- (6) A certification by the Lender that the Lender meets each of the requirements of the Program as set forth in the Act and the Board's rules in this part;
- 10. Section 400.206 is amended by revising paragraphs (c)(5)(ii)(C) and (c)(5)(iv) to read as follows:

#### § 400.206 Environmental requirements.

- \* \* \* (c) \* \* \*
- (5) \* \* \*
- (ii) \* \* \*
- (C) Adopt a previously completed EIS reasonably related to the project for which the proceeds of the loan sought to be guaranteed under the Program will be used, as described in 40 CFR 1506.3.
- (iv) If, on the basis of an environmental assessment, it is determined that an EIS is not required, a FONSI, as described in 40 CFR 1508.13 will be prepared. The FONSI will include the environmental assessment or a summary of it and be available to the public from the Board. The Executive Director shall maintain a record of these decisions, making them available to interested parties upon request. Requests should be directed to the Executive Director, Emergency Steel Guarantee Loan Program, 1099—14th Street, NW, Suite 2600 East, Washington, DC 20005. Prior to a final loan guarantee decision, a copy of the NEPA documentation shall be sent to the Board for consideration.
- 11. Section 400.207 is amended by revising paragraphs (b) introductory text and (c) to read as follows:

#### § 400.207 Application evaluation. \* \* \*

(b) Evaluation criteria. Applications that are determined to be eligible pursuant to paragraph (a) of this section shall be subject to a substantive review by the Board based upon the following evaluation factors, in order of importance:

(c) Decisions by the Board. Upon completion of the evaluation of an application and as soon as possible after its receipt, the Board will approve or deny an eligible application that is timely received under this Program. The Board shall notify the Applicants and the Borrower in writing of the approval or denial of an application as soon as possible. Approvals for loan Guarantees shall be conditioned upon compliance with § 400.208.

12. Section 400.208 is amended by revising paragraph (a)(3) to read as follows:

#### § 400.208 Issuance of the Guarantee.

- (a) \* \* \*
- (3) The Board's receipt of the Loan documents and any related instruments, in form and substance satisfactory to the Board, and the Guarantee, all properly executed by the Lender, Borrower, and any other required party other than the Board; and

13. Section 400.210 is amended by removing paragraph (c)(2), redesignating paragraph (c)(3) as paragraph (c)(2), and revising it, and adding a new paragraph (d), to read as follows:

#### § 400.210 Assignment or transfer of loans. \* \*

- (c) \* \* \*
- (2) Transfers by a non-Agent Lender of the non-guaranteed portion of the loan after payment of the guaranteed portion of the loan has been made under the Guarantee.
- (d) The Agent must hold and may not assign or transfer an interest in a loan guaranteed under the Program equal to at least the lesser of \$25 million or 15 percent of the aggregate amount of the loan. The Agent must hold and may not assign or transfer an interest in the unguaranteed portion of the loan which as a percentage of the Agent's overall interest in the loan is no less than the aggregate percentage of the loan which is not guaranteed. A non-Agent Lender must hold and may not transfer or assign an interest in the unguaranteed portion of the loan representing no less than five percent of such Lender's total interest in the Loan, except as provided paragraph (c)(2) of this section.
- 14. Section 400.211 is revised to read as follows:

#### § 400.211 Lender responsibilities.

The Lender shall have such obligations and duties to the Board as are set forth in the Guarantee.

15. Section 400.212 is revised to read as follows:

#### § 400.212 Guarantee.

The Board shall adopt a form of Guarantee to be used by the Board under the Program, and shall publish the Guarantee on its website. Modifications to the provisions of the form of Guarantee must be approved and adopted by the Board.

16. Section 400.213 is revised to read as follows:

#### § 400.213 Termination of obligations.

The Board shall have such rights to terminate the Guarantee as are set forth in the Guarantee.

17. Section 400.214 is revised to read as follows:

#### § 400.214 Participations in guaranteed loans.

- (a) Subject to paragraphs (b), (c) and (d) of this section, a Lender may distribute the risk of a portion of a loan guaranteed under the Program by sale of participations therein if:
- (1) Neither the loan note nor the Guarantee is assigned, conveyed, sold, or transferred in whole or in part;
- (2) The Lender remains solely responsible for the administration of the loan; and
- (3) The Board's ability to assert any and all defenses available to it under the Guarantee and the law is not adversely affected.
- (b) The following categories of entities may purchase participations in loans guaranteed under the Program:
  - (1) Eligible Lenders;
- (2) Private investment funds and insurance companies that do not usually invest in commercial loans;
- (3) Steel company suppliers or customers, who are interested in participating as a means of commencing or solidifying the supplier or customer relationship with the borrower; or
- (4) Any other entity approved by the Board on a case-by-case basis.
- (c) The Agent may not grant participations in that portion of its interest in a loan that may not be assigned or transferred under § 400.210(d). A Lender, other than the Agent, may not grant participations in that portion of its interest in a loan that may not be assigned or transferred under § 400.210(d).
- (d) At least 5 percent of any participation interest in a loan must be unguaranteed.

[FR Doc. 00-29812 Filed 11-17-00; 2:17 pm] BILLING CODE 3510-NC-U

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. 2000-NM-265-AD; Amendment 39-11980; AD 2000-23-10]

RIN 2120-AA64

Airworthiness Directives; Lockheed Model 188A and 188C Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to all Lockheed Model 188A and 188C series airplanes, that requires a revision of the Airplane Flight Manual (AFM) to add procedures for donning the flightcrew oxygen masks when the cabin altitude warning horn is activated. The actions specified by this AD are intended to prevent incapacitation of the flightcrew as a result of lack of oxygen and consequent loss of control of the airplane due to absence of AFM procedures for donning the flightcrew oxygen masks when the cabin altitude warning horn is activated.

DATE: Effective December 27, 2000.

ADDRESSES: Information pertaining to this amendment may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

#### FOR FURTHER INFORMATION CONTACT:

Thomas Peters, Aerospace Engineer, Systems and Flight Test Branch, ACE– 116A, FAA, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia 30349; telephone (770) 703–6063; fax (770) 703–6097.

#### SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Lockheed Model 188A and 188C series airplanes was published in the **Federal Register** on August 30, 2000 (65 FR 52677). That action proposed to require a revision of the Airplane Flight Manual to add procedures for donning the flightcrew oxygen masks when the cabin altitude warning horn is activated.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

#### Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

### Cost Impact

There are approximately 75 Model 188A and 188C series airplanes of the

affected design in the worldwide fleet. The FAA estimates that 32 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$1,920, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

#### **Regulatory Impact**

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### §39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**2000–23–10** Lockheed: Amendment 39–11980. Docket 2000–NM–265–AD.

Applicability: All Model 188A and 188C series airplanes, certificated in any category. Compliance: Required as indicated, unless accomplished previously.

To prevent incapacitation of the flightcrew and consequent loss of control of the airplane due to delays in donning oxygen masks in response to the activation of the cabin altitude warning horn; accomplish the following:

#### Revision to the Airplane Flight Manual

(a) Within 90 days after the effective date of this AD, revise the Emergency Procedures Section of the FAA-Approved Airplane Flight Manual (AFM) to include the following (which may be accomplished by inserting a copy of this AD in the AFM):

"Low Cabin Pressure Warning Light Comes On and Horn Starts Blowing

- a. Oxygen Masks—Don. Select 100% oxygen.
- b. If conditions dictate, initiate emergency descent.
  - c. Check cabin differential pressure gage.
- 1. If differential pressure is below 13.34 + 0.30 in. Hg, lower cabin altitude selector wheel
- 2. If differential pressure is at 13.34 + 0.30 in. Hg, descend to lower aircraft altitude.

**Note:** Warning horn can be silenced with cabin altitude warning horn switch."

#### **Alternative Methods of Compliance**

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Atlanta Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Atlanta ACO.

**Note 1:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Atlanta ACO.

#### **Special Flight Permit**

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### **Effective Date**

(d) This amendment becomes effective on December 27, 2000.

Issued in Renton, Washington, on November 6, 2000.

#### Donald L. Riggin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 00–28963 Filed 11–21–00; 8:45 am] BILLING CODE 4910–13–P

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. 2000-NM-213-AD; Amendment 39-11987; AD 2000-23-15]

#### RIN 2120-AA64

# Airworthiness Directives; Saab Model SAAB SF340A and SAAB 340B Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT. **ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain Saab Model SAAB SF340A and SAAB 340B series airplanes, that requires inspecting the connector on the refuel/defuel panel and the electrical connector on the illuminated placard to detect signs of fluid ingression or corrosion, and corrective actions. The actions specified by this AD are intended to prevent electrical shorts or arcing at the illuminated placard connector at the refuel/defuel panel, which could result in a potential ignition source for fuel vapors during fueling procedures. This action is intended to address the identified unsafe condition.

DATES: Effective December 27, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of December

27 2000

ADDRESSES: The service information referenced in this AD may be obtained from Saab Aircraft AB, SAAB Aircraft Product Support, S–581.88, Linkping, Sweden. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

#### FOR FURTHER INFORMATION CONTACT:

Norman B. Martenson, Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–2110; fax (425) 227–1149.

#### SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Saab Model SAAB SF340A and SAAB 340B series airplanes was published in the **Federal Register** on September 19, 2000 (65 FR 56507). That action proposed to require inspecting the connector on the refuel/defuel panel and the electrical connector on the illuminated placard to detect signs of fluid ingression or corrosion, and corrective actions.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

#### Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

#### **Cost Impact**

The FAA estimates that 289 Model SAAB SF340A and SAAB 340B series airplanes of U.S. registry will be affected by this AD, that it will take approximately 4 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$69,360, or \$240 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

#### **Regulatory Impact**

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not

have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000–23–15 **Saab Aircraft AB:** Amendment 39–11987. Docket 2000–NM–213–AD.

Applicability: Model SAAB SF340A series airplanes, serial numbers –004 through –159 inclusive; and Model SAAB 340B series airplanes, serial numbers –160 through –459 inclusive; certificated in any category; on which a refuel/defuel panel having part number 7239160–505 is installed.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.