CDCCLs may be used unless the CDCCLs are part of a later revision of EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006, that is approved by the Manager, ANM–116, International Branch, Transport Airplane Directorate, FAA, or the European Aviation Safety Agency (EASA) (or its delegated agent); or unless the CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (g)(1) of this AD.

FAA AD Differences

Note 2: This AD differs from the MCAI and/or service information as follows: Although EASA Airworthiness Directive 2007–0007, dated January 9, 2007, specifies to incorporate into the aircraft maintenance

program the fuel airworthiness limitation maintenance and inspection tasks defined in EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006, we have not included that action. The EADS CASA Technical Document does not contain FAL maintenance and inspection tasks.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, ANM–116, International Branch, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Shahram Daneshmandi, Aerospace Engineer, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–1112; fax (425) 227–1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective

actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

Related Information

(h) Refer to MCAI EASA Airworthiness Directive 2007–0007, dated January 9, 2007; and EADS CASA CN–235/C–295 Technical Document DT–0–C00–05001, Issue C, dated October 2006; for related information.

Material Incorporated by Reference

(i) None.

Issued in Renton, Washington, on April 24, 2008.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8–9569 Filed 4–30–08; 8:45 am]

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