

at the time the mortgage was originated. To determine whether mortgages may be counted under a particular family income level (e.g., low- or very low-income), the income of the mortgagor is compared to the median income for the area at the time the mortgage was originated, using the appropriate percentage factor provided under § 1281.1.

(b) *No double-counting.* A mortgage may be counted only once toward the achievement of the prospective mortgage purchase housing goal, even if it satisfies multiple criteria for the prospective mortgage purchase housing goal.

(c) *Application of median income.* For purposes of determining an area's median income under § 1281.1, the area is:

(1) The metropolitan area, if the residence that secures the mortgage is in a metropolitan area; and

(2) In all other areas, the county in which the property is located, except that where the State non-metropolitan median income is higher than the county's median income, the area is the State non-metropolitan area.

(d) *Sampling not permitted.* Performance under the housing goals for each year shall be based on a tabulation of each mortgage during that year; a sampling of such purchases is not acceptable.

■ 6. Amend § 1282.13 by:

■ a. Revising paragraph (b) introductory text, (b)(1) and (8);

■ b. Adding paragraph (c)(4);

■ c. Removing paragraph (d);

■ d. Redesignating paragraph (e) as paragraph (d); and

■ e. Adding new paragraph (e).

The revisions and additions read as follows:

**§ 1281.13 Special counting requirements.**

\* \* \* \* \*

(b) *Not counted.* The following transactions or activities shall not be counted for purposes of the housing goals, meaning that in calculating the applicable percentage target level, they shall be excluded from both the numerator (i.e., AMA mortgages acquired during the calendar year that are for very low-income families, low-income families, or families in low-income areas) and the denominator (i.e., total AMA mortgages acquired during the calendar year), even if the transaction or activity would otherwise be counted under paragraph (c) of this section:

(1) Purchases of participation interests in AMA mortgages from another Bank,

except as provided in paragraph (e) of this section;

\* \* \* \* \*

(8) Purchases of subordinate lien mortgages;

\* \* \* \* \*

(c) \* \* \*

(4) *Non-conventional mortgages.* The purchase of a non-conventional single-family mortgage shall be treated as a mortgage purchase for purposes of the housing goals only if the mortgage was acquired from a community-based AMA user.

\* \* \* \* \*

(e) *Mortgage participation transactions.* Where two or more Banks acquire a participation interest in the same mortgage simultaneously, the mortgage will be counted on a *pro rata* basis for the prospective mortgage purchase housing goal for each Bank with a participation interest.

■ 7. Amend § 1281.14 by revising paragraph (a) to read as follows:

**§ 1281.14 Determination of compliance with housing goals; notice of determination.**

(a) *Determination of compliance with housing goals.* On an annual basis, FHFA will determine each Bank's performance under each housing goal and will publish the final determinations. FHFA will publish its final determination including the numbers and percentages for each Bank's AMA purchases that meet each of the housing goals criteria, including loans to low-income families, loans to very low-income families, and loans to families in low-income areas, including by each of the defined categories. FHFA's determination will include these numbers in total and separated into purchase money mortgages, refinancing mortgages, conventional mortgages, and non-conventional mortgages.

\* \* \* \* \*

■ 8. Amend § 1281.15 by revising paragraphs (a) and (b) to read as follows:

**§ 1281.15 Housing plans.**

(a) *Housing plan requirement.* For any year after 2023, if the Director determines that a Bank has failed to meet any housing goal and that the achievement of the housing goal was feasible, the Director may require the Bank to submit a housing plan for approval by the Director.

(b) *Nature of plan.* If the Director requires a housing plan, the housing plan shall:

(1) Be feasible;

(2) Be sufficiently specific to enable the Director to monitor compliance periodically;

(3) Describe the specific actions that the Bank will take to achieve the housing goal for the next calendar year;

(4) Address any additional matters relevant to the housing plan as required, in writing, by the Director; and

(5) Address any alternative target levels for which the Bank has submitted a request under § 1281.11(c)(1).

\* \* \* \* \*

■ 9. Revise Subpart C to read as follows:

**Subpart C—Reporting Requirements**

**§ 1281.20 Reporting requirements.**

(a) *General.* Each Bank must collect and submit to FHFA any data that FHFA determines to be necessary for FHFA to evaluate transactions and activities under the Bank housing goals.

(b) *Reporting for prospective mortgage purchase housing goal.* Each Bank must collect data on each AMA mortgage purchased by the Bank. The data must include any data elements specified by FHFA. On no less frequent than an annual basis, each Bank must submit such data to FHFA in accordance with the Data Reporting Manual.

(c) *Reporting for small member participation housing goal.* Each Bank must collect data on AMA user asset size. On no less frequent than an annual basis, each Bank must submit such data to FHFA in accordance with the Data Reporting Manual.

(d) *Other reporting.* Each Bank must provide to FHFA such additional reports, information, and data as FHFA may request from time to time.

**Mark A. Calabria,**

*Director, Federal Housing Finance Agency.*

[FR Doc. 2020–12345 Filed 6–24–20; 8:45 am]

BILLING CODE 8070–01–P

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA–2020–0568; Project Identifier MCAI–2020–00505–A; Amendment 39–21148; AD 2020–13–03]

RIN 2120–AA64

**Airworthiness Directives; XtremeAir GmbH Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is superseding Airworthiness Directive (AD) 2018–07–15 for certain XtremeAir GmbH Model XA42 airplanes. This AD results from

mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracking of the diagonal struts of the engine mount frame with potential detachment of the engine from the airplane. The FAA is issuing this AD to require actions to address the unsafe condition on these products.

**DATES:** This AD is effective July 15, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of July 15, 2020.

The FAA must receive comments on this AD by August 10, 2020.

**ADDRESSES:** You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** (202) 493-2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD, contact XtremeAir GmbH, Harzstrasse 2, Am Flughafen Cochstedt, D-39444 Hecklingen, Germany; phone: +49 39267 60999 0; fax: +49 39267 60999 20; email: [info@xtremeair.de](mailto:info@xtremeair.de); internet: <https://www.xtremeair.com>.

You may review copies of the referenced service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148. It is also available on the internet at <https://www.regulations.gov> by searching for locating Docket No. FAA-2020-0568.

#### Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0568; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, any comments received, and other information. The street address for

Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Jim Rutherford, Aerospace Engineer, FAA, Policy and Innovation Division, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: [jim.rutherford@faa.gov](mailto:jim.rutherford@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The FAA issued AD 2018-07-15, Amendment 39-19246 (83 FR 15036; April 9, 2018) ("AD 2018-07-15") to address an unsafe condition on XtremeAir GmbH Model XA42 airplanes equipped with an engine mount part number XA42-7120-151. AD 2018-07-15 required repetitive inspections of the engine mount strut junction for cracks and was based on MCAI originated by an aviation authority of another country. The FAA issued AD 2018-07-15 to detect and address cracking of the engine mount frame, which could lead to detachment of the engine in-flight and result in loss of control.

##### Actions Since AD 2018-07-15 Was Issued

Since the FAA issued AD 2018-07-15, during a scheduled maintenance inspection of a Model XA42 airplane, an occurrence was reported of separation of both left-hand and right-hand diagonal struts of the engine mount frame.

The European Union Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, superseded its MCAI and issued EASA AD No. 2019-0239R1, dated December 18, 2019 (referred to after this as "the MCAI"), to correct this unsafe condition for XtremeAir GmbH Model XA42 airplanes. The MCAI states:

After that AD [EASA Emergency AD 2018-0050-E] was issued, during a scheduled maintenance inspection of an XA42 airplane, an occurrence was reported of separation of both left-hand and right-hand diagonal struts of the affected part.

This condition, if not detected and corrected, could lead to in-flight detachment of the engine, possibly resulting in loss of control of the airplane.

Prompted by this new finding, XtremeAir issued the SB [XtremeAir Service Bulletin SB-XA42-2019-008, Issue A.00] to provide limitations and inspection instructions. Consequently, EASA issued Emergency AD 2019-0239-E, retaining the requirements of EASA Emergency AD 2018-0050-E, which was superseded, reducing the inspection interval to each pre-flight check and requiring an Aircraft Flight Manual (AFM) limitation to prohibit aerobatic manoeuvres, and installation of a corresponding placard.

Since that AD was issued, the investigations were completed, and based on the outcome, XtremeAir revised the SB (now at revision B.00), providing instructions to accomplish the repetitive inspections at a different regime.

For the reason described above, this AD is revised accordingly to update the inspection interval, and to allow the removal of the prohibition to accomplish aerobatic manoeuvres.

This AD is still considered an interim action and further AD action may follow.

You may examine the MCAI on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0568.

#### Relative Service Information Under 1 CFR Part 51

The FAA reviewed XtremeAir GmbH Mandatory Service Bulletin SB-XA42-2019-008, Issue B.00, dated December 4, 2019. The service information contains procedures for inspection of the area of the junction of the left-hand and right-hand diagonal struts of the engine mount, forward of the oil cooler, for cracks. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### FAA's Determination and Requirements of the AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI and service information referenced above. The FAA is issuing this AD because it evaluated all information provided by the State of Design Authority and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### FAA's Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because cracking of the engine mount frame could lead to in-flight detachment of the engine and result in loss of control of the airplane. AD 2018-07-15 only required an inspection of the left-hand diagonal strut. Since AD 2018-07-15 was issued, an occurrence was reported of separation of both the left-hand and right-hand diagonal struts of the engine mount frame. The

additional inspection required by this AD is necessary to detect cracks in the right-hand diagonal strut of the engine mount frame. Therefore, the FAA finds that notice and opportunity for public comment before issuing this AD are impracticable. In addition, for the reasons stated above, the FAA finds that good cause exists for making this amendment effective in fewer than 30 days.

#### Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and the FAA did not precede it by notice and opportunity for public comment. The FAA invites you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA–2020–0568; Project Identifier MCAI–2020–00505–A” at the beginning of your comments. The FAA specifically invites comments on the overall regulatory, economic, environmental, and energy aspects of this AD. The FAA will consider all comments received by the closing date and may amend this AD because of those comments.

The FAA will post all comments we receive, without change, to <https://www.regulations.gov>, including any personal information you provide. The FAA will also post a report summarizing each substantive verbal contact received about this AD.

#### Costs of Compliance

The FAA estimates that this AD will affect 16 products of U.S. registry. The FAA also estimates that it will take 0.5 work-hour per product per inspection cycle to inspect the diagonal struts of the engine mount. The average labor rate is \$85 per work-hour. Required parts would cost about \$0 per product.

Based on these figures, the FAA estimates the cost of the inspection on U.S. operators to be \$680.00 or \$42.50 per product per inspection cycle.

In addition, the FAA estimates that replacing an engine mount, if necessary, will take 24 work-hours and require parts costing \$5,000, for a cost of \$7,040 per product. The FAA has no way of determining the number of products that may need this action.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: general requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Flexibility Act

The requirements of the Regulatory Flexibility Act (RFA) do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because FAA has determined that it has good cause to adopt this rule without notice and comment, RFA analysis is not required.

#### Regulatory Findings

The FAA determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866, and
- (2) Will not affect intrastate aviation in Alaska.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by:
  - a. Removing Airworthiness Directive 2018–07–15, Amendment 39–19246 (83 FR 15036, April 9, 2018); and
  - b. Adding the following new AD:

**2020–13–03 XtremeAir GmbH Airplanes:**  
Amendment 39–21148; Docket No. FAA–2020–0568; Project Identifier MCAI–2020–00505–A.

#### (a) Effective Date

This airworthiness directive (AD) becomes effective July 15, 2020.

#### (b) Affected ADs

This AD replaces AD 2018–07–15, Amendment 39–19246 (83 FR 15036, April 9, 2018).

#### (c) Applicability

This AD applies to XtremeAir GmbH Model XA42 airplanes, all serial numbers, certificated in any category.

#### (d) Subject

Air Transport Association of America (ATA) Code 71: Power Plant.

#### (e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and address an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracking of the diagonal struts of the engine mount frame. The FAA is issuing this AD to prevent the in-flight detachment of the engine, which could result in loss of control of the airplane.

#### (f) Actions and Compliance

For airplanes with an engine mount part number (P/N) XA42–7120–151 installed, unless already done, do the following actions in paragraphs (f)(1) and (2) of this AD.

(1) Before the next acrobatic flight after July 15, 2020 (the effective date of this AD) or before the engine mount accumulates 50 hours time-in-service (TIS), whichever occurs later, and thereafter at intervals not to exceed 10 acrobatic flights, visually inspect the junction of the left-hand and right-hand diagonal struts of the engine mount, forward of the oil cooler, for cracks in the area shown in the photographs in XtremeAir GmbH Mandatory Service Bulletin SB–XA42–2019–008, Issue B.00, dated December 4, 2019. For purposes of this AD, an acrobatic flight is counted for each flight during which a load factor of 6g is exceeded.

(i) If there is a crack, before further flight, replace the engine mount with an airworthy engine mount with zero hours TIS, an engine mount that has passed the inspection required by this AD, or an engine mount that is not P/N XA42–7120–151.

(ii) After completing the initial inspection, begin or continue to count the acrobatic flights and record in the maintenance records.

(2) As of July 15, 2020 (the effective date of this AD), do not install on the airplane an engine mount P/N XA42–7120–151 unless it is an airworthy engine mount with zero hours TIS or is an engine mount that has passed the inspection required by this AD.

#### (g) Installation Prohibition

For airplanes without engine mount P/N XA42–7120–151 installed, as of July 15, 2020 (the effective date of this AD), do not install engine mount P/N XA42–7120–151.

**(h) Alternative Methods of Compliance (AMOCs)**

The Manager, Small Airplane Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Policy and Innovation Division, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: [jim.rutherford@faa.gov](mailto:jim.rutherford@faa.gov). Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

**(i) Special Flight Permit**

Special flight permits are prohibited.

**(j) Related Information**

Refer to MCAI European Union Aviation Safety Agency AD No. 2019-0239R1, dated December 18, 2019, for related information. You may examine the MCAI on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0568.

**(k) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) XtremeAir GmbH Mandatory Service Bulletin SB-XA42-2019-008, Issue B.00, dated December 4, 2019.

(ii) [Reserved]

(3) For XtremeAir GmbH service information identified in this AD, contact XtremeAir GmbH, Harzstrasse 2, Am Flughafen Cochstedt, D-39444 Hecklingen, Germany; phone: +49 39267 60999 0; fax: +49 39267 60999 20; email: [info@xtremeair.de](mailto:info@xtremeair.de); internet: <https://www.xtremeair.com>.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148. It is also available on the internet at <https://www.regulations.gov> by searching for locating Docket No. FAA-2020-0568.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email: [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on June 10, 2020.

**Lance T. Gant,**

*Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2020-13659 Filed 6-24-20; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA-2020-0091; Product Identifier 2020-NM-012-AD; Amendment 39-19916; AD 2020-11-12]**

**RIN 2120-AA64**

**Airworthiness Directives; The Boeing Company Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain The Boeing Company Model 737-8 and 737-9 airplanes. This AD was prompted by a report that certain exterior fairing panels on the top of the engine nacelle and strut (the thumbnail fairing and mid strut fairing panels) may not have the quality of electrical bonding necessary to ensure adequate shielding of the underlying wiring from the electromagnetic effects of high intensity radiated fields (HIRF), which could potentially lead to a dual-engine power loss event and/or display of hazardously misleading primary propulsion parameters. This AD requires a detailed inspection of the thumbnail fairing panels and mid strut fairing panels for excessive rework of the metallic (aluminum foil) inner surface layer, replacement of any excessively reworked panels, and modification of the thumbnail fairing assembly to ensure adequate bonding. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective July 30, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of July 30, 2020.

**ADDRESSES:** For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740-5600; telephone 562-797-1717; internet <https://www.myboeingfleet.com>. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0091.

**Examining the AD Docket**

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0091; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:**

Christopher Baker, Aerospace Engineer, Propulsion Section, FAA, Seattle ACO Branch, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3552; email: [christopher.r.baker@faa.gov](mailto:christopher.r.baker@faa.gov).

**SUPPLEMENTARY INFORMATION:****Discussion**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain The Boeing Company Model 737-8 and 737-9 airplanes. The NPRM published in the **Federal Register** on February 26, 2020 (85 FR 11000). The NPRM was prompted by a report that certain exterior fairing panels on the top of the engine nacelle and strut (the thumbnail fairing and mid strut fairing panels) may not have the quality of electrical bonding necessary to ensure adequate shielding of the underlying wiring from the electromagnetic effects of HIRF, which could potentially lead to a dual-engine power loss event and/or display of hazardously misleading primary propulsion parameters. The NPRM proposed to require a detailed inspection of the thumbnail fairing panels and mid strut fairing panels for excessive rework of the metallic (aluminum foil) inner surface layer, replacement of any excessively reworked panels, and modification of the thumbnail fairing assembly to ensure adequate bonding.

The FAA is issuing this AD to address this condition, which could result in a forced off-airport landing or excessive flightcrew workload.

**Comments**

The FAA gave the public the opportunity to participate in developing this final rule. The FAA received comments from several organizations and individuals. The following discussion presents the comments